

# **INSTRUCTION AND MAINTENANCE MANUAL FOR JABIRU 2200 AIRCRAFT ENGINE**

**Document Number JEM2201-7**



This Manual has been prepared as a guide to correctly operate, maintain and service the Jabiru 2200 engine.

Should you have any questions or doubts about the contents of this manual, please contact Jabiru Aircraft Pty Ltd.

**Applicable to Jabiru 2200A, 2200B & 2200J Models**



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### Issue Notes:

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|   |                             |

## 2 DESCRIPTION OF DESIGN

It is said that "aircraft are designed around available engines".

Jabiru believe that the Jabiru range of very light engines will now offer new opportunities for light aircraft designers, to develop a new generation of light aircraft.

Jabiru engines are designed to be manufactured in small batch quantities using the very latest Computer Numerically Controlled (CNC) machine tools. The vast majority of the components are manufactured in Southern Queensland in a network of high technology small companies. The crankcase halves, cylinder, crankshaft, starter motor housings, gearbox cover (the gearbox powers the distributor rotors) and coil mounts together with many smaller components are machined using the latest CNC machine tools. The sump (oil pan) is the only casting. The cylinders are machined from solid bar 4140 chrome molybdenum alloy steel, with the pistons running directly in the steel bores. The crankshaft is also machined from solid bar 4140 chrome molybdenum alloy steel, the journals of which are precision ground prior to being Magnaflux inspected. The camshaft is manufactured from 4140 chrome molybdenum alloy steel. (journals and cams nitrided).

The propeller is direct crankshaft driven and does not use a reduction gearbox. This facilitates its light-weight design and keeps maintenance costs to a minimum. The crankshaft features a removable propeller flange which enables the easy replacement of the front crankshaft seal and provides for a propeller shaft extension to be fitted, should this be required for particular applications. Cylinder heads are machined from solid aluminium billet which is purchased directly from one of Australia's largest aluminium companies, as is all alloy used in the engine, thereby providing a substantive quality trail to material source. Conrods are machined from 4140 alloy steel, the 45mm big end bearings are of the automotive slipper type.

Various components of the engines are sourced from many Suppliers. These items include camshaft followers, and the bendix gear in the starter motor. The ignition coils are also sourced from outside Suppliers, and are modified by Jabiru for their own particular application.

An integral alternator using rare earth magnets, provides alternating current for battery charging and electrical accessory drive. The alternator is attached to the flywheel and is driven directly by the crankshaft. The ignition system is a transistorised electronic system; two fixed coils mounted adjacent to the flywheel are energised by rare earth magnets attached to the flywheel. The passing of the coils by the magnets creates the high voltage current which is then transported by high tension leads to the centre post of two automotive type distributors (which are simply rotors and caps) before distribution to automotive spark plugs, two in the top of each cylinder head. The ignition system is fixed timing and, therefore, removes the need for timing adjustment. It is suppressed to prevent radio interference. The ignition system is fully redundant, self-generating and does not depend on battery power.

The crankshaft is designed with a double bearing at the propeller flange end and a main bearing between each big end; it therefore does not have flying webs. 48mm main bearings are also of the automotive slipper type. Thrust bearings are located for and aft of the front double bearing allowing either tractor or pusher installation.

Pistons are sourced in Australia and are re-machined to include a piston pin circlip groove. They are fitted with 3 rings, the top rings being cast iron to complement the chrome molybdenum cylinder bores. Valves are 7mm (stem dia) which are purpose manufactured for the Jabiru engine.

The valve gear includes pushrods from the camshaft from the camshaft followers to valve rockers which are CNC machined from steel plate, induction hardened and polished on contact surfaces and mounted on a shaft through a teflon bronze-steel bush. Valve guides are manufactured from aluminium/bronze, as is found in larger aero engines and high performance racing engines. Replaceable valve seats are of nickel steel and are shrunk into the aluminium cylinder heads. The valve gear is lubricated from the oil gallery. Engines use solid lifters and require periodic checking of rocker to valve clearance.

An internal gear pump, direct mounted on the camshaft and incorporating a small automotive spin-on filter, provides engine lubrication. An oil cooler adapter is provided. Most installations require an oil cooler to meet oil temperature limits.

The standard engines are supplied with two RAMAIR cooling ducts, which have been developed by Jabiru to facilitate the cooling of the engine and direct air from the propeller to the critical areas of the engine, particularly the cylinder heads and barrels. The fitment of these RAMAIR cooling ducts is a great bonus for the home builder or engine installer, as they obviate the need to design and manufacture baffles and the establishment of a plenum chamber, which is the traditional method of cooling air-cooled aircraft engines. The fact that these baffles and plenum chamber are not required also ensures a "cleaner" engine installation, which in turn facilitates maintenance and inspection of the engine and engine component. So the hard work of engine installation has largely been done for you by the Jabiru design team. RAMAIR ducts are available for tractor or pusher configurations. Special ducts are available for certain installations.

The engine is fitted with a 1.5 kw starter motor, which is also manufactured by Jabiru. The engine has very low vibration levels, however it is also supported by four large rubber shock mounts attached to the engine mounts at the rear of the engine. An optional bed mount is available.

The fuel induction system comprises a BING pressure compensating carburettor. Following carburetion, the fuel/air mixture is transported through a swept plenum chamber bolted to the sump casting, in which the mixture is warmed prior to entering short induction tubes attached to the cylinder heads.



An effective stainless steel exhaust and muffler system is fitted as standard equipment, ensuring very quiet operations, which in the Jabiru aircraft have been measured at 62dB at 1000' full power flyover (for 2200 engine).

For those owners wanting to fit vacuum instruments to their aircraft the Jabiru engine design includes a vacuum pump drive, direct mounted through a coupling on the rear of the crankshaft.

The Jabiru engine is manufactured within an Australian Civil Aviation Safety Authority (CASA) approved Quality Assurance System to exacting standards.

Jabiru recommend a TBO of 1000 hours of engines to s/n 709. From s/n 710 TBO has been extended to 2000 hours with a top end overhaul done at 1000 hours, certain conditions do apply for this situation. A Guaranteed Fixed Price Overhaul Plan\* is offered with both engines. Contact your regional distributor or Jabiru Aircraft for details. Top End overhauls are also available on later model engines.

Jabiru engine Warranty\* is 200 hours or 12 months (whichever comes first) from date of sale or from date of independently verified first flight.

*\*Conditions Apply*

## 2.1 Details

- Stroke
- Cylinder Horizontally Opposed
- 1 Central Camshaft
- Push Rods
- Over Head Valves (OHV)
- Solid Valve Lifters
- Ram Air Cooled
- Wet Sump Lubrication
- Direct Propeller Drive
- Dual Transistorised Magneto Ignition
- Integrated AC Generator
- Electric Starter
- Mechanical Fuel Pump
- Naturally Aspirated – 1 Pressure Compensating Carburettor
- Bearing Crankshaft

## 2.2 Manufacturer

Jabiru Aircraft Pty Ltd,  
P.O. Box 5186,  
Bundaberg West,  
Queensland 4670



## 2.3 Manuals

Instruction and Maintenance Manual  
Installation Manual  
Parts Catalogue



### 3 Specifications

#### 3.1 Dimensions and Weights

|                                      |  |
|--------------------------------------|--|
| Bore:                                | 97.5 mm  |
| Stroke:                              | 74 mm  |
| Displacement:                        | 2209 cc  |
| Compression Ratio:                   | 8.3:1 s/n 1004 onwards have 8:1                          |
| Direction of Rotation on Prop Shaft: | Clockwise (Pilot's view, tractor applications.)          |
| Engine Curb Weight:                  | 60 kg complete with Engine Oil, Exhaust & Starter Motor. |



A.C.N. 010 910 077

JABIRU AIRCRAFT P/L

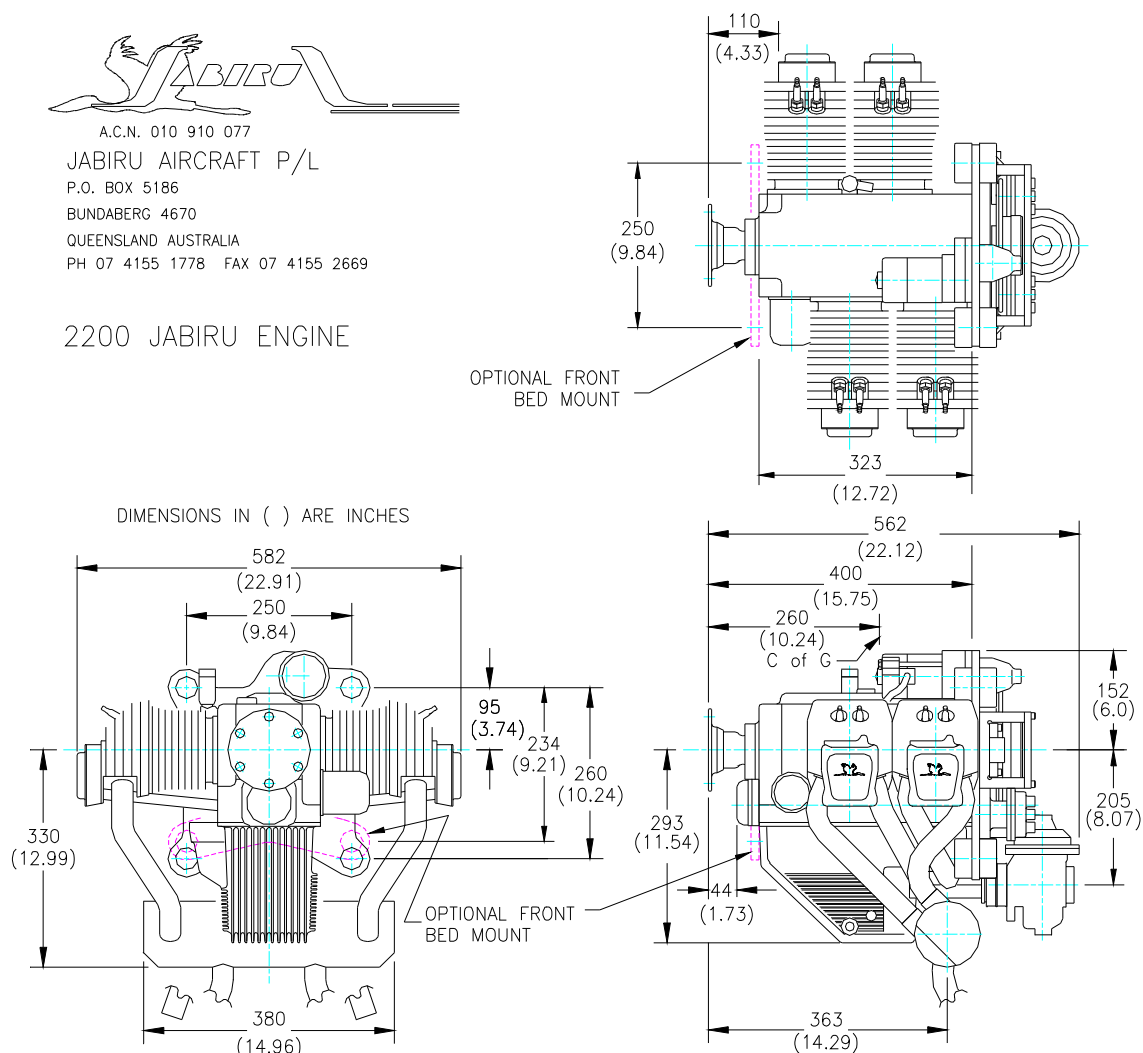
P.O. BOX 5186

BUNDABERG 4670

QUEENSLAND AUSTRALIA

PH 07 4155 1778 FAX 07 4155 2669

#### 2200 JABIRU ENGINE



**Figure 1. Drawing 9404041/1 Engine Dimensions**

#### 3.2 Normal Equipment

|                  |  |
|------------------|--|
| Ignition Unit:   | Jabiru dual ignition - breakerless transistorised. Battery Independent |
| Ignition Timing: | 25 degrees BTDC  |

Firing Order: 1-3-2-4  
 Spark Plugs: NGK D9EA  
 Electrode Gap: 0.55 - 0.6mm (0.022" - 0.024")  
 Generator: Jabiru, permanently excited single phase  
                   AC generator with rectifier/regulator  
 DC Output: 10 amps (continuous)  
 Carburettor: BING constant depression Type 64/32 or type 94/40  
 Air Intake Filter: 1 x folder paper cartridge  
 Fuel Filtration: 0.1 mm (100 Micron) maximum particle size.  
 Fuel Pump: Camshaft driven diaphragm type  
 Starting System: Electric 12 V / 1.5 kW  
 Oil Filter: RYCO Z 386 or equivalent

### 3.3 Performance Data

Take Off/Maximum Continuous RPM 60 kW (80 hp) @ 3300 RPM

### 3.4 Fuel Consumption

Fuel Consumption @ Takeoff/Max Continuous Rating 21 litres/hr

### 3.5 Fuel and Lubricant

Fuel: AVGAS 100 LL & AVGAS 100/130  
           Leaded and Unleaded Automotive Gasoline above 95 Octane Ron

#### 3.5.1 Run in Period

| Oil Grade                  | 80                            | 100                           | 120                        |
|----------------------------|-------------------------------|-------------------------------|----------------------------|
| Ambient Temperature Range. | -17°C to 25°C<br>(1° to 77°F) | 15°C to 35°C<br>(59° to 95°F) | Above 35°C<br>(Above 95°F) |

Use a non-compounded *AVIATION* oil:-

- *Aero Shell 100*
- *Mobil Red Band (EXXON Aviation Oil 100)*
- *BP Aviation Oil 100*

Use for 20-25 hours.

#### 3.5.2 Normal Operations

| Oil Grade                  | W80                           | W100                          | W120                       |
|----------------------------|-------------------------------|-------------------------------|----------------------------|
| Ambient Temperature Range. | -17°C to 25°C<br>(1° to 77°F) | 15°C to 35°C<br>(59° to 95°F) | Above 35°C<br>(Above 95°F) |

Use compounded Aero-Oil:

|                 |   |  |  |  |  |                  |                        |                 |
|-----------------|---|--|--|--|--|------------------|------------------------|-----------------|
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- Aero Shell W100
- Aero Shell 15W50 (for cooler climates)
- Mobil Aero 100 (SAE 50) (EXXON Aero Elite 100)
- BP Aero Oil D100/BP Multigrade Aero Oil D SAE 20 W 50
- Aero Shell W 100 Plus

The normal running oils are detergent and ashless dispersant types. See **MAINTENANCE SECTION 7**

### 3.6 Cooling System

- Free air cooled. Ensure that baffles are correctly fitted & located.
- The required pressure drop across the cylinders at 1.3 Vs in take off configuration is 4.3 cm (1.7") water gauge, minimum.

### 3.7 Operating Speeds and Limits

|                                  |                                   |
|----------------------------------|-----------------------------------|
| Maximum Speed                    | 3300 RPM                          |
| Maximum Continuous Speed         | 3300 RPM                          |
| Idle Speed                       | 900 RPM (set while engine is hot) |
| Oil Pressure – Normal Operations | Min 220 kPa (31 psi)              |
|                                  | Max 525 kPa (76 psi)              |
| – Idle                           | Min 80 kPa (11 psi)               |
| – Starting & Warm up             | Max 525 kPa (76 psi)              |
| Oil Temperature:                 | Min 15°C (59°F)                   |
|                                  | Max. 110 °C (244°F)               |
| Oil Continuous Temperature       | 80 - 100°C (176° - 212°F)         |
| Max. CHT (Climb)                 | 200°C (392°F)                     |
| Max Continuous CHT (Cruise)      | 180°C (356°F)                     |

(Read Cylinder Head Temperature – CHT – under the spark plug nearest to the exhaust on the hottest cylinder).

