

# FX 67-K-150-17-TE

Max thickness 15.04% at 42.0% of the chord

Max camber 4.78% at 44.7% of the chord

Mach = 0.0000 - NCrit = 6.00

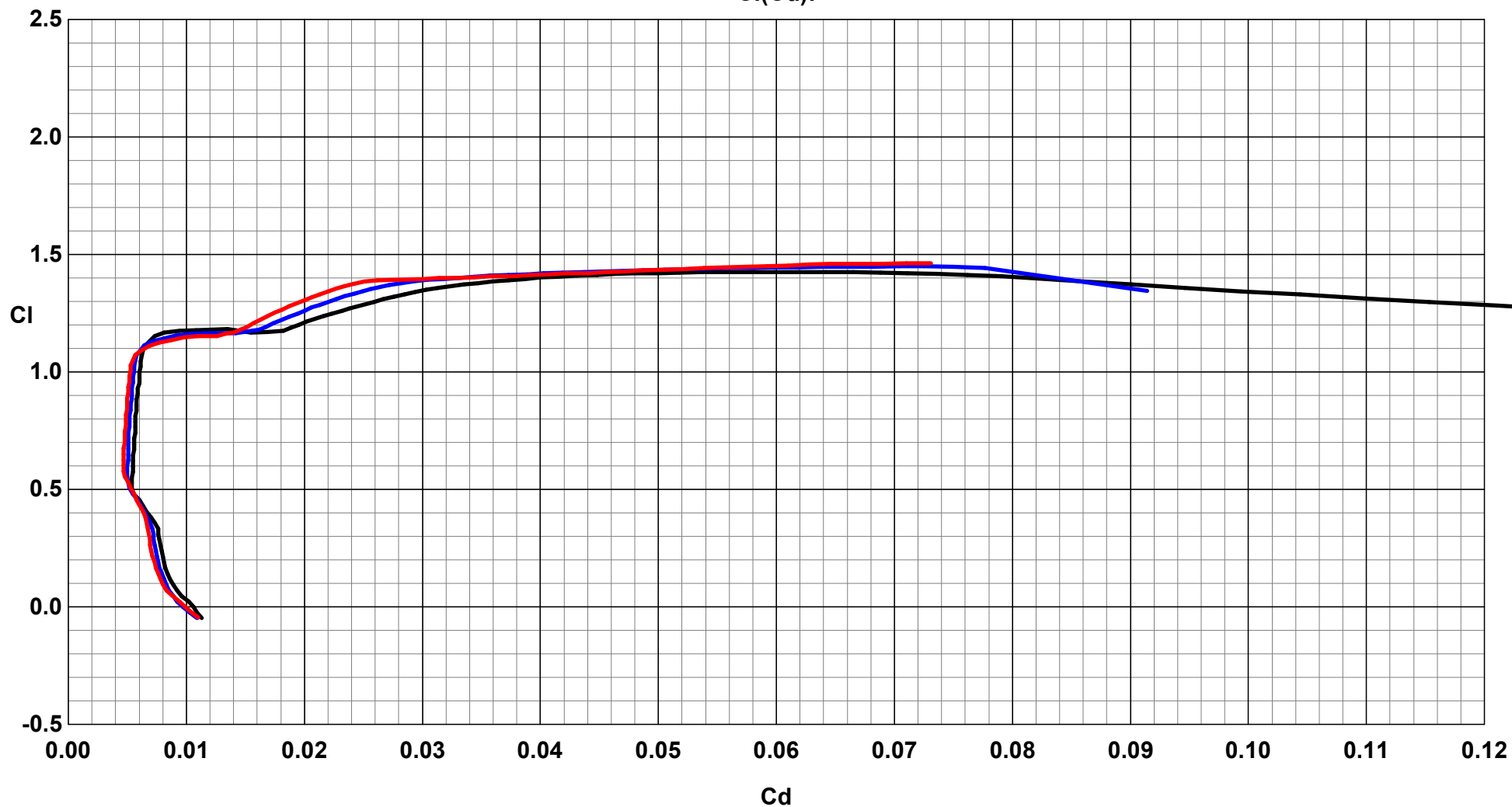
Re 1500000 =

Re 2000000 =

Re 2500000 =



Cl(Cd):



# FX 67-K-150-17-TE

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Mach = 0.0000 - NCrit = 6.00

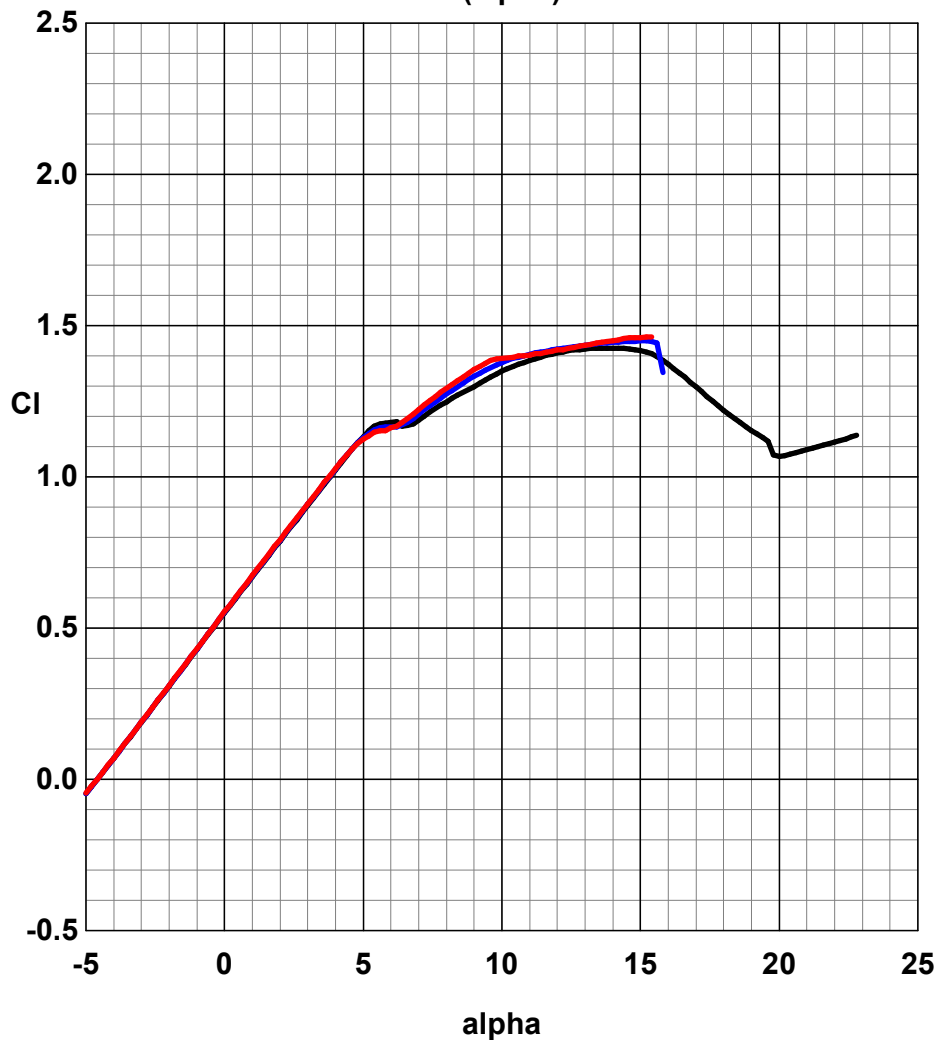
Re 1500000 =

Re 2000000 =