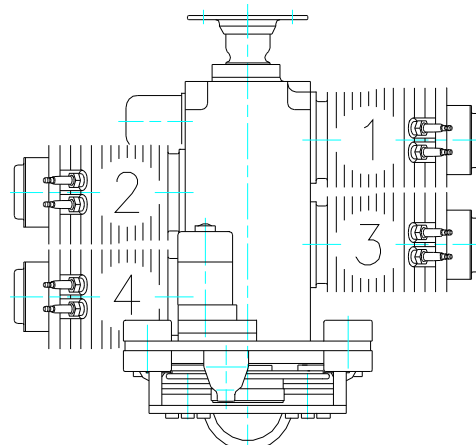
#### EGT Limits:

|                           |                                   |
|---------------------------|-----------------------------------|
| EGT (Mid-Range / Cruise): | Min 680° - 750°C (1256° - 1382°F) |
| EGT (Above 70% Power):    | Min 640° - 780°C (1184° - 1256°F) |

#### Note:

1. An EGT gauge not included as standard equipment on the Jabiru 2200 engine, though a system can be supplied as an option.
2. From Serial Number 710 in the maximum head temperature (climb) has been raised to 200 ° C and maximum at cruise at 180 ° C

### 3.8 Denomination Of Cylinders

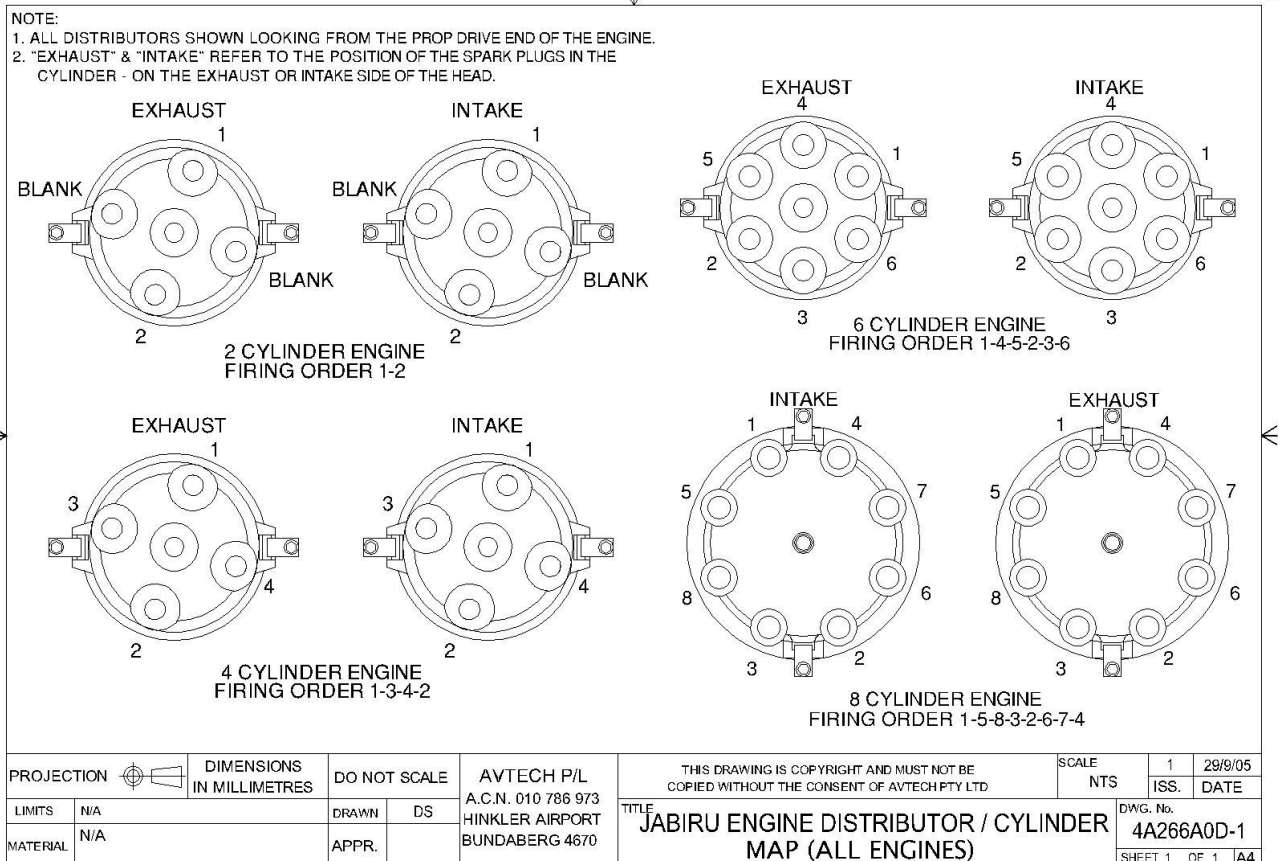


Drawing 9483093/1

**Figure 2. Cylinder Denomination & Firing Order**

Cylinder Firing Order: 1 – 3 – 2 – 4

### 3.9 Distributor Cylinder Map



**Figure 3. Distributor Cylinder Map**

### 3.10 PERFORMANCE CURVES

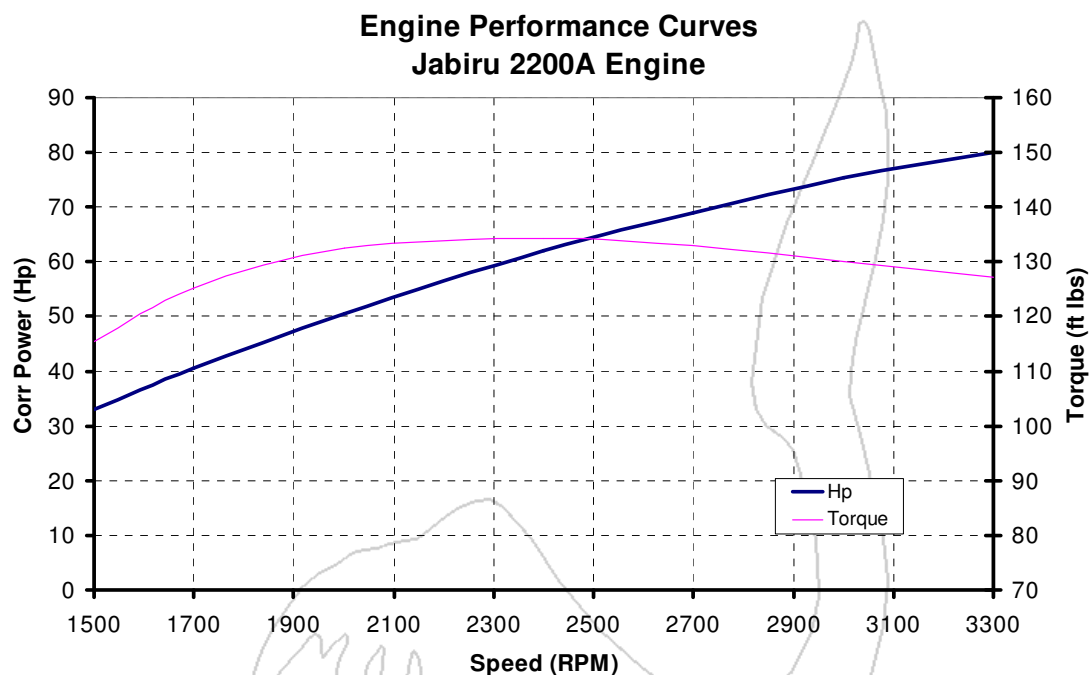


Figure 4 – Power Curve

## 4 OPERATING INSTRUCTIONS

To ensure that the engine operates reliably, carefully observe all of the operating & maintenance instructions.

### 4.1 Pre-Start Checks

- Daily Checks (See Paragraph 6.1)
- Move throttle position to FULL & check for ease of movement over the entire range.

### 4.2 Starting Procedure

|           |                            |
|-----------|----------------------------|
| Fuel Tap  | OPEN                       |
| Choke     | ON (in cold conditions)    |
| Fuel Pump | ON for 10 seconds then off |
| Throttle  | CLOSED to Stop             |
| Master    | ON                         |
| Ignition  | BOTH ON                    |
| Starter   | PRESS                      |

- Attention: Activate Starter for a max. 20 seconds, followed by a cooling period of 1 minute.
- When engine runs, adjust throttle to achieve smooth running at approximately 1200 RPM before Deactivating Choke.
- Check Oil Pressure has risen within 5 seconds - if not, shut down.
- It is stongly advised to close choke while cranking on first attempt. To much choke will cause over-fueling and loss of start.
- It is strongly recommended to crank the engine to obtain oil pressure following an oil change before attempting to start.

### 4.3 Warming Up Period, Ground Test

- Start the warming up period with the engine running at 1200 RPM. Continue at 2000 RPM depending on ambient temperature, until oil temperature reaches 15°C (59°F).
- Check the two ignition circuits at 2000 RPM. Note: - RPM with only one ignition should not drop by more than 100 RPM.

**DO NOT apply full power until CHT reaches 100°C (212°F)**

**DO NOT allow cylinder heads to rise above 150°C during ground running.**



#### 4.4 Take-Off

- Climb with the engine at maximum continuous power.
- Observe Oil & Cylinder Head Temperatures. Limits must not be exceeded ! (Max. continuous RPM at Full Throttle is 3300 RPM ).

#### 4.5 Engine Stop

- In normal conditions, cooling down the engine during descent & taxiing will permit the engine to be stopped by switching OFF the ignitions.

#### 4.6 Engine Stop and Start During Flight

- Reduce power to 2000 RPM to cool engine for 30 seconds, then to idle. Switch ignitions OFF.
- Starting procedure is the same as ground starting, without choke for a warm engine & with choke for a cold engine.

**Note:** Engine cools quickly with propeller stopped in flight. Choke will therefore normally be needed to restart.

## 5 Early Operation of Engine and Maintenance Tips.

## **JABIRU ENGINE NOTES**

### **NEW ENGINE NOTES:**

This engine has been ground run to a specific run in program and is ready for flight. It has been ***INHIBITED*** however if you intend to store for any length of time please refer to Para 6.9 of ***INSTRUCTION AND MAINTENANCE MANUAL***. Removing spark plugs and turning over will help periodically. Before initial start, oil engine 2.3 litres (with cooler) remove one plug per head activate starter to remove excess inhibitor. Once oil pressure is obtained replace plugs and continue start-up sequence.

The following are tips to ensure a long life.

1. Add 2 Litres (without oil cooler)  
2.3 litres (with an oil cooler)
2. Avoid prolonged ground running at elevated RPM. Engine can be over heated, remember air ducts are designed for in flight cooling. **NOTE:** remove ALL plastic bungs on engine. EXHAUST (4) CARBY (2) OIL VENT (1) FUEL PUMP (1)
3. ***DO TAKE OFF AT FULL POWER.***  
For the first few take offs climb at a higher airspeed. Reduce power at cross wind leg and shallow climb (lower nose).
4. ***VARY*** your RPM when flying.
5. Avoid high nose altitude continual climbs. The higher the climb out speed the better for engine cooling. No low speed high nose altitude climb outs.

Ring bed in is accomplished better at 75% power and above. Avoid heat build up. Monitor CHT and oil temps. Vary RPM. Initial temps will be elevated due to friction of a new engine.

6. **CIRCUIT WORK** is possibly a good sequence for initial run in work. Abbreviate circuits initially, step climb, climb shallow. No glide approaches. Gradually reduce power. Avoid sudden heating up and sudden cooling down.
7. **RETORQUE** heads as suggested on warning sheet.  
24 ft lb and .010" valve clearance all done cold. Subsequent torquing of around 22 to 24 seems to help in settling down head bolts and heads.  
After 25 hours 20ft lb is sufficient.
8. Don't **BABY YOUR ENGINE** but monitor carefully CHT and oil temp initially especially during the first few hours of operation.  
Remember engines need to be flown. They are designed for this purpose. Do not use full power before CHT reaches 100°C.
- The purpose of breaking in an engine correctly is to ensure a long reliable life. All moving parts need freeing up especially piston rings to cylinder walls. This is best accomplished when the greatest B.M.E.P. (Break Mean Effective Pressure) occurs. That is at 75% power and above. Careful monitoring by the pilot is needed during this initial period to ensure long life of the engine and its components.
9. Failure to operate at realistic power settings could be detrimental to engine condition.
10. Do not use any type of automotive oil. These oils have not been blended for the purpose of air cooled aero engine operation and will be detrimental to its operation.
11. Use **AVGAS 100 LL** or the highest Octane Mogas above 95 containing lead. Engines not shimmed on the cylinder base should only be run on **AVGAS**.
12. When you change oil from the "run in" type to the "normal" oil at or around 25 hrs replace oil filter. You may want to cut the filter open for inspection. It is usual in Jabiru engines to find a small amount of aluminium but

**definitely no metal. If bearing metal is present contact the Jabiru Service Department.**

**With ignition and master OFF and throttle closed turn the prop by hand and observe engine for odd noises or heavy movements. Check for regular compressions, if irregular firstly check tappet adjustment. Operation with incorrectly adjusted tappets with result in damage to valves, valve seats, guides and overhead gear.**

## 5.1 HEAD TORQUING/VALVE SETTING

**Your tension wrench should be accurate. It should have had some method of calibration. They can vary to manufacturers claims.**

***24 ft lb on head bolts  
.010" on valve clearance cold***

*After 25 Hrs 20 Ft Lb is  
sufficient*

**Early heads require a 1/2 inch ring spanner and 3/16 allen key. Later engines use a 9/16 socket and 3/16 allen key or blade screwdriver.**

**Easiest method of setting valves requires you to pull the prop until exhaust valve on No.1 cylinder is fully depressed. Note the O'clock position. Rotate prop 360° and adjust valve. (This puts the cam and lifter at 180° to the peak lift). Repeat for each valve. Heads and valves should be done at least twice at 5 hrs and 10 hrs on a new engine.**

**Remember to go over exhaust cap screws. They are fitted with shakeproof washers to prevent loosening.**

## 5.2 AIR DUCTS

**Periodic research and development does cause at time changes to occur. Ones supplied need to be fitted. See instructions manual.**

## 5.3 FILTERS

**Inspections down the track are a must for fuel and air filters. Conditions will dictate when changed. The air box has a rubber flap to give partial inspection of filter.**

## 5.4 SPARK PLUGS

**NGK D9EA are recommended.**

**Plug gap of around .022" to .024". Remember plugs are installed at around 8 ft lb or given 1/2 turn after contact with head.**

**18mm Plug spanner used.**

## 5.5 COMPRESSION TESTING

**Condition of compression can be done by a compression gauge. Wide open throttle, engine warm. Turn over on starter. Below 90 PSI would indicate removal of head and possibly cylinder.**

### PRESSURE DIFFERENTIAL TEST (Leak down)

**This is a much better test for condition of rings, bore, head sealing and valve. Engine in warm to hot condition. This is the normal test used in aviation requiring specific equipment for the job. Pressure input of 80 PSI; a second gauge reads the differential when supplying 80 PSI. This is done with piston on TDC on the firing stroke. Prop needs to be restrained. The differential cut off is 80/60. Problems can be better identified eg.**

- **BLOW BY (CRANKCASE VENT) - RINGS, BORE SEAL**
- **LEAKING FROM CARBY - INTAKE VALVE SEAL**
- **LEAKING FROM EXHAUST - EXHAUST VALVE SEAL**
- **HEAD LEAK - HEAD GASKET OR HEAD TO CYLINDER SEAL**

**Correction work can then be carried out.**

## 5.6 COIL GAP

**Best done with a piece of plastic or thin card of a thickness .010". Cut into a strip approximately 15mm wide.**

**Place between magnets on flywheel and coil. Check both sides, that is each coil to each magnet (4 checks).**

## 5.7 STARTING

**A warm idle of around 950 RPM will automatically create the right starting environment. Idle set screw may have to be adjusted.**

**Carby has been factory set. Normal start requires throttle closed, that is the idle criteria has just cracked the butterfly in the throat body and also apply choke. The engine is difficult to start if throttle is cracked open somewhat.**

**As the engine is cranked the choke should be pushed off. Engine should fire. The choke is only used for a cold start. Prolonged cranking with choke will only “flood” the intake system making starting difficult. Should this occur, leave or clear the system on full throttle momentarily (mags off). Cranking speed if too slow (poor battery or starter fault) will prevent magneto operation. Jump starting (with care) will point towards poor battery condition or faulty alternator charging. Further testing would then be required in these areas to identify the problem.**

## 5.8 POSSIBLE PROBLEMS

**See “Trouble Shooting” Section of 10.0 of Maintenance Manual.**

**It is unusual for a problem to occur however a few of “common” type are listed.**

### 1. LOW OIL PRESSURE

**A sudden drop of pressure usually is caused by a small piece of foreign matter being lodged under the relief valve. Simply remove oil filter and cooler adaptor (if fitted). Remove matter by depressing plunger or removing relief mechanism. Replace and check operation by ground run. If low pressure persists the problem will need further investigation.**

**We strongly recommend the fitting of an approved oil cooler to Jabiru engines.**

### 2. FLICKING OF OIL PRESSURE GAUGE

**It is not uncommon for the Jabiru engine to display flicking of the needle pointer for brief periods but still within normal operating regions. If it continues, check continuity of sender lead and or possible sender change. (We are assuming oil level is OK).**

### 3. CHT

**Possible not reading can be cause by a break in the sender wires or incorrect polarity. High readings can result with poorly centred sender under the spark plug. Normal cruise CHT should not exceed 150°C and climb must not exceed 175°C. Air ducts supplied at present give results below these figures.**

**4. RPM**

Tacho's may need adjustment when a new engine is fitted.

Inducted magnet sender units require coil gap to flywheel gear teeth of .014" or .35mm. Tacho's using this sender require a "pot" adjustment, access through outside of case.

Tacho's using the magneto as sender requires a sequence of operation for correction. This information can be faxed if required.

Later engines use 2 tags under the ring gear in conjunction with the induction sender.

**5. MAGNETO CHECKS**

Possible causes of abnormal drop could be loose leads, faulty leads, rotor buttons, coil gaps, spark plugs.

**6. ROUGH CYLINDER RUNNING**

Check plugs/valve clearance and the induction system for looseness.

**7. CARBY BREATHER**

Later model carbies have a brass fitting for venting. This is easily connected via a clear plastic hose to a fitting screwed into the carby heat box. On earlier carbies we supplied a kit for this purpose as no internal carby fitting was installed.

**8. LIMITATIONS**

Warranty notes follow.

Jabiru recommend TBO of 1000 hrs. From s/n 710 TBO has increased to 2000 hours with Top End at 1000. A guaranteed fixed price

overhaul exists through the Jabiru factory. Warranty is 200 hours or 12 months which ever comes first.

**CAREFULLY READ MANUALS**



## 6 CHECKS ON ENGINE & INSTALLATION

### 6.1 Daily Checks

- Ensure free movement of throttle & choke cables.
- Check Oil Level, replenish if necessary. Oil level should be between the MAX & MIN marks - but must never be below the MIN mark. Before long periods of operation, ensure that the level is at least at the mid position. Difference in the oil quantity between MAX & MIN mark is 0.5 litres (0.528 US Quarts). See section 6.3 for first 25 hours of Operation. Check oil level by screwing in cap fully before withdrawing. Overfilling is detrimental to the engine.
- Check security of spark plugs, leads & electrical connections.
- Check lubrication & fuel system for leaks.
- Check exhaust system for security & leaks.
- With Ignition & Master OFF, and throttle closed, turn propeller by hand & observe engine for odd noises or heavy movements. Check for regular compression. If irregular, firstly check tappet adjustment (see para 7.8).