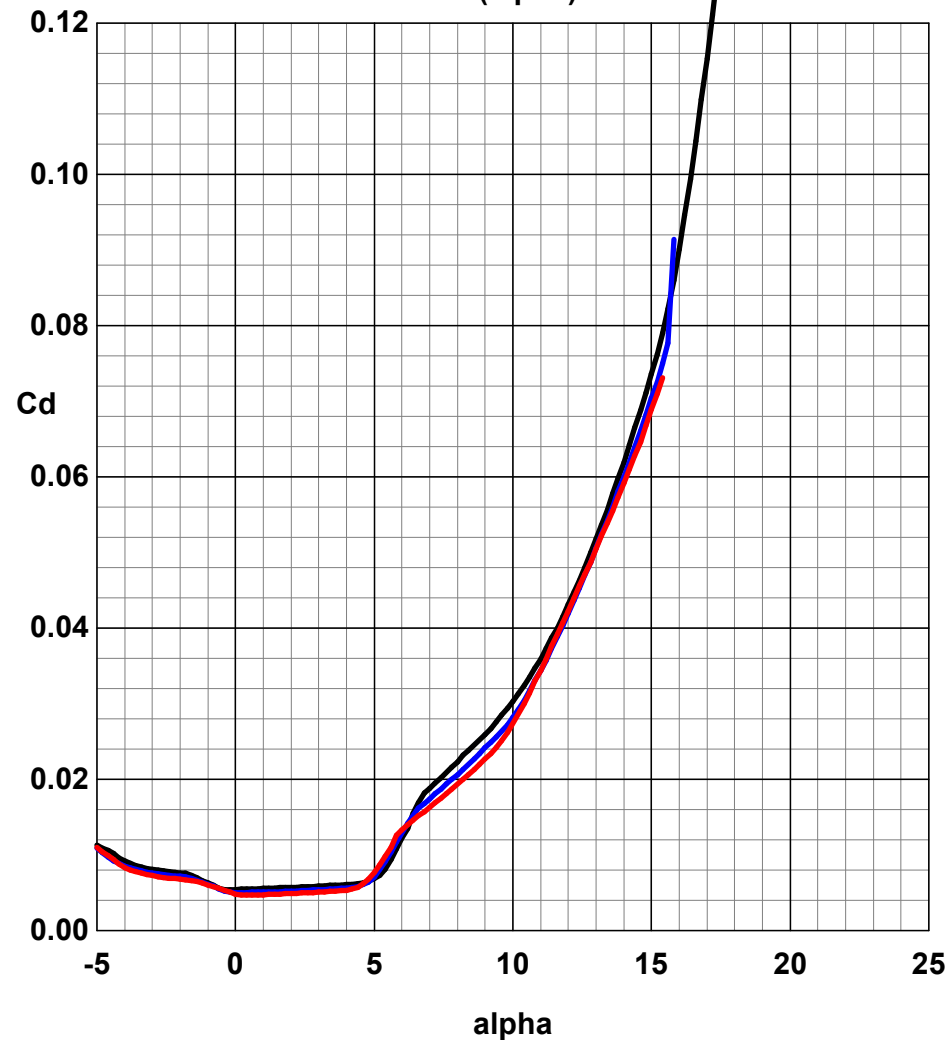
Re 2500000 =



Cl(alpha):



Cd(alpha):



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Max thickness 15.04% at 42.0% of the chord

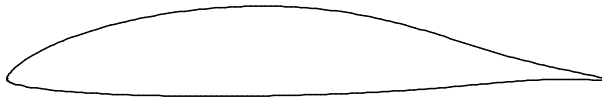
Max camber 4.78% at 44.7% of the chord

Mach = 0.0000 - NCrit = 6.00

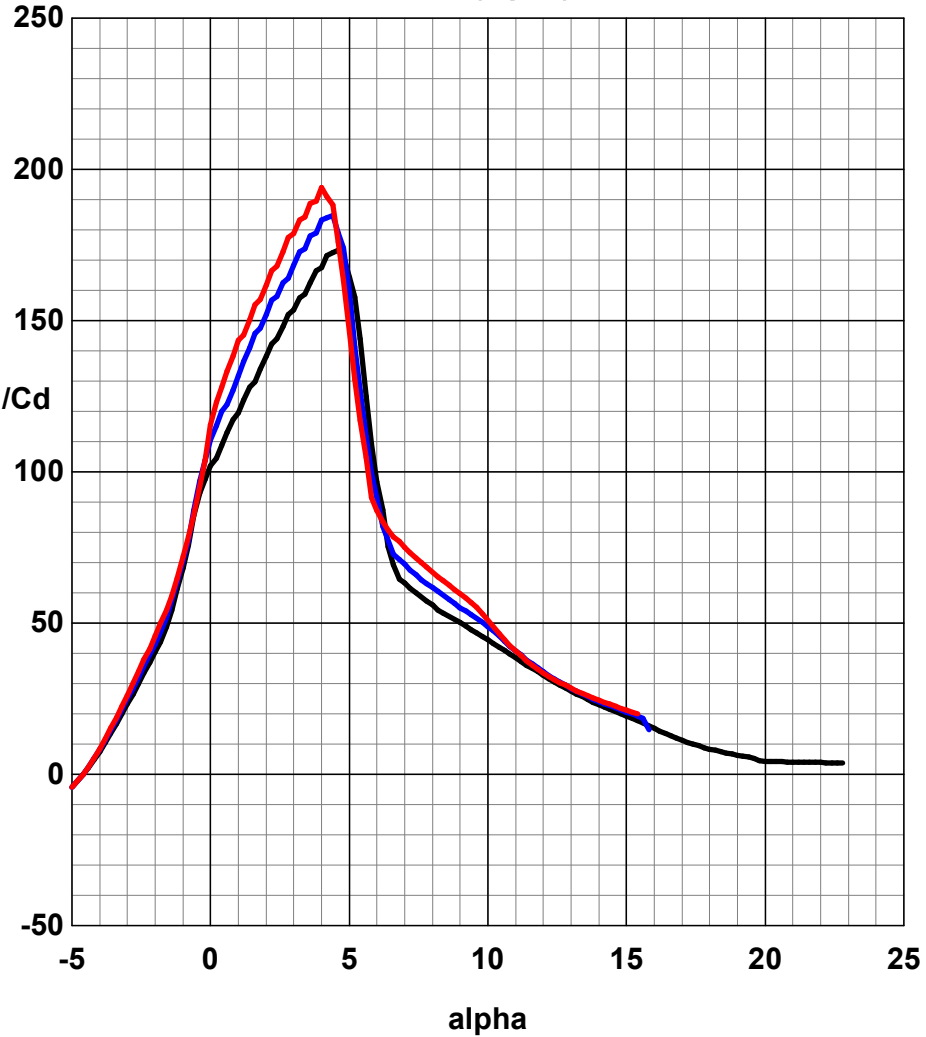
Re 1500000 =

Re 2000000 =