### IMPORTANT

**Prior to pulling through the propeller by hand, both ignition circuits & the Master must be switched OFF, the brakes applied, throttle closed & the cockpit attended by a trained person.**

### WARNING

**A hot engine may fire with the ignition/s switched OFF.**

### CAUTION

**Continued operation with incorrectly adjusted tappets will result in damage to valves, valve seats, valve guides & overhead gear. Head torque and valve adjustment at 5/10 hrs from initial start up.**

- \*Prior to takeoff - follow the Starting & Warm Up procedure, observe the engine behaviour & throttle response.
- Check temperatures & pressures. Conduct a short ground test at full power (a few seconds) (consult aircraft Flight Manual).

**NOTE:** Any prolonged running at full power can cause engine damage while on the ground. Continual operation with leaky valves or partially sealing exhaust gaskets will result in engine damage.

## 6.2 Periodic Checks

- After the initial 25 hours, check in accordance with para. 6.3.
- After 50 hours of operation, check in accordance with para. 6.4 & thereafter after each 50 hours of operation.
- After 100 hours of operation, check in accordance with para. 6.5 and thereafter after each 100 hours of operation.
- At TBO, overhaul in accordance with para. 6.7.

## 6.3 Check After Initial 25 Hours

Details of specific operations are shown in Chapter 7 "Maintenance".

- Remove engine cowlings, check engine mounts.
- Thoroughly check engine for missing or loose bolts, nuts, pins, etc., & for abrasions.
- Check induction and exhaust flange for loose bolts.
- Check safety wires, cooling air ducts & baffles, ignition wiring & hose connections.
- Oil Change        2.0 litres. Use normal aviation running oil.
- Change oil filter.
- Inspect old filter.
- Retorque cylinder head bolts (20 ft lbs) in diagonal pattern
- Check tappet clearance and adjust as necessary (refer Para 7.8). (0.010" cold inlet and exhaust)
- Check exhaust system, check exhaust cap screw tensions and gasket condition.
- Check fuel system for leaks & abrasion.
- Check wiring for damage & for tightness.
- Check prop bolt torque.
- Engine test run.  
Observe starting, warm up & acceleration behaviour to maximum RPM (10 seconds max)

Check temperatures & pressures.  
Engine stop.

#### 6.4 50 Hours Check

Details of specific operations are shown in Chapter 7 "Maintenance".

- Conduct the items shown under 25 Hour Check at para. 6.3. (include oil and filter change)

#### 6.5 100 Hours Check

Details of specific operations are shown in Chapter 7 "Maintenance".

- Conduct the items shown under 25 Hour Check at para. 6.3. (include oil and filter change)
- Renew spark plugs, if necessary.

#### 6.6 200 Hours Check

Details of specific operations are shown in Chapter 7 "Maintenance".

- Treat as per 100 Hour check, with the addition of a flywheel bolt check.

#### 6.7 **TBO**

Details of specific operations are shown in Chapter 7 "Maintenance".

- Engine Overhaul, in accordance with Service Bulletins
- Conduct the items shown under 25 Hour Check at para. 6.3.
- Check clearance on throttle valve shaft. If radial clearance exceeds 0.5mm (0.020"), repair carburettor.
- The overhaul work must be carried out to Jabiru specifications at an approved aeronautical service facility or by an approved Jabiru Service Centre.
- If necessary, changes to the TBO Limit due to operational experience, will be announced by Jabiru in a Service Bulletin.

## 6.8 Engine Preservation

The following procedures assume that the engine is installed in a Jabiru airframe. For other aircraft types, refer to the manufacturer's service manual. If the engine is not fitted to an airframe, ignore those items referring to the airframe.

**Note:** Failure to store the engine for a long period of time, in excess of 90 days, without taking the preventative measures as outlined in the manual will affect warranty claims.

## 6.9 Flyable Storage

- Flyable storage is defined as a maximum of 30 days non-operational storage.
- Ensure that the engine has been stopped by turning off the fuel valve, thereby draining all fuel from the carburettor bowl.
- Every 7th day the propeller should be rotated through 5 revolutions, without running the engine.
- Leave the propeller in the horizontal position to ensure even distribution of liquids in the wood. If left in the vertical position, liquids will drain to the lower tip resulting in an unbalanced propeller.

### CAUTION

**Ensure that the Master and Ignition Switches are OFF!**

## 6.10 Returning Engine to Service

After flyable storage, returning the engine to service is accomplished by performing a thorough pre-flight inspection. Ensure all protective covers are removed.

## 6.11 Temporary or Indefinite Storage

Temporary storage is defined as aircraft in non-operational status for a maximum of 90 days.

Treat as for flyable storage (see Paragraph 6.9), plus:

- For temporary storage, fill fuel tank with correct grade of fuel (to prevent moisture accumulation).
- For indefinite storage, drain fuel tank, ensure carburettor bowl is empty by running engine with fuel valve off until it stops or by draining bowl.

Then:

- a) Disconnect spark plug leads and remove spark plugs from each cylinder.

- b) Using a spray atomiser, spray the oil through the spark plug hole with the piston down, then rotate until both valve are open and respray to coat the induction and exhaust system.
- c) When all cylinders are treated leave prop horizontal and retreat each cylinder.

**NOTE:** Use Shell Aero fluid 2UN (MIL-C-6529C Type 1) Corrosion Preventive Concentrate or similar engine preservative.

### CAUTION

**Ensure that the Master and Ignition Switches are OFF!**

- d) 3. Install spark plugs and connect leads.
- e) 4. Seal exhaust pipes. Attach a red streamer to each seal. DO NOT seal fuel tank breather.
- f) Attach a warning placard to the propeller stating that vents and breathers have been sealed and prop should not be turned.
- g) The engine must not be started with the seals in place.

## 6.12 Inspection During Storage

- a) Generally inspect airframe and clean as necessary.
- b) Inspect the interior of at least one cylinder through the spark plug hole for corrosion at least once a month.
- c) If, at the end of the 90 day period, the aircraft is to be continued in non-operational storage -- repeat Steps 1-5 above (most will only need to be checked).

### 6.12.1 Long term storage

- Flood the cylinders / sump with preventative oil.
- Turn over periodically and assist by removing 1 plug per head.

## 6.13 Returning Engine to Service

After temporary storage, the procedures for returning the aircraft to service are as follows:

- Check battery and install.
- Check carburettor air filter and service if necessary.

- Remove warning placard from propeller.
- Remove materials used to cover openings.
- Remove, clean and gap spark plugs.
- Drain all oil before rotating. While spark plugs are removed, rotate propeller several revolutions to clear excess preservative oil from cylinders.

### CAUTION

**Ensure that the Master and Ignition Switches are OFF!**

- Install spark plugs -- torque to 11 Nm (8 ft/lbs).
- Check fuel filter -- replace if necessary.
- If returning to service after indefinite storage, fill fuel tank with correct grade of fuel.
- Check fuel tank and fuel lines for moisture and sediment. Drain enough fuel to eliminate any moisture and sediment.
- Check fuel tank breather is clear.
- Perform a thorough pre-flight inspection.
- Start and warm engine.

## 6.14 Operation in Winter

- It is recommended to carry out an engine service prior to the start of the cold season.
- For selection of oil, consult the table of lubricants at Paragraph 3.5.
- Follow the following advice for operation at extremely low temperatures:

## 6.15 Carburettor Icing

It is important to distinguish between two kinds of icing:

- 1) Icing due to water in fuel
- 2) Icing due to high air humidity

1) Water in fuel will accumulate at the lower parts of the fuel system & can lead to freezing of fuel lines, filters or jets. Remedies are:

- Drain, using fuel tank water drain.
- Install a generously sized water separator.
- Ensure that fuel lines do not permit the accumulation of water.

- Prevent condensation of humidity, ie avoid temperature differences between the aircraft & fuel
- Ensure fuelling without traces of water. If in doubt, use a chamois as a filter.

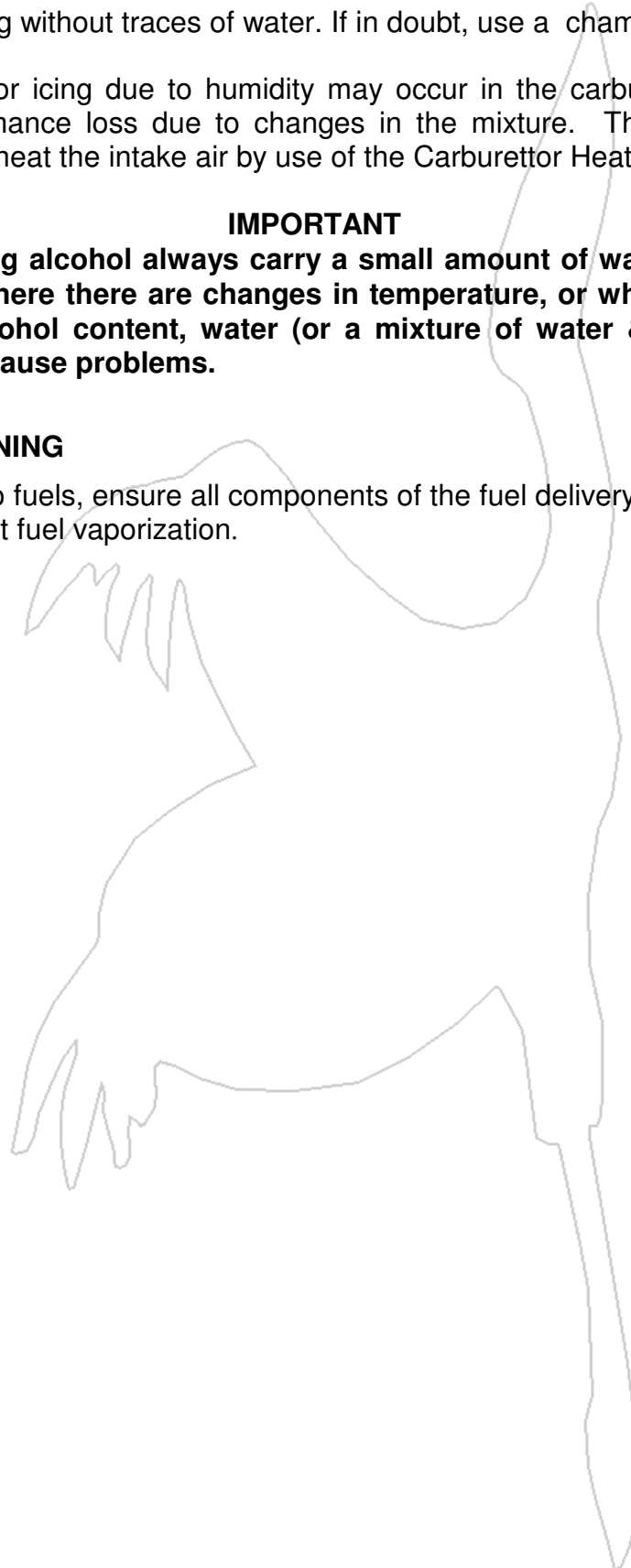
2) Carburettor icing due to humidity may occur in the carburettor venturi & leads to performance loss due to changes in the mixture. The only effective remedy is to preheat the intake air by use of the Carburettor Heat Control

### **IMPORTANT**

**Fuels containing alcohol always carry a small amount of water in solution. In situations where there are changes in temperature, or where there is an increase in alcohol content, water (or a mixture of water & alcohol) may settle & could cause problems.**

#### **6.15.1 WARNING**

When using auto fuels, ensure all components of the fuel delivery system are cooled to prevent fuel vaporization.





## 7 MAINTENANCE

### 7.1 Lubrication System

Oil Change, Oil Filter Change, Visual Check for Leaks.

- Drain the oil while engine is still warm.
- Change the oil filter at every 50 hours inspection.
- Fill with oil. Capacity is 2.3 litres (2.43 US Quarts)
- Check oil level. The MAX mark must not be exceeded.
- Use only registered brand oils meeting the specification detailed in Para. 3.5.
- DO NOT drain the oil cooler during a normal oil change. The cooler holds only a small amount of old oil which has negligible effect on the new oil. Taking the hoses on & off the cooler can prematurely age the oil lines and lead to hoses slipping off the cooler.

### 7.2 Air Intake Filter

- Clean filter by removing from the intake housing & blowing compressed air against the direction of the intake flow.
- For operation in heavy dusty conditions, clean air filter at shorter intervals than recommended for normal conditions.
- A clogged filter will reduce engine performance as well as promote premature engine wear.

### 7.3 Carburettor Adjustment

- Idle stop screw is 7mm against throttle lever.
- Open idle mixture screw approximately 1-1/4 turns, fine adjust for a smooth idle. The determination of the main jet is carried out on a dyno at 107 ft above Mean Sea Level.

**IMPORTANT**  
**Check & oil carburettor joints & linkage**

### 7.4 Compression Check

- Measure compression using a compression tracer. Readings are taken with fully open throttle valve at engine oil temperature between 30 & 70 degrees C (90 to 160 degrees F).

- If readings are below 6 bar (90 psi) a check of the pistons, cylinders, valves & cylinder heads must be undertaken.

Alternatives:

- Pressure loss or leakage tester eg SUN or BOSCH tester; max. allowable pressure loss is 25%.
- Checking by commonly used pressure difference method; place orifice of 1 mm ID and 3mm length between the two pressure gauges. This will give the same result as with the above instrument. Max. pressure drop is 25%.

## 7.5 Spark Plugs

- Do not use steel or brass brushes for cleaning & never sandblast plugs.
- Clean with plastic brush in a solvent.
- Check electrode gap & if necessary, adjust to 0.55 - 0.6mm (0.022" - 0.024") by carefully bending the electrode. Recommended Plugs: NGK D9EA use suitable anti-seize on thread.

### IMPORTANT

**Only tighten spark plugs on cold engine & only to the torque values shown in para. 7.5 using appropriate anti-seize compound**

Note: When plugs are removed from a warm engine, the following are indicators:

|                                     |   |
|-------------------------------------|---|
| <i>Light Coloured to Brown :-</i>   | Plug & calibration is correct.  |
| <i>Velvet Black:-</i>               | Mixture too rich. Check choke. Insufficient air intake. Check for clogged air filter. |
| <i>Oily, Glossy Coating:-</i>       | Misfiring. Too much oil in combustion chamber.<br>Worn cylinder & piston rings.       |
| <i>Whitish with Melt Droplets:-</i> | Mixture too lean. Leaking valves.   |

## 7.6 Exhaust System

- Visual check for damage, leaks & general condition.
- Operation of the engine with partially sealing exhaust gaskets will cause engine damage.

## 7.7 Bolts and Nuts

- Check for tightness, re-torque if necessary (see para. 5).

## 7.8 Tappet Adjustment

Tappets must be adjusted to:

|         |                 |
|---------|-----------------|
| Inlet   | 0.254mm (.010") |
| Exhaust | 0.254mm (.010") |

Adjust the tappets when the engine is cold. Head torqued to 24 ft.lb. when cold. Carry out this adjustment after five hours of operation and again after ten hours of operation. At the 25 hour inspection this is done again. There is no need to back off studs when checking, we are trying to check if movement has occurred and seal of cylinder to head is in tact. Bolts in the exhaust area may move on checking.

Subsequent checking at 20 – 22 Ft Lb is sufficient.

### CAUTION

**Continued operation with incorrectly adjusted tappets will result in damage to valves, valve seats, valve guides & overhead gear.**

## 7.9 Tachometer and Sender

Many apparent engine problems can be caused through inaccurate tachometers. Where engine performance is observed to be outside limits, the tachometer should be checked against a calibrated instrument. Tachometer sender gap is 0.4mm (0.016"). When using two tags behind flywheel sender must have at least 60% covering by the tag. Ensure both tags are equal distance from sender.

## 7.10 Additional Checks

- Check engine for ease of starting. Conduct idle test run. Checklist of engine inspection.

### 7.10.1 Propeller

- Check spinner, spinner flange, screws, prop, tension, spinner and prop tracking.

### CAUTION:

**The maintenance schedules given in this manual are designed for engines which use a propeller and propeller flange approved by Jabiru Aircraft Australia.**

**Jabiru Aircraft strongly recommend owners using non-approved parts follow a system of extra preventative maintenance checks to monitor their engine – particularly propeller installation and flywheel attachment.**

**Warranty issues may also result.**

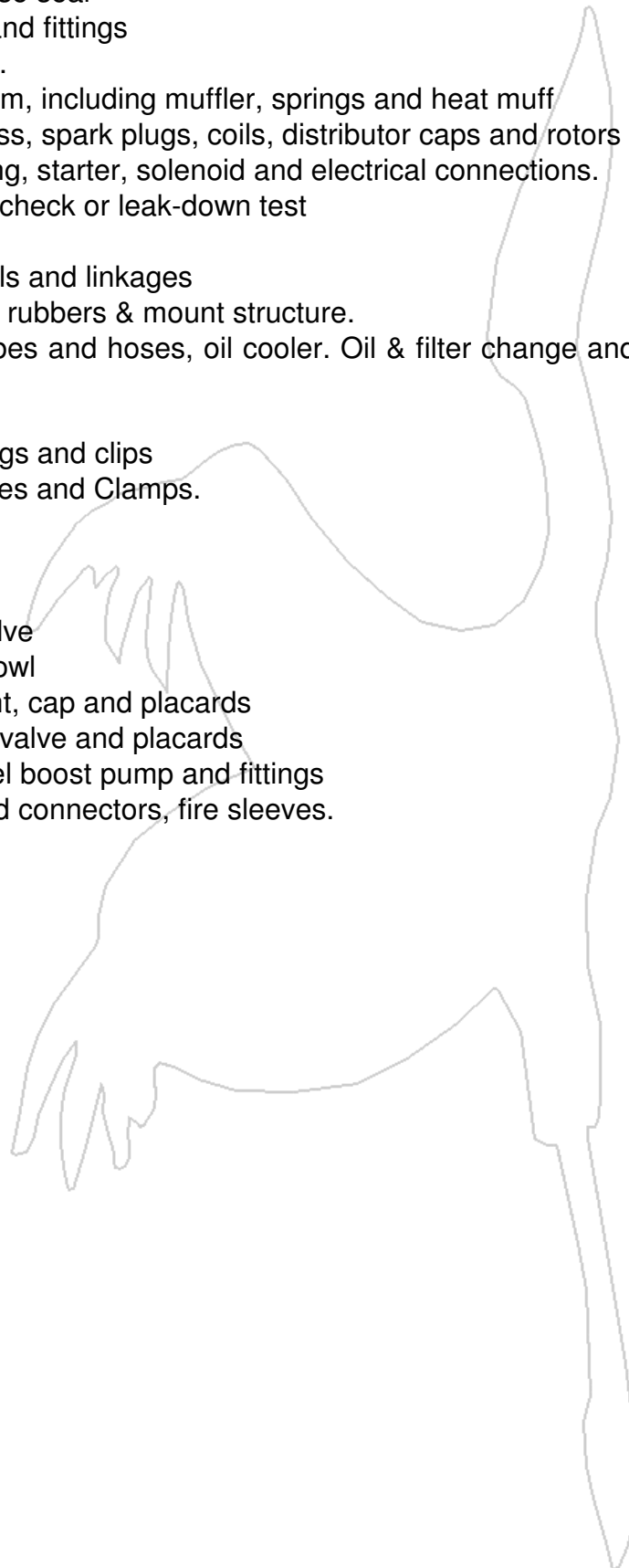
### 7.10.2 ENGINE

- Carburettor, carburetor heat system, carburetor attachment
- Air filter
- Head bolt tension
- Valve clearances

- Engine baffles and air ducts
- Front crankcase seal
- Hoses, lines and fittings
- Intake system.
- Exhaust system, including muffler, springs and heat muff
- Ignition harness, spark plugs, coils, distributor caps and rotors
- Electrical wiring, starter, solenoid and electrical connections.
- Compression check or leak-down test
- Fuel pump
- Engine controls and linkages
- Engine mount rubbers & mount structure.
- Oil system tubes and hoses, oil cooler. Oil & filter change and filter inspection if due.
- Firewall
- Engine cowlings and clips
- All SCAT Hoses and Clamps.

### **Fuel System**

- Fuel filter
- Fuel drain valve
- Carburetor bowl
- Fuel tank vent, cap and placards
- Fuel shut-off valve and placards
- Electronic fuel boost pump and fittings
- Fuel lines and connectors, fire sleeves.



## 7.11 Engine Inspection Chart

**Note:**

**READ ALL INSPECTION REQUIREMENTS PARAGRAPHS PRIOR TO USING THESE CHARTS.**

|  |   | Annual Inspection |   |    |   |
|--|---|-------------------|---|----|---|
|  |   | Each 200 Hours    |   |    |   |
|  |   | Each 100 Hours    |   |    |   |
|  |   | Each 50 Hours     |   |    |   |
|  |   |                   |   |    |   |
| PROPELLER  |   |                   |   |    |   |
| 1  | Spinner   |                   | * |    | * |
| 2  | Spinner Flange  |                   | * |    | * |
| 3  | Spinner screws  |                   | * |    | * |
| 4  | Propeller   |                   | * |    | * |
| 5  | Propeller bolts/nuts - Tension  |                   | * |    | * |
| 6  | Spinner / Prop Tracking   |                   | * |    | * |
| ENGINE & ENGINE COMPARTMENT  |   |                   |   |    |   |
| Check for oil, fuel exhaust & induction leaks, then clean entire engine & compartment before inspection. |   |                   |   |    |   |
| 1  | Check flywheel screw tensions <sup>1</sup><br>To 24lb.ft for 5/16" Screws <sup>2</sup><br>To 15lb.ft for 1/4" Screws <sup>3</sup> |                   |   | *4 | * |
| 2  | Carburetor air filter –<br>check & replace if required.   | *                 | * |    | * |
| 3  | Engine baffles and air ducts  |                   |   |    | * |
| 4  | Cylinders   |                   |   |    | * |
| 5  | Crankcase & front crankcase seal  |                   |   |    | * |

<sup>1</sup> Note: if any screws rotate they must be removed, screw & crank threads cleaned then re-installed with fresh Loctite 620 (use no more than the size of a match head). Do not remove all flywheel screws at once or loss of valve timing may occur.

<sup>2</sup> Engines from S/No. 436 to 2102 were assembled using 18 lb.ft.

<sup>3</sup> Engines from S/No 0 to 436

<sup>4</sup> 200-hourly check of flywheel bolts only required on engines WITHOUT flywheel dowels.

| Annual Inspection  |  |   |   |   |
|--------------------|--|---|---|---|
| Each 200 Hours     |  |   |   |   |
| Each 100 Hours     |  |   |   |   |
| Each 50 Hours      |  |   |   |   |
| 6                  | Fuel hoses, lines and fittings   |   | * | * |
| 7                  | Intake and exhaust systems   |   | * | * |
| 8                  | Ignition harness, distributor caps & rotors                            |   |   | * |
| 9                  | Spark plugs  | * | * | * |
| 10                 | Compression check or leak-down check                                   |   | * | * |
| 11                 | Electrical wiring  |   |   | * |
| 12                 | Fuel pump  |   | * |   |
| 13                 | Engine controls and linkages   |   | * | * |
| 14                 | Engine mounts, mount structure   |   | * | * |
| 15                 | Starter, solenoid and electrical connections                           |   | * | * |
| 16                 | Coils and electrical connections                                       |   | * | * |
| 17                 | Carburettor heat system  |   | * | * |
| 18                 | Throttle and linkage   |   | * | * |
| 19                 | Carburettor  |   | * | * |
| 20                 | Oil system tubes and hoses   |   | * | * |
| 21                 | Firewall   |   |   | * |
| 22                 | Engine cowlings and clips  |   | * | * |
| 23                 | Exhaust system – including muffler, springs & gaskets (where equipped) | * |   | * |
| 24                 | Head bolt tension & valve clearance adjustment                         |   | * | * |
| 25                 | Oil & filter change  | * |   | * |
| 26                 | SCAT hose condition  |   | * | * |
| <b>Fuel System</b> |  |   |   |   |


**Annual Inspection**
**Each 200 Hours**
**Each 100 Hours**
**Each 50 Hours**

|   |   |  |   |  |   |
|---|---|--|---|--|---|
| 1 | Fuel filters, drain valves, carburetor bowl |  | * |  | * |
| 2 | Electronic fuel boost pump and fittings     |  |   |  | * |
| 3 | Fuel lines and connectors                   |  | * |  | * |
| 4 | Fire sleeves                                |  | * |  | * |



## 8 SERVICE & REPAIR

### 8.1 Overhaul and TBO

- These are carried out only by the manufacturer, Jabiru Aircraft Pty Ltd or by a specifically approved Jabiru Engine Service Centre (contact Jabiru for details).
- The engine must be sent in a complete state, with logbook, to Jabiru (or the Approved Service Centre) after reaching the TBO limit.
- Changes to the TBO due to operational experience will be advised by Jabiru through Service Bulletins.
- It is recommended that engines prior to s/n 710 have full overhauls at or before 1000 hours. Very early engines were at 500 hours. Engines after s/n 710 have full overhauls at 2000 hours with a Top End being done at around 1000 hours. This is also subject to certain conditions. Overhauls are influenced by condition.

#### 8.1.1 Engine Overhaul

An overhaul must present the engine with a life of 2000 hrs. During that life at 1000 hours a top end overhaul is undertaken. Overhauls are dependant on condition. Particular attention must be paid to Torque Specifications, Prescribed sealants, Tolerances and Maximum Allowable Clearances.

MPI testing is carried out on the crankshaft, camshaft, prop drive and conrods. Any necessary upgrades to represent as close as possible a current engine must be undertaken.

Parts that are replaced are:- Pistons, gudgeons and circlips, rings, main bearings, conrod bearings, thrust bearings, conrod bolts, all orings gaskets and seals, fuel pump, valves, induction hoses, rubber oil feed tees, spark plugs, rotors, head bolts, intake/exhaust studs, fuel hose, oil filter, oil hoses, flywheel bolts, prop flange bolts, woodruff key (oil pump). Cylinder, heads and dizzy shafts are usually replaced at a full overhaul.

All parts are cleaned, measured, inspected and recorded in the prescribed build sheets in the manual. Engine assembled according to maintenance section and build sheet directions. The RUN IN procedure also to be used as outlined in this section. If the run in is performed on the ground large air cooling ducts and a large oil cooler must be used. Engine can be run in, in the aircraft in the air with caution, using the run in program as the guide. Section 5 of an engine must be understood. Initial performance data must be recorded.

#### 8.1.2 Top End Overhaul

Top end overhauls are performed on engines after S/N 710 and dependant on condition.

Parts replaced are:- Conrod bearings, pistons, gudgeon pins and circlips, rings, valves, all relevant orings and gaskets, head bolts, spark plugs, induction hose joiners, rotors, relevant capscrews and fuel line. Other parts for inspection, measure and clean or replace are oil pump, starter, alternator, fuel pump, coils, ignition leads, oil seals, induction and exhaust capscrews, carby, heads and barrels.

Engine assembled and test run ( Run In Program)



## 8.2 ENGINE REMOVAL PROCEDURE

| <u>No.</u> | <u>Operation</u>   | <u>Tools</u>  |
|------------|--|---|
| 1          | Remove Spinner and Propeller   | Phillips Screwdriver<br>7/16" Socket<br>7/16" Spanner |
| 2          | Remove Carby Heat hose from Hot Box Muffler                                      | Screwdriver   |
| 3          | Remove Air Inlet Hose from Carburettor and blank off Carburettor and Air Cleaner | Screwdriver<br>2 Plugs                                |
| 4          | Disconnect Throttle Cable  | Long Nose Pliers                                      |
| 5          | Disconnect Choke Lever   | Long Nose Pliers                                      |
| 6          | Remove Oil Breather Line   | Screwdriver   |
| 7          | Remove Fuel Line from Fuel Pump and plug Fuel Line and Fuel Pump                 | Screwdriver<br>1/4" Plugs                             |
| 8          | Remove starter Motor Cable from Solenoid   | 7/16" R/OE  |
| 9          | Disconnect Earth at Battery  | 10mm R/OE   |
| 10         | Remove Oil Pressure Gauge Lead   | -   |
| 11         | Remove Oil Temperature Gauge Lead  | -   |
| 12         | Remove Hourmeter Lead  | Screwdriver   |
| 13         | Remove Cylinder Head Temperature Gauge Lead                                      | -   |
| 14         | Remove Exhaust Gas temperature Gauge Lead  | -   |
| 15         | Remove Tacho Lead  | -   |
| 16         | Remove Left and Right Ignition Coil Leads  | -   |
| 17         | Remove Muffler Assy  | 3/16" Ball End Allen Key                              |
| 18         | Undo Engine Mount Bolts  | 7/16" Tube Socket<br>7/16" Spanner                    |
| 19         | Remove Engine from Engine Mount Frame  | -   |

### 8.3 ENGINE DISASSEMBLY

| <u>No.</u> | <u>Operation</u>   | <u>Tools</u>                               |
|------------|--|--|
| 1          | Mount Engine on an engine stand in vertical position                     |  |
| 2          | Remove Spark Plugs   | Spark Plug Socket                          |
| 3          | Remove Oil Pressure Sensor   | 17mm Spanner                               |
| 4          | Remove Oil Filter  | -  |
| 5          | Oil Pressure Valve may be removed by removing snap ring (1)              | Snap Ring Pliers                           |
| 6          | Extract Valve Seat (1), Valve (1), Spring (1)                            | -  |
| 7          | Remove Oil Pump Assembly (Housing, Rotor, Stator & Backing Plate)        | 7/32" Allen Key                            |
| 8          | Remove 4 capscrews in front oil seal carrier and break seal on crankcase | 3/16" Allen Key                            |
| 9          | Remove Fuel Line and Carburettor   | Screwdriver                                |
| 10         | Remove Fuel Pump, Gasket and Push Rod                                    | 7/32" Allen Key                            |
| 11         | Remove Distributor Cap Clamps, Caps and Rotors                           | 5/32" Allen Key                            |
| 12         | Remove Starter Motor   | 7/32" Allen Key                            |
| 13         | Remove Alternator Mount  | 7/16" Socket<br>5/16 Ring O/End<br>Spanner |
| 14         | Remove Ignition Coils from Alternator Mount                              | 3/16" Allen Key                            |
| 15         | Remove Flywheel  | 3/16" Allen Key                            |
| 16         | Remove Gearbox Cover   | 3/16" Allen Key                            |
| 17         | Remove Engine Mount Plate  | 7/32" & 3/16" Allen<br>Key                 |
| 18         | Remove Crankshaft Timing Gear  | -  |
| 19         | Remove Lower Induction Pipes   | Screwdriver                                |

- |                              |  |  |
|------------------------------|--|--|
| 20                           | Remove Sump  | 3/16" Allen Key  |
| 21                           | Remove Exhaust and Induction Pipes   | 3/16" Allen Key  |
| 22                           | Remove Tappet Covers   | 3/16" Allen Key  |
| 23                           | Remove Rocker Blocks and Push Rods<br>(From S/N 225 rocker blocks were not used)   | 1/2" Spanner<br>1/2" Socket<br>5/32" Allen Key<br>1/4" Allen Key |
| 24                           | Remove Cylinder Head   | 7/32" Allen Key<br>1/2" Socket                                   |
| 25                           | Remove Push Rod Tubes  | -  |
| 26                           | Remove Cylinder  | 7/16" Crowsfoot  |
| <i>Removal of Piston</i>     |  |  |
| 27                           | Remove 1 Wrist Pin Circlip   | Long Nose Pliers   |
| 28                           | Remove Wrist Pin (keep pin matched with its piston)  | -  |
| 29                           | Remove Piston Rings (keep rings matched with its piston)   | -  |
| 30                           | Repeat Operations 27-30 on other pistons if required.  | -  |
| <i>Crankcase Disassembly</i> |  |  |
| 31                           | Remove main Stud Nuts on Front 2 studs   | 7/16" Socket   |
| 32                           | Double check that every bolt is removed and sump,<br>gear case engine mount plate, oil pump and<br>Front Oil Seal Carrier is removed | -  |
| 33                           | Tap Crankcase slightly and open Crankcase by<br>removing half Crankcase  | Plastic Mallet   |
| 34                           | Remove both Crankcase Halves of Crankshaft and<br>Remove Cam Shaft   | -  |
| 35                           | Remove Valve Lifters (8)<br><i>Note - Keep valve lifters matched to each<br/>Crankcase half.</i>                                     | -  |
| 36                           | Remove Thrust Washers (front and rear)<br>- 2 without tang on right half case<br>- 2 with tang on left half case                     |  |



Note - visually check for marks or scratches  
- keep matched to their seats

37 Remove Main Bearing Inserts (12)

Note - visually check for marks or scratches  
- keep matched to seats

38 Remove Crankcase dowels. Remove O'Rings and Discard

Note - do not attempt to remove studs as they  
are loctited in with 620 loctite

39 Remove Oil Gallery Plugs and Oil Pressure Sender.  
Remove Oil Relief Valve.

#### Cylinder Head Disassembly

40 Remove Valve Springs and Valves

Valve Spring  
Compressor

41 Remove Push Rod Circlips and O'Rings

Internal Circlip Pliers

#### Gearbox Cover Disassembly

42 Remove Distributor Shafts

43 Remove Oil Seals

#### Crankshaft & Camshaft Disassembly

44 Remove Bolts from Conrod

1/4 allen key/10mm  
Socket  
3/8 Ratchet  
Hot Air Gun Rags

Note - Rods will need to be heated  
- Big End Bolts are Discarded on Removal

45 Remove Rod Bearing Inserts

Note - visually inspect for marks and scratches  
- keep matched to their seats

46 Remove Camshaft Gears. Discard Camshaft Bolts.  
Remove welsh plug in rear of camshaft.

## 8.4 2200 ENGINE OVERHAUL

### 8.4.1 Inspection and Assembly of Sub-Assemblies

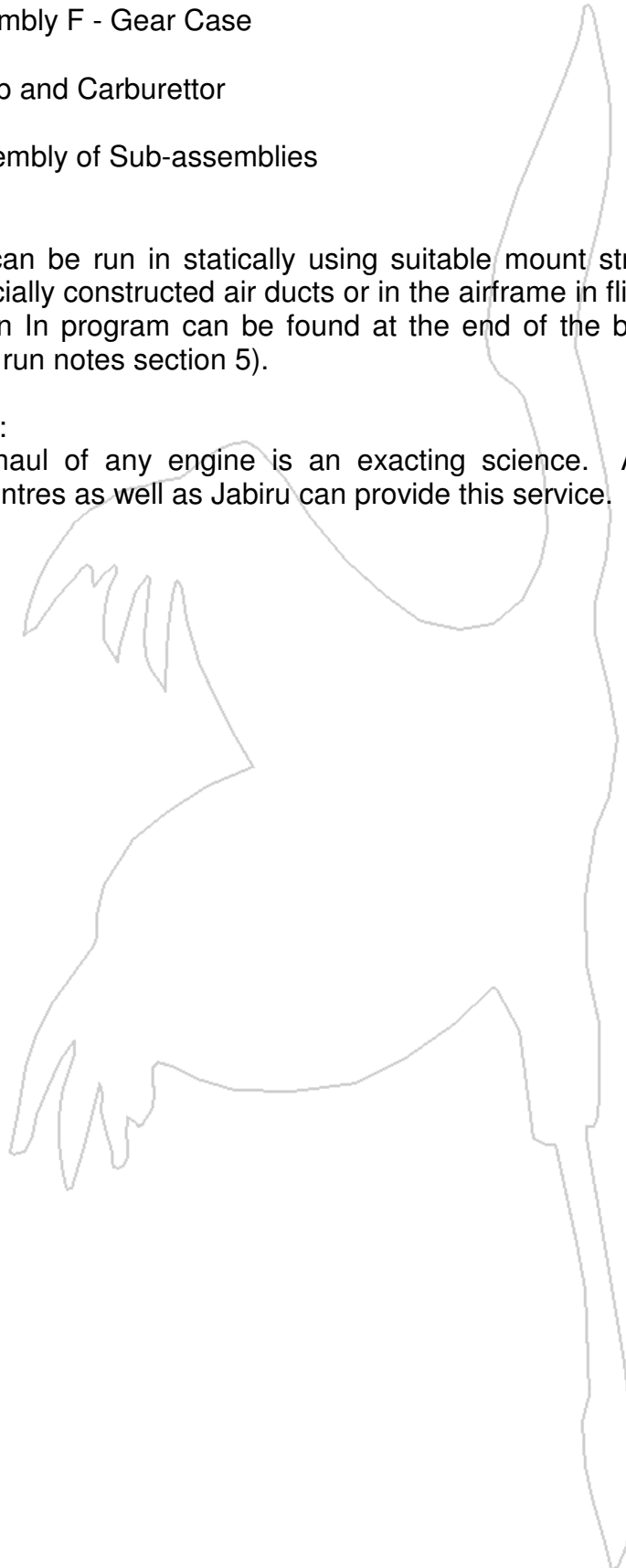
1 Sub-Assembly A - Crankshaft, Prop Mount and Con Rods.