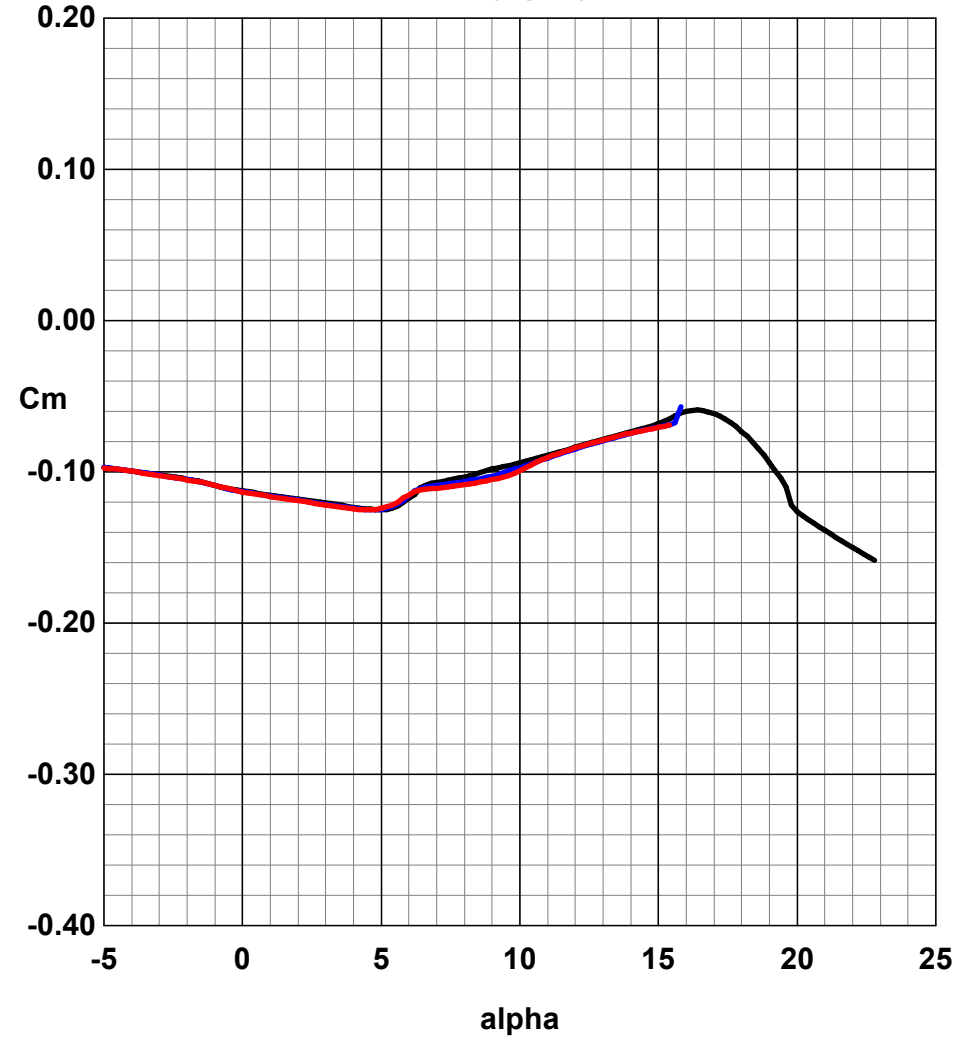
Re 2500000 =



Cl/Cd(alpha):



Cm(alpha):



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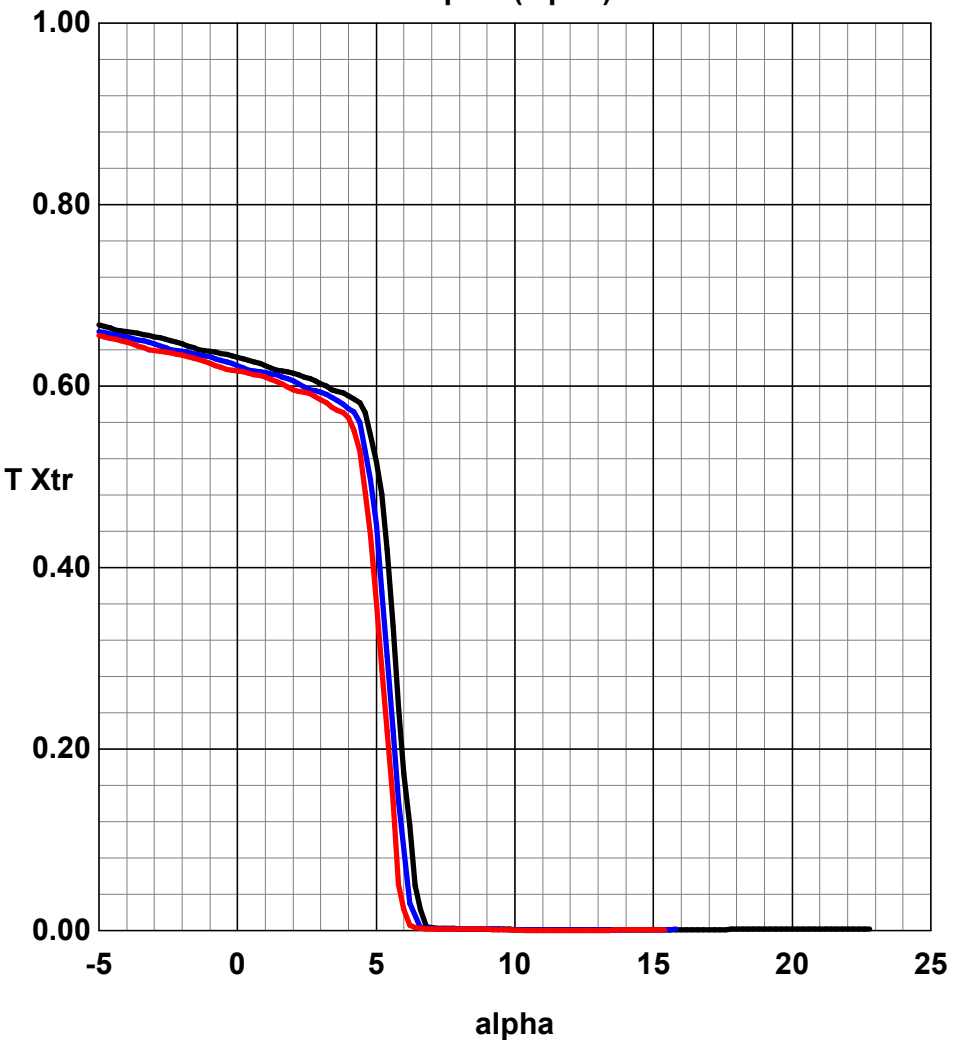
Re 1500000 =