2 Sub-Assembly B - Crankcase and Camshaft.

3 Sub-Assembly C - Pistons, Cylinders and Cylinder Heads.

4 Sub-Assembly D - Sump and Oil Pump

- 5 Sub-Assembly E - Flywheel, Ignition Coils and Alternator
- 6 Sub-Assembly F - Gear Case
- 7 Fuel Pump and Carburettor
- 8 Final Assembly of Sub-assemblies
- 9 Run In:  
Engines can be run in statically using suitable mount structure and very large specially constructed air ducts or in the airframe in flight with extreme care. Run In program can be found at the end of the build sheets (see also early run notes section 5).
10. Overhaul:  
The overhaul of any engine is an exacting science. Approved Jabiru service centres as well as Jabiru can provide this service.



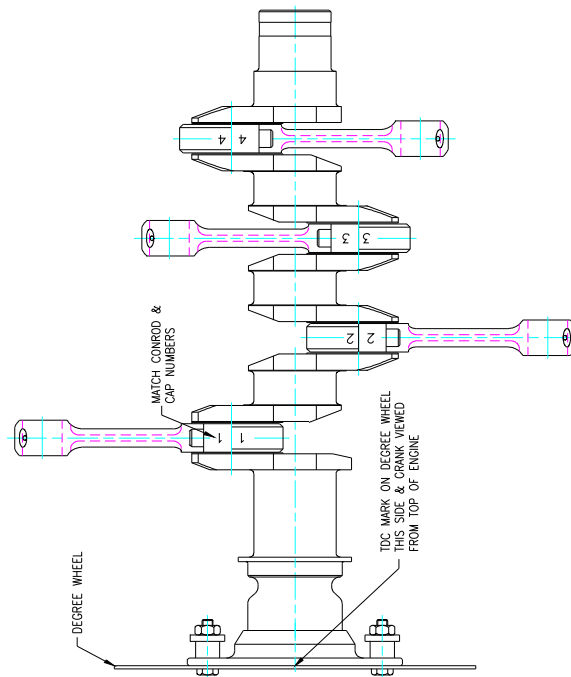
## 8.5 Jabiru 2200 Component Inspection and Assembly Procedure

### 8.5.1 Sub-Assembly A - Crankshaft, Prop Mount and ConRods

- A1
- 1) Remove welsh plugs from crankshaft
  - 2) Clean conrods and crankshaft  
*Note - Make sure all old loctite is removed from threads*
  - 3) Strip paint off prop flange and crankshaft.
  - 4) Magnet particle inspect crankshaft , prop flange, cam and conrods.
- A2 Measure crankshaft and record in Build Sheet.
- A3 Inspect Oil Holes for Cleanliness and insert new front welsh plugs (2 off) and rear welsh plug.
- A4 Inspect prop flange. Paint prop flange and end of crankshaft to prevent rusting.
- A5
- 1) Magnet particle Inspect conrods for cracks.
  - 2) Fit bearings with a light smear of bearing blue on back of shells.
  - 3) Torque Caps to 18 lbs (with new bolts)
  - 4) Measure Big End Bearings. Record in Build Sheet.
  - 5) Measure Piston Pin Bore. Record in Build Sheet. Check CRUSH.
  - 6) Remove caps and check bearing back contact (Must be at least 90%)
  - 7) Remove all blue and thoroughly clean surfaces. Refit bearing shells.
  - 8) Clearance as close to 0.05mm is recommended.
- A6 Temporarily mount prop flange to crankshaft and mount to stand vertically. (Flange is removed later to fit front oil seal)
- A7 and
- 1) Spray conrod bolts and conrod bolt threads with Loctite 7471 primer allow to dry.
  - 2) Mount conrods to crankshaft. Use plenty of oil on journals. Closest to prop flange is number 1. Pins to prop drive. Use Loctite 620 on rods and bolts torque up to 24.0 NM (18 ft/lbs)
  - 3) Wipe off excess locktite at bases of cap screws on the end cap.

*Note: Alloy rods are not reused but replaced with steel type*

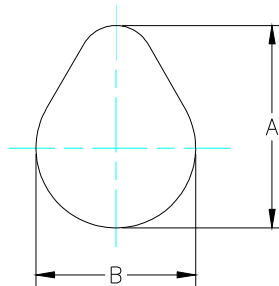




Drawing 9429063/2 CRANKSHAFT & CONROD ASSEMBLY

Figure 5- Drawing 9429063

- A8
- 1) Clean and visually inspect camshaft.
  - 2) Measure journal diameter.
  - 3) Measure lobe lift.

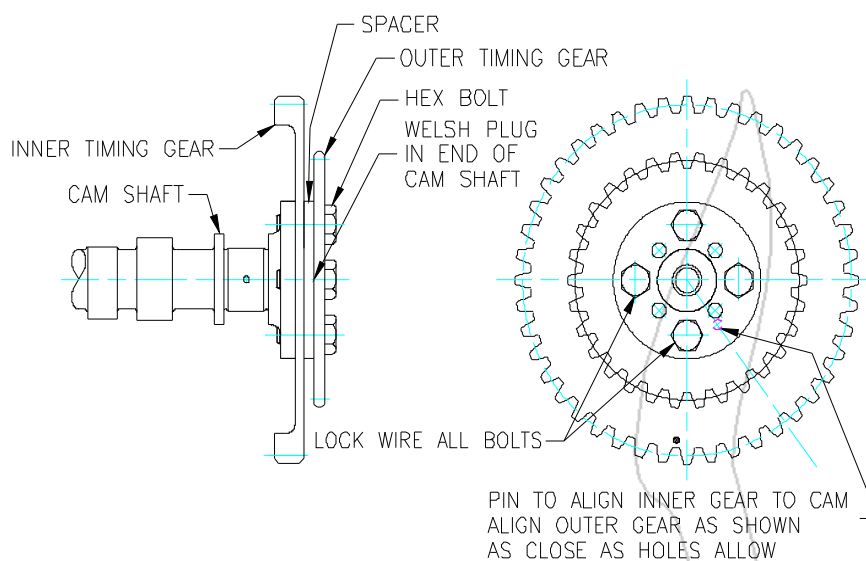


$$\text{CAM LIFT} = A - B$$

Drawing 9433064/1 CAM LIFT MEASUREMENT

Figure 6 - Drawing 9433064

- 4) Inspect fuel pump lobe for wear.
- 5) Spray camshaft rear gears and bolts with Loctite 7471 Primer.
- 6) Fit gears and new bolts. Use Loctite 620 torque to 8 ft lbs & Lock Wire.
- 7) Fit end welsh plug.



**Figure 7 - Drawing 9448064**

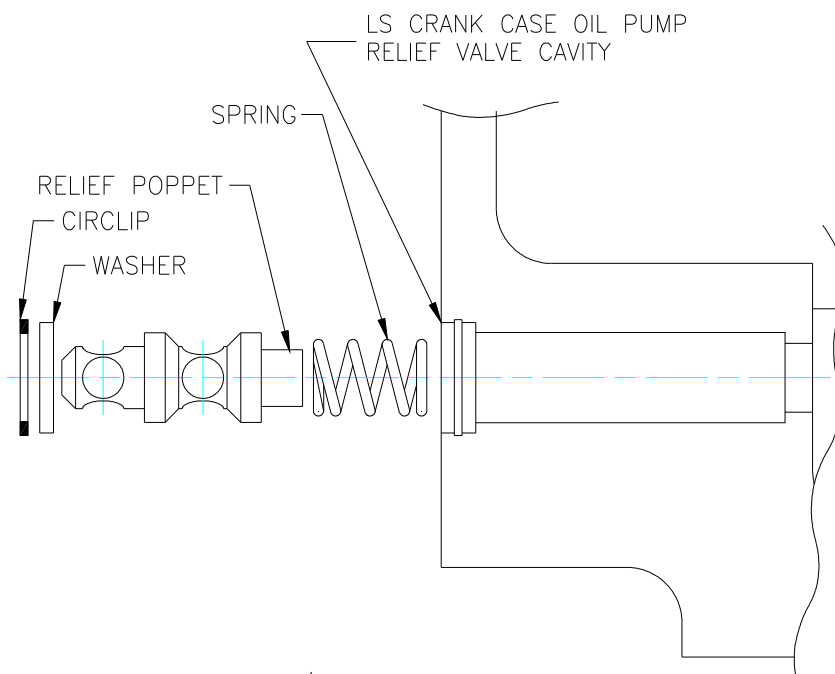
A9 Stage Inspection by 2nd person of Sub-Assembly

## 8.6 JABIRU 2200 Component Inspection and Assembly Procedure

### 8.6.1 Sub-Assembly B – Crankcase and Camshaft

- B1 1) Clean crankcase, remove old Loctite and clean oil galleries.  
2) Visually inspect crankcase.
- B2 Fit inner stud O Rings with a small smear of rubber grease or relevant lubricant.
- B3 Inspect all studs for tightness threads and stretching
- B4 1) Check oil suction pipe for security, replace oring.  
2) Check conrod welsh plugs (They don't need to be removed for overhaul. If removing a conrod in the field the plug has to be removed first to gain access to conrod bolts.) Steel conrods are removed from cylinder direction.
- B5 Fit oil relief valve assembly, oil pressure sender and front plug/hourmeter switch.

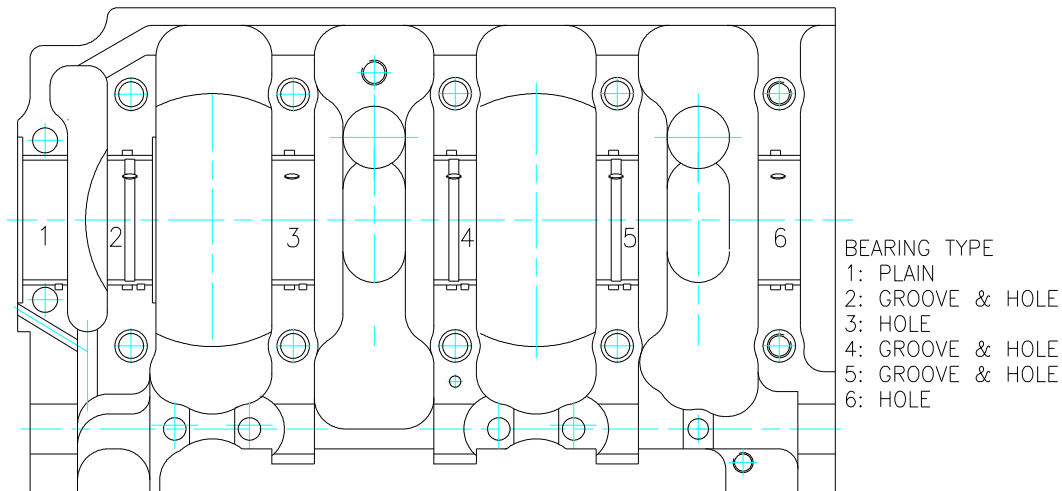
Relief washer should have a bevel to accept plunger (relief poppet) To increase oil pressure 1 or 2 AN4 washers can be placed behind the relief spring.



Drawing 9435064/1 OIL RELIEF VALVE ASSEMBLY

**Figure 8 - Drawing 9435064**

- B6 1) Lightly blue bearing shells and one crankcase half.  
2) Fit bearing shells.

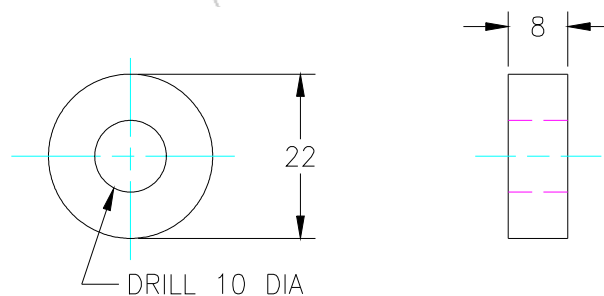


Drawing 9441064/1 BEARING POSITION &amp; TYPE

Figure 9 - Drawing 9441064

**NOTE: MAIN BEARINGS supplied now contain**

- B7 1) Make spacers as per drawing. only grooved type. These are used on all 6 locations



Drawing 9442064/1 SPACER FOR MEASURING CRANKCASE

Figure 10 - Drawing 9442064

- 2) Join crankcase halves.
- 3) Put through studs in case.  
*Note the cylinders by themselves can be used as spacers. It is just a bit harder to measure the main tunnels.*
- 4) Fit old nuts and torque to 40 NM (30 ft/lbs) in two stages.

B8 Measure main tunnel and cam tunnel and record in build sheet.

- B9
- 1) Disassemble.
  - 2) Remove shells and check back contact (must be at least 90%).
  - 3) Check mating surface on crankcase.
  - 4) Remove all blue and thoroughly clean surfaces. Refit bearing shells.

B10 Measure cam lifters and bores and record on build sheet. Check Lifter face for excessive scuffing. Oil and refit. Use a small amount of Molybdenum disulphide grease on lifter faces.

B11 Check cam shaft end float in both halves. Record in build sheet.

B12 Check crankshaft end float in both halves and record in build sheet.

## 8.7 Jabiru 2200 Component Inspection and Assembly Procedure

### 8.7.1 Sub-Assembly C - Pistons, Cylinders and Cylinder Heads

#### Cylinder Head Clean Up

- C1
- 1) Clean oil off heads.
  - 2) Dip heads in a cold dip solvent (i.e. Redik dkt Degreaser - Decarboniser - Paint stripper) as per manufactures instructions to remove all carbon deposits.
  - 3) Note. The heads can be bead blasted, but care must be taken.
  - 4) Wire buff valves.
  - 5) Clean all other parts.

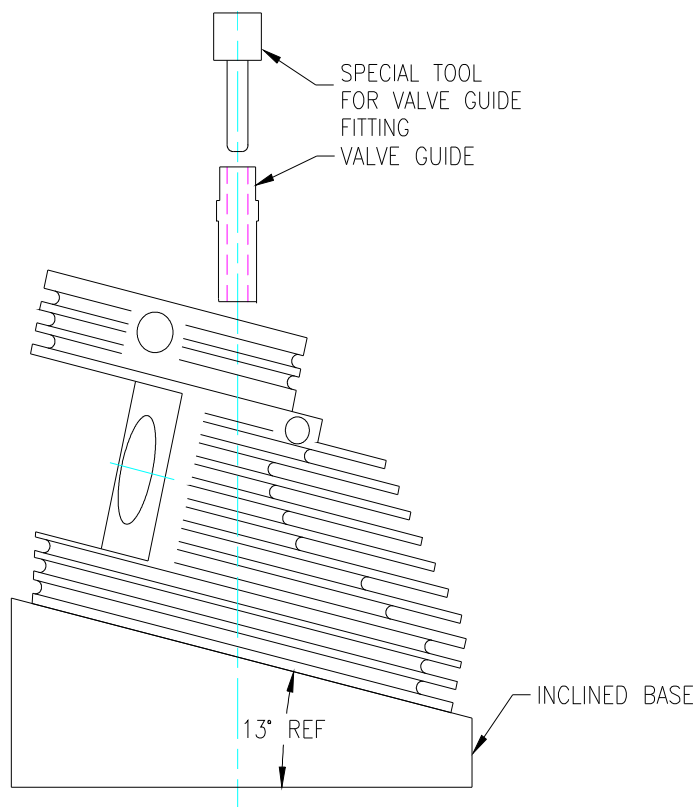
#### Cylinder Head Inspection and Repair

- C2
- 1) Inspect cylinder head for flatness. If sealing surface show signs of leakage it may need refacing. When refacing, remove only the minimum to clean up.
  - 2) Measure valve guide wear.
  - 3) If wear is excessive remove guides. Measure valve guide bore in the head. The new guide must have at least 0.05mm (0.002") interference on the outside diameter.