Re 2000000 =

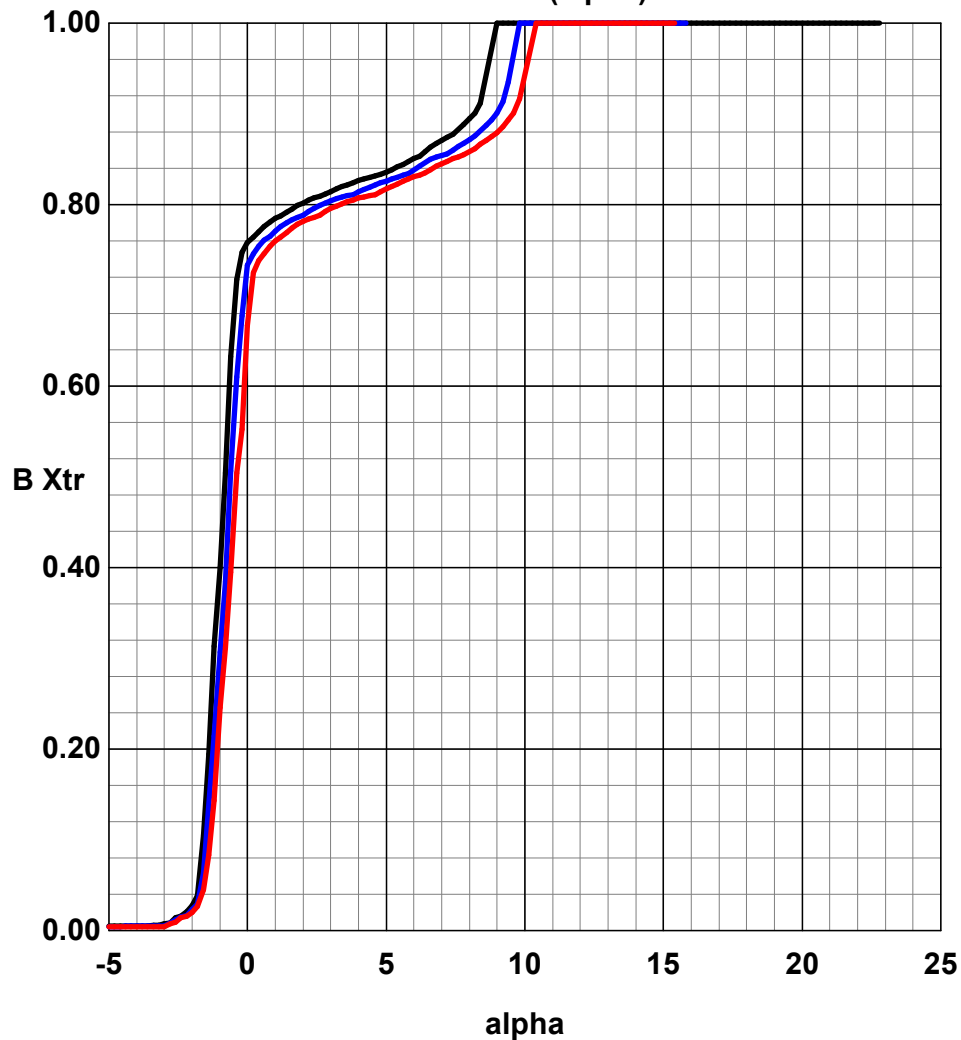
Re 2500000 =



Top Xtr(alpha):



Bottom Xtr(alpha):



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Re 1500000 =

Re 2000000 =

Re 2500000 =



Power Factor(alpha) =  $Cl^{1.5}/Cd(alpha)$  :