*Note - The standard guide has no dimples in top  
(P.N.4518064) In  
(P.N. 4519064) Ex*

*0.05mm (0.002") has one dimple  
0.05mm (0.004") has two dimples  
0.05mm (0.006") has three dimples*

- 4) Press guides back in.  
*Note - Oil hole must go up*
- 5) Special Note Guides: Guides are installed at 7.05mm bore diameter. Guides from 7.08mm can cause oil usage in the engine (max. valve stem to guide clearance is 0.12mm).



Drawing 9446064/1 VALVE GUIDE REPLACEMENT

Figure 11 - Valve Guide Replacement

- 5) Mount head up and cut seats as per drawing

**Note -** When cutting seats cut enough to clean up but try to remove as little as possible. This will give maximum number of overhauls on heads. The above diagram gives general valve seat cut geometry. Changes have occurred to production engines. S/N: 01 – 208 used above geometry. S/N: 209 – 743 used wider contact seat area. Engines 745 + used seat width cut from approved dwg. 4797004(I) and 4790004(E). Above valve diameters change S/N: 189 on.

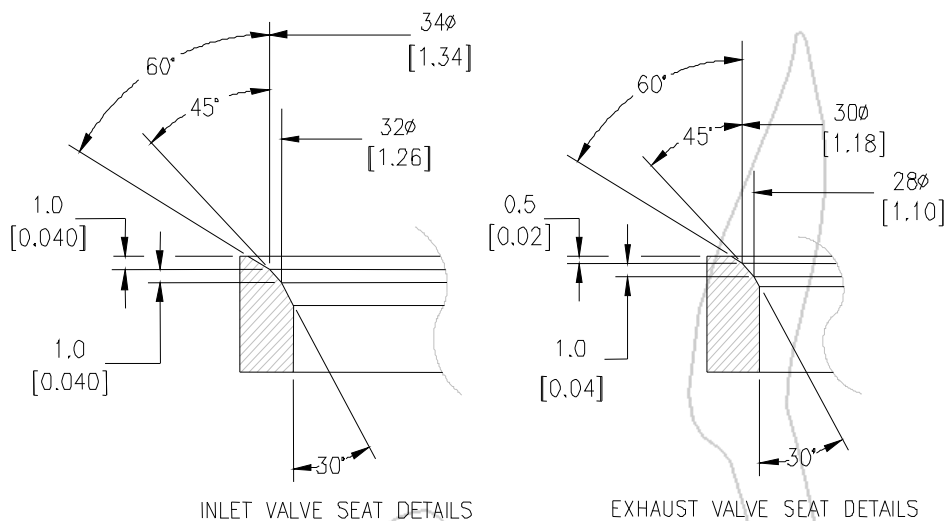
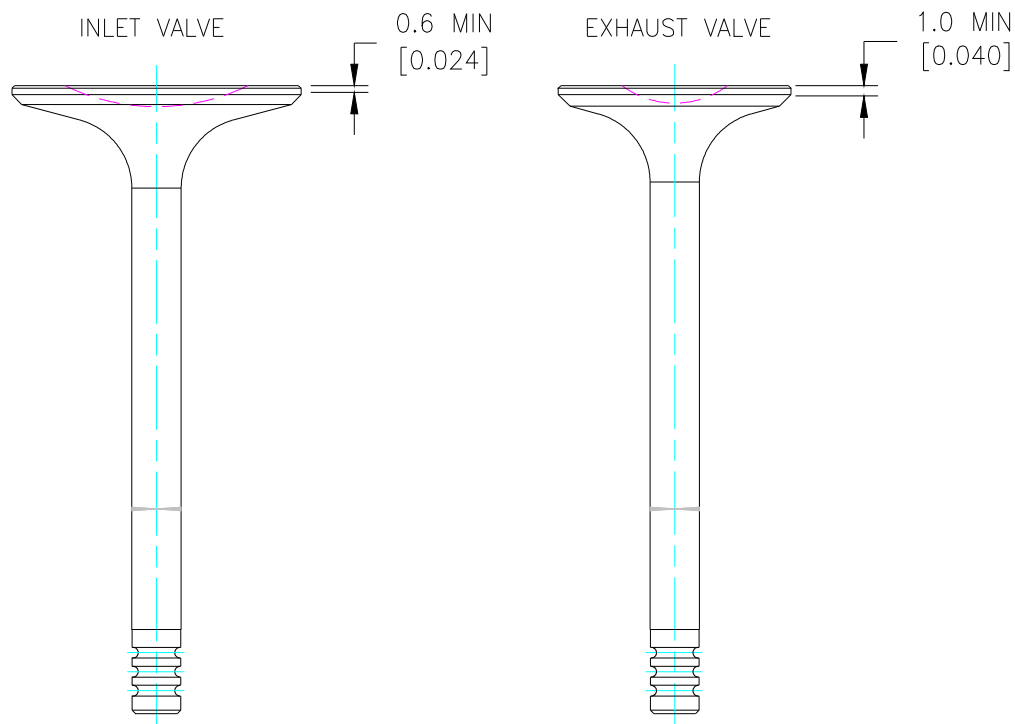


Figure 12 – Valve Seat Details

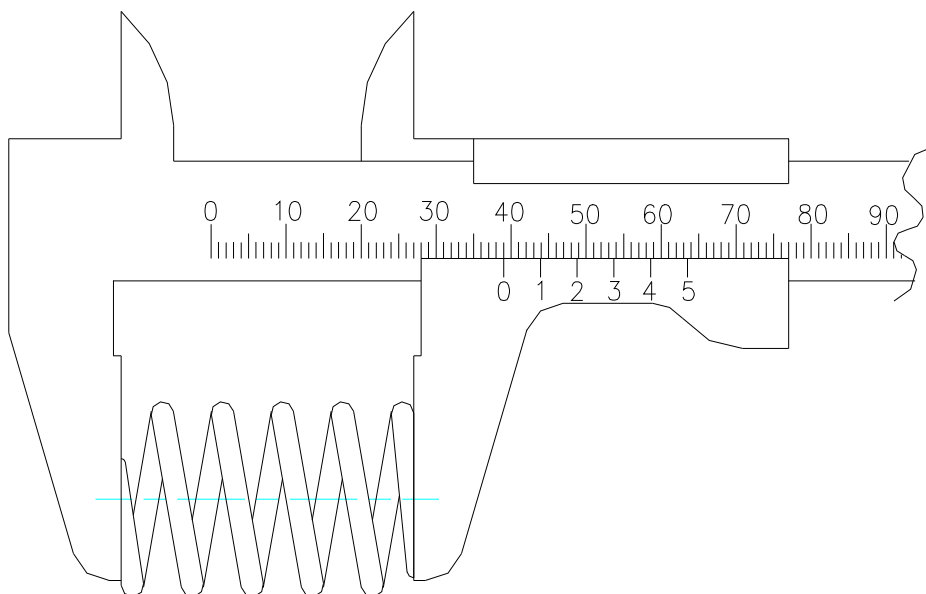
- 6) Use ground and serviceable valves to check seat contact.  
*Note - If machine has a vacuum tester, vacuum test valves.*
- 7) Heads of valves must be above the level of the combustion chamber. If they are below, the valve seats and valves must be replaced.
- 8) Measure valve stem for wear. Record in build sheet if within limits or replace valves.
- 9) Grind valve seats. If margin is too small, replace valves.



Drawing 9444064/1 VALVE DETAILS

Figure 13 - Drawing 9444064

- C3
- 1) Clean heads and valves ready for assembly.
  - 2) Measure valve springs. Replace if they are shorter than service limits.  
40.0mm – 41.8mm (1.580" - 1.688")



Drawing 9443064/1 VALVE SPRING MEASUREMENT

Figure 14 - Drawing 9443064

- 3) Inspect valve spring seats, valve spring retainers and valve collets. Replace any that show signs of wear.
- 4) Lube valves well. Install valves, valve spring seats, springs, retainers and collets.

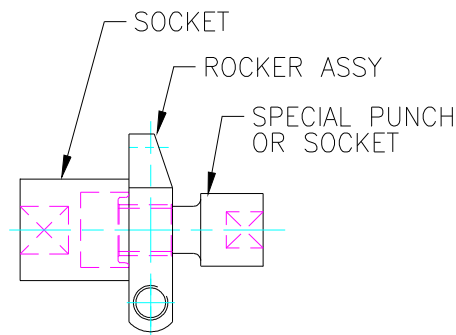
*Note: All engines have the same spring retainer on exhaust and inlet valve.*

- C4 Install Push Rod Tube O'Rings with a small amount of rubber grease.

#### Rocker Assembly

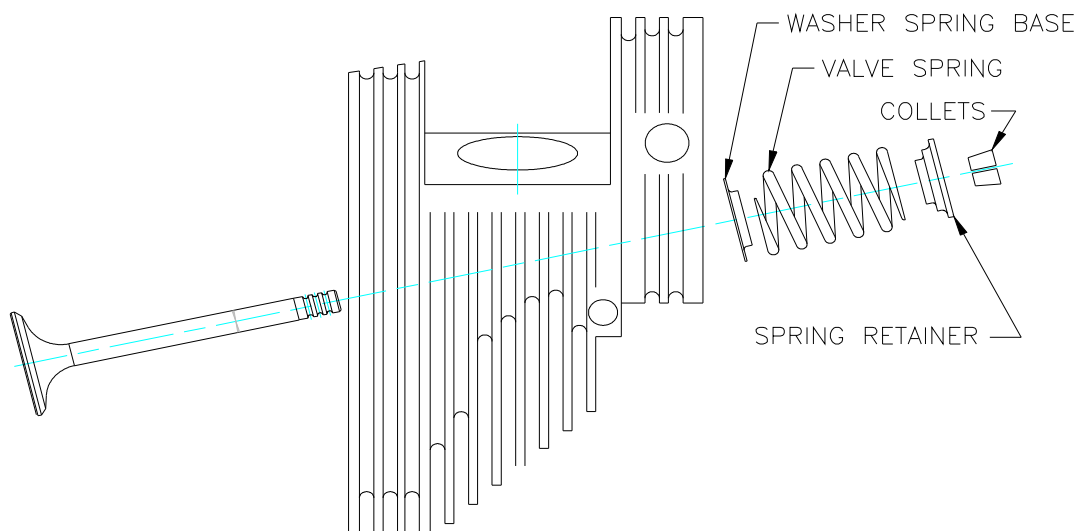
- C5
- 1) Disassemble and clean off all carbon.
  - 2) Inspect rocker tip for wear. Replace any with excessive wear.
  - 3) Measure bushes and record on build sheet.
  - 4) If over service limits, replace.





Drawing 9449064/1 ROCKER BUSH REMOVAL

Figure 15 – Rocker Bush Removal



Drawing 94470641/2 VALVE ASSEMBLY

Figure 16 – Drawing 9447064

- 5) Clean off old Loctite from rocker bore.
- 6) Press in Bushes.

*Note - The width of the bushes is critical for the correct amount of end float. so they must be pushed right home.*

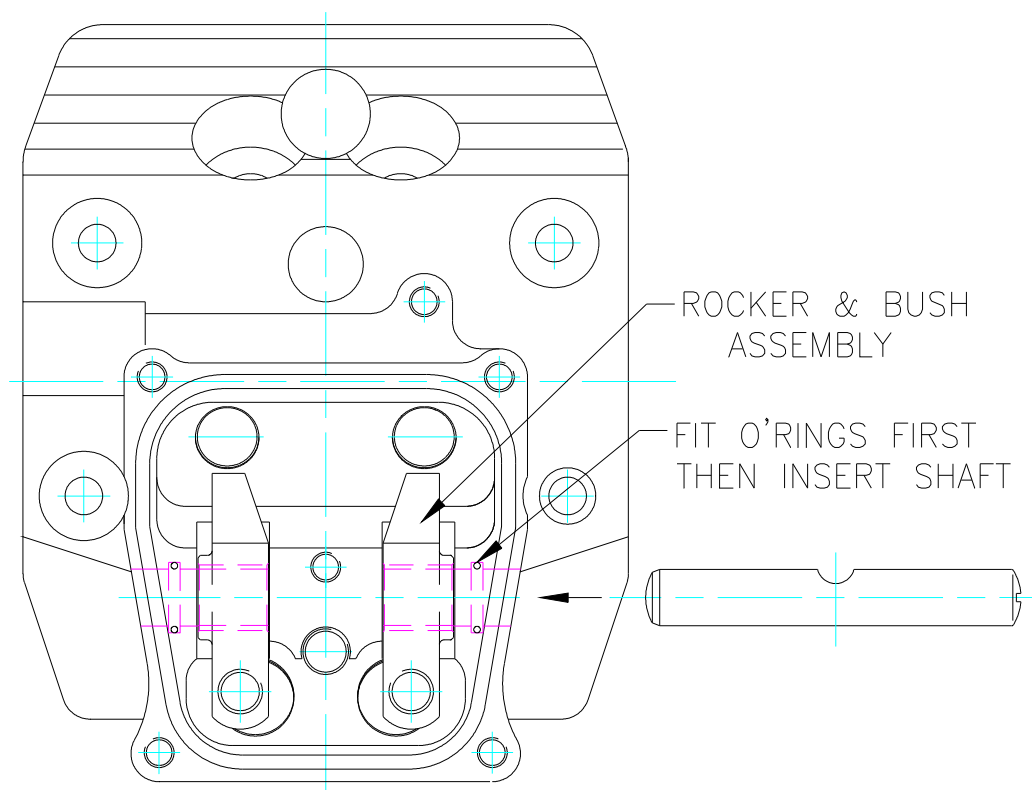
*- By putting the outer (wider flange) bushes to the left makes a left bush and vice versa. Always keep rockers matched i.e. left stays left and right stays right.*

- 7) Put a small amount of Loctite 290 (wickin) on bushes to lock them on.
- 8) Ream bushes with an adjustable ream to dia 12.03mm - 12.05mm (0.474" - 0.475").
- 9) Measure rocker shafts for wear and inspect for scratching, scuffing etc. Record in Build Sheet.
- 10) Inspect rocker blocks.

*Note - If the clamp capscrew comes loose it will let the shaft rock.*

*Any blocks like this must be replaced.*

- 11) Assemble rocker block.
- 12) Actions 5 – 8 refer to earlier bushed rockers up to engine 118. Engines from 189 onwards use either 12mm or 15mm wide GLASIER type bush. These were pushed in to fit (friction fit, no compound used to lock). Bush join goes to top of rocker.



Drawing 9449064/1 ROCKER ASSEMBLY

**Figure 17 - Drawing 9449064**

- 13) Put capscrew through clamp hole and check rocker end float. Must be 0.1 to 0.4mm (0.004" to 0.010"). Face off outside bushes to adjust.
  - 14) Clean up, oil well and reassemble.
- Cylinders**
- C6**
- 1) Clean and visually inspect.
  - 2) Mount cylinders in honing machine by base and hone with Sunnen mm 33-J85 Stone

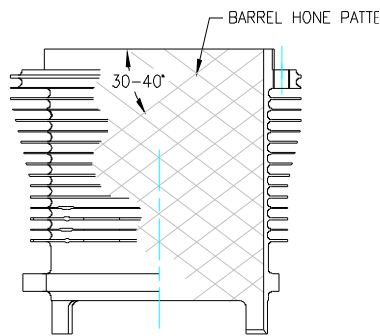


Figure 18 – Barrel Hone

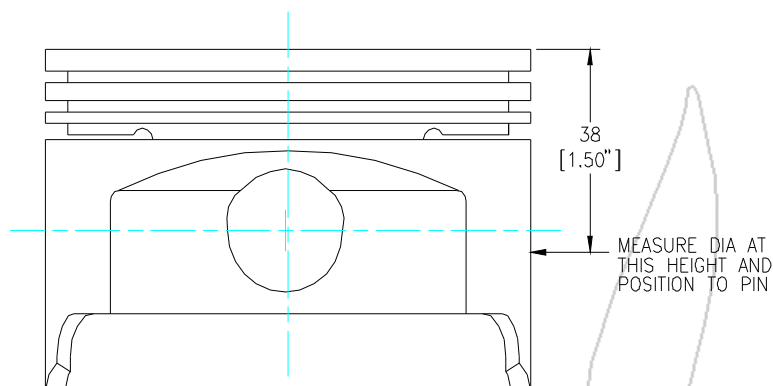
- 3) Measure and record in Build Sheet
- 4) Bead blast cylinder outer to remove all old paint and rust.
- 5) Etch prime and paint with high temperature black paint.
- 6) Run 5/16 UNF Tap through head threads to remove all beads and to insure that head bolts torque up fully.
- 7) Clean in kero then in hot soapy water to remove all honing oil etc.  
*Note: Once cleaned, oil up cylinders straight away. Store in a cool dry place in a sealed container.*

#### Pistons

- C7
- 1) Clean oil out of pistons.
  - 2) Dip Pistons in cold dip solvent (i.e. Redik DKT degreaser - decarboniser - paint stripper) as per manufacturers instructions to remove all carbon.  
*Note: The head of the piston can be bead blasted, but never bead blast the ring grooves, piston skirt and piston pin bores.*
  - 3) Use an old ring to clean carbon out of ring grooves but care must be taken not to scratch grooves. Any scratching will cause Gas Leakage past the rings. A piece of 2mm (0.080") perspex can also be used as it is kinder to the pistons.
- C8
- 1) Clean thoroughly.
  - 2) Visually inspect pistons for cracks or damage.
  - 3) Measure pistons and record in build sheet.

#### NOTE:

***Top ring has uninterrupted vertical edge.***  
***2<sup>nd</sup> Ring has 2 cut aways on vertical face (Top and Bottom)***


**Figure 19 - Drawing 9434064**

- 4)
  - (i) Clean inhibitor off new rings.
  - (ii) Measure end gap of rings in the cylinder and record in build sheet.
  - (iii) Fit rings with dots up  
*Note: End of oil ring expanded must be butted together. Piston rings orientated with joins opposite and not in thrust line.*
  - (iv) Fit pistons in cylinders. Oil Ring should only just enter so piston pin can be fitted later.  
*Note: Arrows on inside of pistons point in direction of rotation. Oil Rings and pistons and bore well. Also notch on the piston crown faces the Propeller Flange.*
  - (v) Fit front piston pin circlip.
  - (vi) Fit cylinder base O Ring.
  - (vii) Seal up in plastic bags ready for final assembly.

## 8.8 Jabiru 2200 Component Inspection and Assembly Procedure

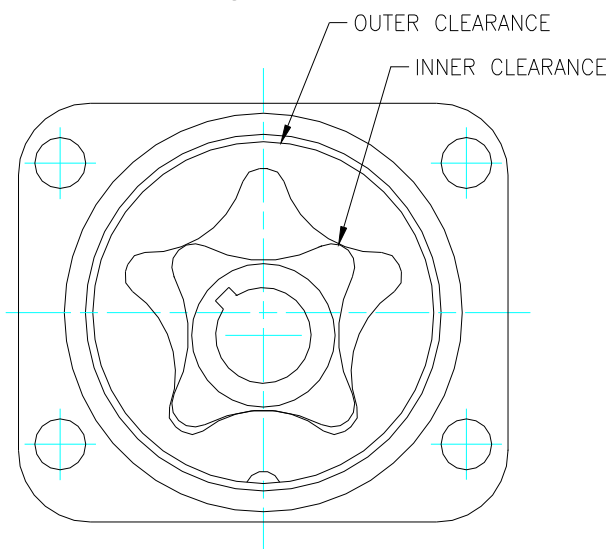
### 8.8.1 Sub-Assembly D - Sump and Oil Pump

#### Sump

- D1 1) Clean sump and remove all gasket eliminator.  
2) Inspect oil pick up.
- D2 Remove induction O Rings.
- D3 Inspect carburettor coupling replace if it is damaged or showing signs of deterioration.  
*Note: It is very important that the carburettor and coupling are free of oil. If there is any oil present at all, it is possible for the carburettor to slip out of the coupling or if the clamp is not positioned to the rubber edge facing the carby*
- D4 Fit new induction O Rings to Induction Pipes.

#### Oil Pump

- D5 1) Inspect oil pump inner and outer for damage to rotor surfaces.  
2) Measure rotor clearance.



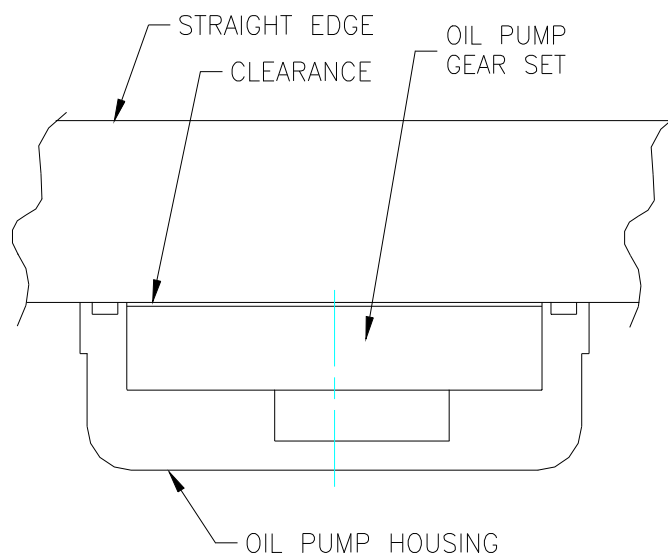
Drawing 9440064/1 OIL PUMP CLEARANCES

**Figure 20 - Drawing 9440064**

Housing to Outer Rotor clearance 0.07mm to 0.14mm (0.003" to 0.006")

Inner Rotor to Outer Rotor 0.07mm to 0.14mm (0.003" to 0.006")

- 3) Fit Inner Rotor into Outer Rotor. Using straight edge check end clearance.  
End Clearance 0.03mm to 0.006mm (0.0015" to 0.003")



Drawing 9428064/1 OIL PUMP END CLEARANCE

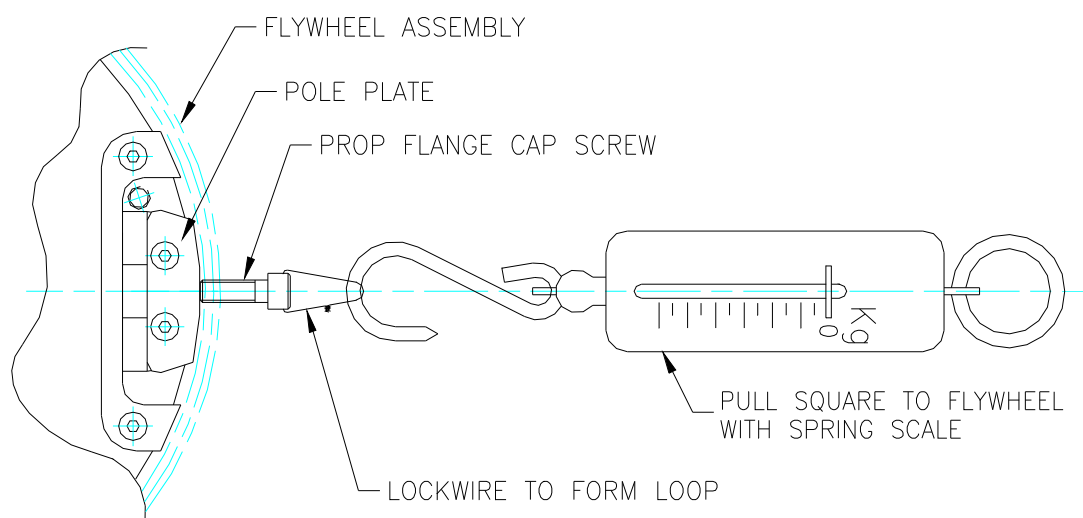
**Figure 21 - Drawing 9428064**

## 8.9 Jabiru 2200 Component Inspection and Assembly Procedure

### 8.9.1 Sub-Assembly E - Flywheel & Ignition Coils & Alternator & Alternator Operation

#### *Flywheel Inspection*

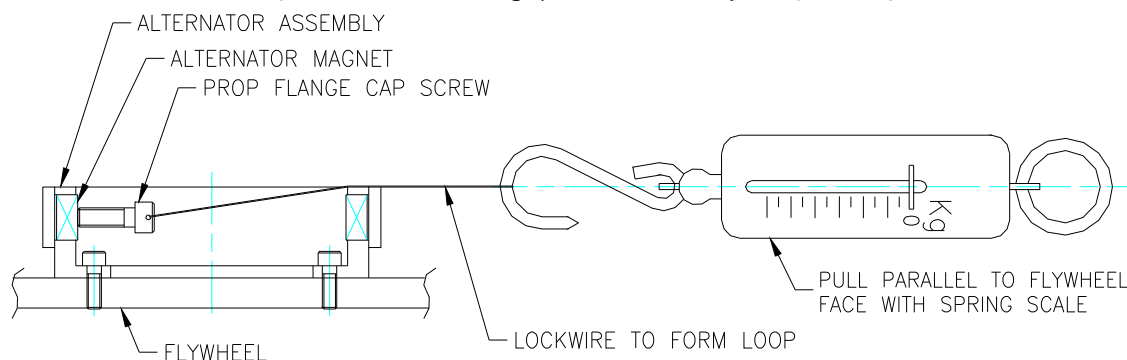
- E1 Inspect ring gear for damaged teeth
- E2 Use a spring scale and a prop cap screw as per drawing to test ignition magnet strength. Pull off should be 1.5 to 2.5 kg (3.3 to 5.5 lbs)



Drawing 9437064/1 MAGNET PULL TEST – FLYWHEEL

**Figure 22 - Drawing 9437064**

- E3 Use spring scale and prop cap screw as in E2 to test alternator magnets. Pull off should be 1.5 to 2.5 kg (3.3 to 5.5 lbs).



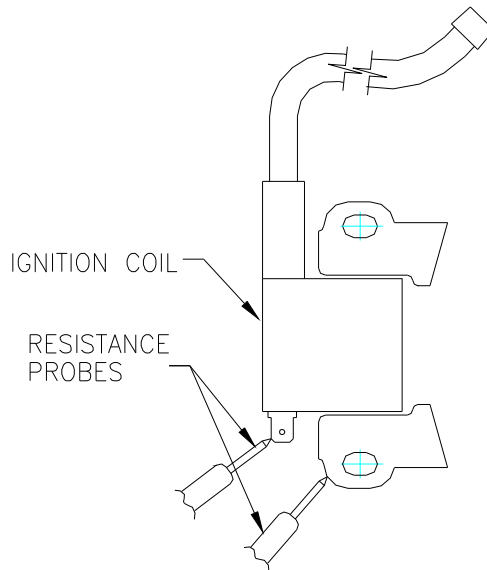
Drawing 9438064/1 MAGNET PULL TEST – ALTERNATOR

**Figure 23 - Drawing 9438064**

#### *Ignition Coil Inspection*

- E4 1) Using a multimeter measure the primary resistance (from the earth

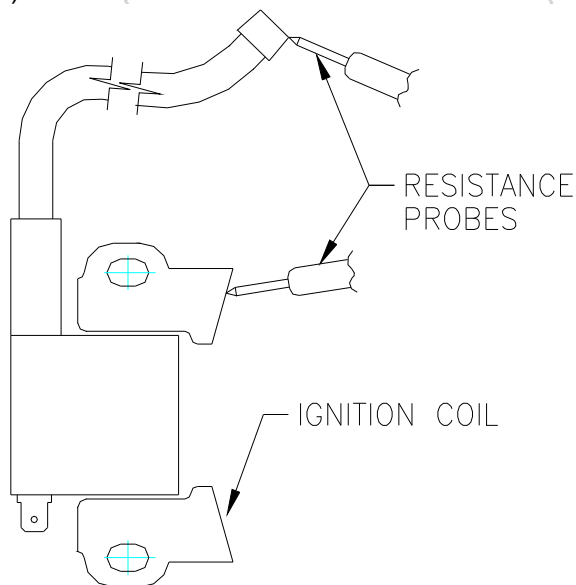
terminal to the iron core). It should be 0.8 R to 1.0 R.



Drawing 9439064/1 IGNITION PRIMARY RESISTANCE

Figure 24 - Drawing 9439064

- 2) Measure the secondary resistance (from the high tension lead to the iron core) it should be between 5.9 KR to 7.1 KR.



Drawing 9439064/1 IGNITION SECONDARY RESISTANCE

Figure 25 - Drawing 9439064

### Alternator Inspection

- E5 1) Use multimeter to test resistance of windings. Resistance should be 0.5 to 1.1 R.

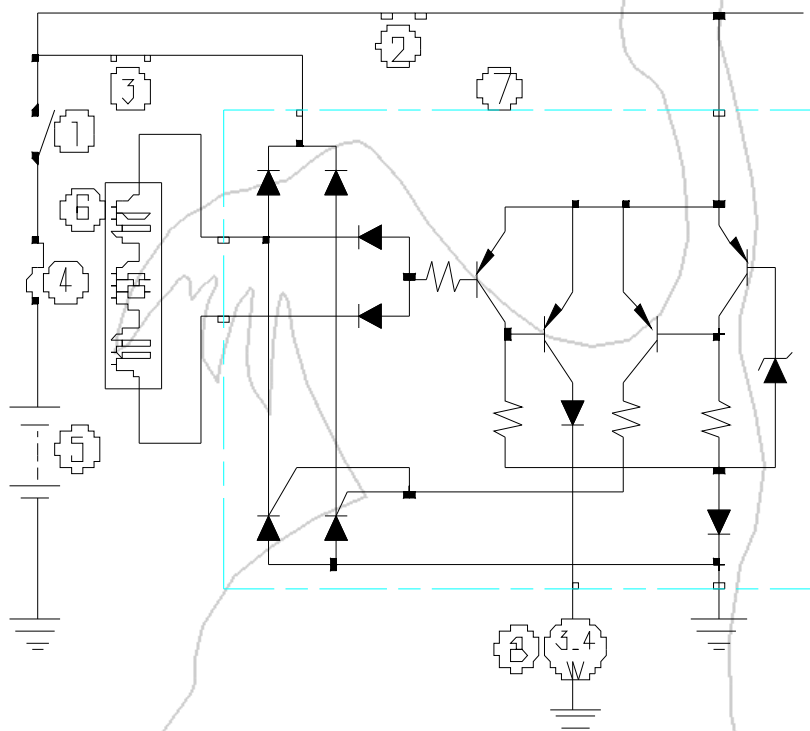


- 2) Use multi meter to test resistance to ground. Resistance should be infinite.

### Permanent Magnet Alternator (PMA) Operation

- (1) Master Switch (3) Main Bus Bar (5) Battery (7) Regulator  
(2) Voltage Reference Line (4) Fusible Link (6) Permanent Magnet Alternator (8) Charge Lamp

The charging system supplies electrical devices and also charges the battery while the engine runs. It consists of a Permanent Magnet Alternator (PMA) and a Regulator.



**Figure 26 – Alternator Internal Wiring**

1. This permanent magnet alternator (PMA) is a 10 pole rotating magnet type generator. It is a simple construction consisting of a stator and rotor. The rotor is made up of ten permanent magnets, alternator body and magnet retainer mounted on the flywheel.

The stator has ten poles with coils and is mounted on the rear plate the stator goes inside the rotor.

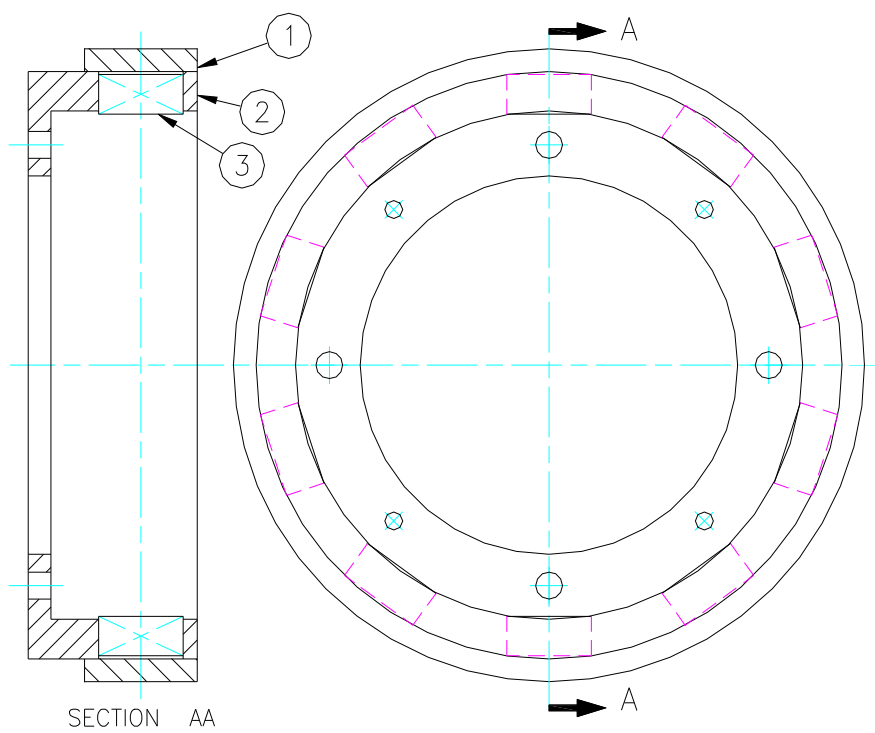
Coils are installed with insulation washes.

They are positioned such that coil on cylinders #1, 3 the output lead from the coil points down to the distributor. Output from coil at cylinder # 2 , 4 point up, then curled down to the distributor cap.

They then follow direction of prop rotation.

Coils are set at a gap of between .010" - .012"

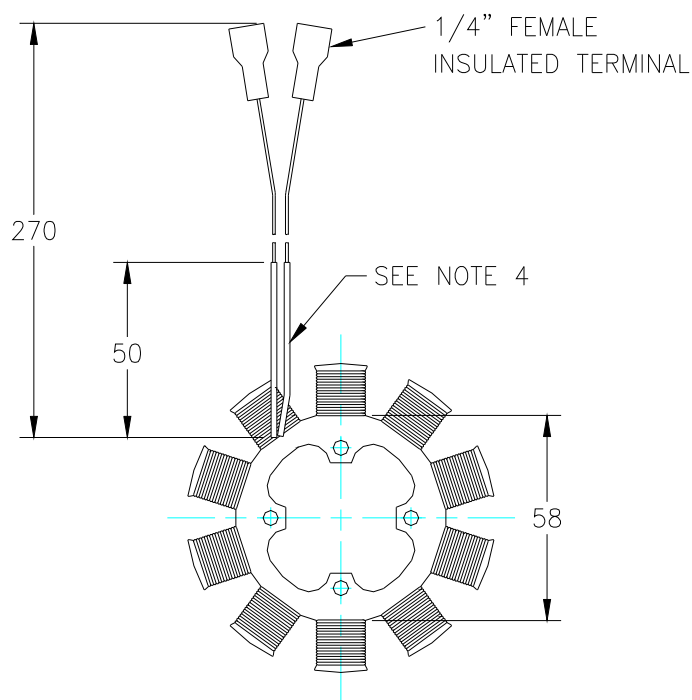




| 3    | PM00656  | MAGNET 15ø X 7  | 10   |
|------|----------|-----------------|------|
| 2    | 4532164  | ALTERNATOR BODY | 1    |
| 1    | 4532264  | MAGNET RETAINER | 1    |
| ITEM | PART No. | DESCRIPTION     | QTY. |

Drawing 4532064/2 ALTERNATOR ASSEMBLY

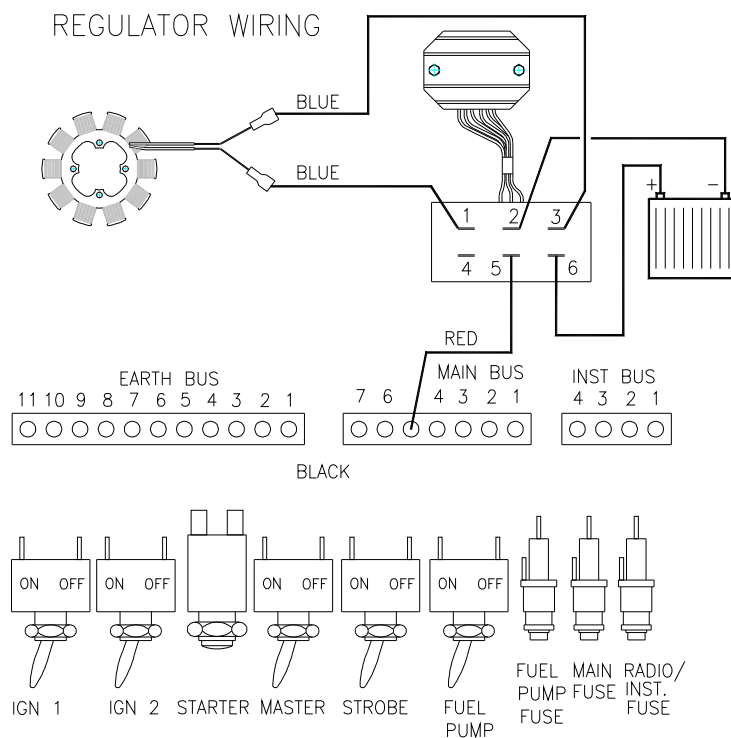
**Figure 27 – Alternator Body Assembly**



Drawing 4535064/1 ALTERNATOR CORE ASSEMBLY

**Figure 28 – Alternator Stator Assembly**

## 2. Regulator



**Figure 29 – Regulator Wiring**

- The regulator performs rectification and voltage regulation.
- The regulator converts A.C. into D.C. which flows through the power consuming circuits and the battery, and also charges the battery.
- If however, the battery voltage exceeds a certain level, the D.C. is cut off from the charging circuit to prevent over charging. An isolating switch or fuse could be used.
- For best results it is recommended that positive and negative of the regulator go directly to the battery.

## 8.10 Jabiru 2200 Component Inspection and Assembly

### 8.10.1 Sub-Assembly F - Gear Case

- F1 Remove distributor shaft seals and rotor seal and discard.
- F2 Inspect Distributor shaft posts for signs of leaking.  
*Note: If there is no signs of leaking don't disassemble. But if it is leaking disassemble and clean off old master gasket. Inspect surfaces and reassemble.*
- F3 Clean gear case distributor shaft and engine mount plate.
- F4 Measure distributor shafts and record on build sheet.
- F5 Measure distributor shaft posts and record in build sheet.
- F6 Fit shafts into gearcase and using a straight edge and feeler gauge measure end float. Record in build sheet.
- F7 Visually inspect gears for wear.
- F8 Fit seals to gear case. Use suitable seal lubricant.
- F9 Lube shafts and fit to gear case.
- F10 Fit rotors, rotors are glued to shafts.
- F11 Visually inspect engine mount plate for
  - 1) Damaged Threads
  - 2) Wear from distributor shafts.
- F12 Seal up and store for final assembly.

## 8.11 Jabiru 2200 Component Inspection and Assembly

### 8.11.1 Sub-Assembly G - Fuel Pump and Carburettor and Carburettor Operation

#### *Fuel Pump*

The fuel pump on the 2200 engine is replaced as a complete unit at overhaul. It can be inspected for faults by removing the top half.

Things to look for are:

- 1) Stuck or leaking valves
- 2) Torn diaphragm
- 3) Weak springs
- 4) Broken actuating lever.

#### *Carburettor Operation*

The BING constant depression carburettor type 64-3 or 94/40 comprises a cross draught butterfly-valve carburettor with variable choke tube, double-float system arranged centrally below the carburettor venturi and a rotary-valve type starting carburettor. It features a throttle slide which is suspended from a roller diaphragm and projects into the venturi. It changes the smallest cross-section ("choke tube") of the venturi as a function of the vacuum at this point.

Early throttle valve diameter is 36mm, later is 42mm

#### MOUNTING

The carburettor is secured to the engine using a 52mm diameter push-on connection which takes a flexible connecting piece with clamps. On the intake side the carburettor is provided with a socket having a diameter of 55mm and a length of 12mm for connecting an air filter or intake silencer.

#### FUEL INTAKE CONTROL

The float (40) of the carburettor consists of two plastic float elements joined by a metal hinge. The float is arranged centrally below the carburettor choke tube so that the carburettor can be tilted very far in all directions without impairing operation. The object of the float is to maintain the fuel level in the float chamber (44) constant. When the fuel has reached a specified level in the float chamber, then the float (40) mounted on pin (41) is lifted until the float needle (42) is pressed against the seat of the float needle valve, thus preventing any further supply of fuel. When the engine draws in fuel from the carburettor, the level in the float chamber (44) drops and so does the float. The float needle then opens the valve again and allows fuel to flow in from the tank. The float needle valve regulates the fuel supply in conjunction with the float but it does not act as a stop valve when the engine is at a standstill. Minute foreign bodies may be deposited between valve seat and needle tip, thus preventing complete closure of the valve.

When stopping the engine, therefore, the fuel cock on the tank should always be closed. In addition the fuel should be filtered before it reaches the carburettor. The filter should be selected so that foreign bodies greater than 0.1mm are filtered out and the fuel supply is not impeded to too great an extent.

The float needle (42) contains a spring-loaded plunger which contacts the float hinge. This absorbs vibrations on the float (40). In addition the float needle (42) is connected to the float hinge by the retaining spring (43) to prevent it from moving between float and valve seat and thus reducing the fuel supply. Spring and retaining guide make a considerable contribution towards keeping the fuel level in the float chamber constant.

When fitting a new float, the fuel level must be adjusted. When doing this care must be taken to ensure that the fuel needle spring is not compressed by the float weight. It is therefore advisable to put the carburettor in a horizontal position until the float just contacts the float needle. In this position the point on the float hinge is set in such a way that the float top edges are parallel to the top edge of the float chamber.

The float chamber (44) is secured to the carburettor housing by a spring yoke (45). A seal (46) is provided between float chamber and carburettor housing. The space above the fuel level is connected to atmosphere by two ducts. When these ducts are blocked, an air cushion forms above the fuel. The fuel will not lift the float sufficiently to close the needle valve and the carburettor overflows.

The float chamber (44) incorporates an overflow pipe to allow fuel to drain off if the specified level in the float chamber is exceeded substantially due to a leaking needle valve.

## MAIN REGULATING SYSTEM WITH PRESSURE REGULATOR

The amount of mixture drawn in by the engine and thus its performance is determined by the cross-sectional area in the choke tube which is opened by the throttle valve (23). The throttle valve is secured to the valve shaft (24) by two screws (25). The end projecting from the carburettor housing carries the throttle levers (27) + (28) which are secured by the nut (3) and washer (29) to which the Bowden cable operating the throttle shaft is attached. The sealing ring (26) provides the seal between valve shaft and housing. The retaining arm (31) attached to the carburettor housing by means of screws (32) and washers (33) engages the notch in the valve shaft and thus prevents it from moving in axial direction. The return spring (35) whose action opposes the Bowden cable is attached between a bent-over tab at the lower end of the retaining arm and the throttle lever (28). If the throttle valve (23) is opened while the engine is running, the increased air flow in the choke tube results in a vacuum building up at the outlet of the needle jet (3) which draws fuel from the float chamber through the jet system. At low speeds and in particular in the case of four-stroke engines, this vacuum is not sufficient for an adequate fuel supply; it must therefore be increased artificially by using a pressure regulator. For this purpose the BING constant depression carburettor type 94 is provided with a plunger (13) operating in conjunction with a diaphragm (16); which reduces the cross-sectional area or

the needle jet outlet by virtue of its own weight or, in some applications with the additional pressure from a spring (22), and thus increases air velocity and vacuum at this point.

The plunger (13) is located centrally in the cover (20) which is secured to the carburettor housing by screws (21). The diaphragm (16) is connected to the plunger (13) by a retaining ring (17) and four screws (18) and washers (19) each. The vacuum in the choke tube acts on the top of the diaphragm and the plunger via a bore (U) in the plunger (13) and attempts to lift the plunger against its own weight and spring (22). The considerably lower vacuum between air filter and carburettor is applied to the underside of the diaphragm via duct (V) as a reference pressure.

If the throttle valve (23) is opened when the plunger (13) is closed, then a vacuum will build up in the small cross-section at the bottom of the plunger (13) which is sufficient to provide a supply of fuel. The weight of the plunger (13) and the force of the spring (22) are matched in such a way that this vacuum will be maintained with increasing speed until the plunger has fully opened the carburettor cross-section. From this point onwards the carburettor acts as a throttle valve carburettor with fixed choke tube. The vacuum increases with increasing speed.

The space in the cover (20) above the plunger guide is vented through bore (D). Its diameter is designed in such a way that it acts as a restrictor for air flowing in and out and therefore acts as a vibration damper for the plunger.

On its way from the float chamber to the choke tube the fuel passes through the main jet (1), the jet carrier (10) and the needle jet (3); as it leaves the needle jet it is pre-mixed with air which is brought in from the air filter via an air duct (Z) and the atomiser (2) in an annular flow around the needle jet. This air flow assists the atomising process to form minute fuel droplets and thus favourably affects the fuel distribution in the intake manifold and combustion in the engine.

The conical section of the jet needle (4) which is secured to the plunger (13) with the retaining spring (14) and the serrated washer (15) engages into the needle jet (3). Depending on the dimension of the flat cone at the end of the jet needle, the annular gap between jet needle and needle jet is enlarged or decreased and thus the fuel supply is throttled to a lesser or greater extent. The jet needle (4) can be located in the plunger (13) in four different positions which, similarly to the jet needle cone, affect the amount of fuel drawn in. For example "needle position 3" means that the jet needle has been suspended from the retaining spring (14) with the third notch from the top. To achieve the height adjustment the jet needle is turned through 90° and pushed up or down, the retaining spring engaging the next notch in the jet needle. If the needle is suspended higher up, this will result in a richer mixture and vice versa.

In short the main regulating system is set using main jets and needle jets of various diameters and also jet needles, plungers and pistons of various types.



Between main jet (1) and nozzle stock (10) a washer (12) is provided which, together with the float chamber, forms an annular gap. In particularly severe operating conditions this ensures that the fuel is not spun away from the main jet.

A rubber ring (11) seals the nozzle stock (10) off from the carburettor housing to avoid any fuel being drawn in via the thread and thus bypassing the main jet.

## IDLING SYSTEM

During idling and low-load running the throttle valve (23) is closed to such an extent that the air flow underneath the plunger (13) no longer forms a sufficient vacuum. The fuel is then supplied via an auxiliary system, the idling system, which consists of the idling jet (5), the idling air jet (LLD) - no spare part - and the mixture control screw (7) which is sealed off against the carburettor housing by the rubber ring (9) and secured by spring (8) to prevent it from becoming slack. The fuel passes through the idling jet (5) whose bore will determine the amount of fuel. Behind the jet bore the fuel mixes with air which is supplied via cross ducts in the jet throat from the idling air channel, the amount of air admitted being determined by the size of the idling air jet at the inlet of this duct. This initial mixture then flows through the idling outlet bore (LA), the cross-sectional area of which can be adjusted by the mixture control screw (7); it then reaches the choke tube via bypass or transition passages (BP) where it is mixed further with pure air.

Idling should always be with the engine at operating temperature. First the mixture control screw (7) is turned fully clockwise and then backed off by the number of turns specified for the particular engine. Turning in clockwise direction results in a leaner mixture and turning in anti-clockwise direction in a richer mixture.

The idling setting quoted serves as a guide only. The optimum will generally differ slightly. First select the desired idling speed by using the idling stop screw (34). When subsequently adjusting the mixture control screw - starting from the basic setting - a speed drop will be noticed in both directions. The optimum setting will generally be found half-way between the two settings at which this speed drop was noticed.

To facilitate the idling setting on engines having several carburettors where it is important that they are evenly adjusted, it is possible to connect a pressure gauge (in the simplest case a "U-tube pressure gauge") to a nipple below the throttle valve shaft bearing point which is normally closed off by screw (39). To select the idling speed, the idling screw (34) is in this case adjusted until the same vacuum is indicated for all carburettors. By slightly opening the throttle valve via a turning handle or the accelerator it is also possible to adjust Bowden cables or linkages evenly by making this vacuum comparison.

## STARTING CARBURETTOR

BING constant depression carburettor is provided with a rotary valve starting carburettor as an aid for starting a cold engine using a Bowden cable. A disc (47)

resting against the carburettor housing is turned via a shaft in the starting carburettor housing (48) so that the starting carburettor chamber into which air enters from the air filter side of the carburettor is connected to the engine side of the carburettor via a duct. The airport in the disc (47) is shaped in such a way that depending on the disc position, more or less air is drawn in. At the same time the disc opens the fuel system of the starting carburettor via bores matched to the disc position. The fuel flows from the float chamber through the starting jet into the vented starting chamber also contained in the float chamber (44) and from there through a riser where it is pre-mixed with air via transverse bores, into the starting carburettor. There it forms a particularly rich mixture with the air drawn in, and this mixture bypasses the main carburettor to flow into the intake manifold of the engine direct. **During starting the throttle valve has to be closed to make sufficient vacuum available for the starting carburettor.** When the engine is at a standstill and also during normal operation the fuel level in the float chamber compartment incorporating the riser will be the same as in the rest of the float chamber. When starting with opened-up starting carburettor, the fuel will initially be drawn in from this compartment which forms a very rich mixture. The fuel supplied subsequently will only be the amount allowed through by the starting jet. This ensures that, once the engine has started, it is not supplied with an excessively rich mixture and stalled. The starting carburettor is therefore matched to any given engine by modifying the starting jet and matching the space behind it.

The starting carburettor is secured to the carburettor housing by four screws (51) and protected against ingress of dirt and water by the seal (50) between the two. The starting shaft is also sealed against the starting carburettor housing by a rubber ring (49).

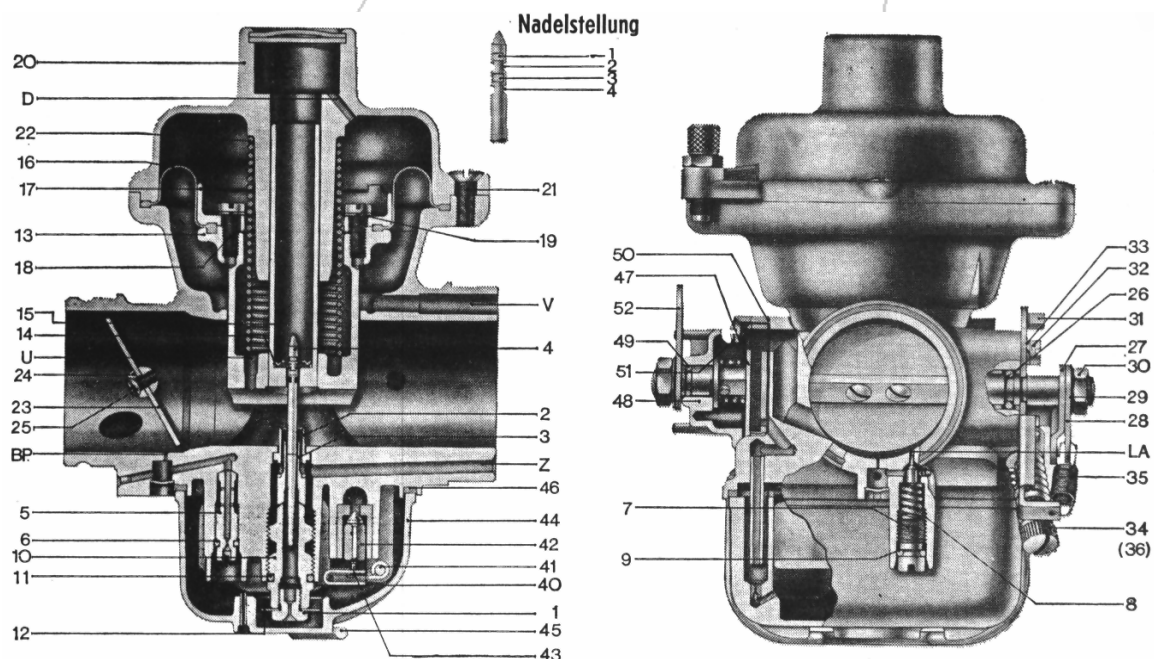


Figure 30 – Carburettor Assembly

### 8.11.2 Carburettor Overhaul

Note: As a guide to overhaul read carburettor operation.

- G1 Disassemble carburettor
- G2 Using over haul kit replace all components
- G3 Record jet size
- G4 Record needle position
- G5 Reassemble carburettor

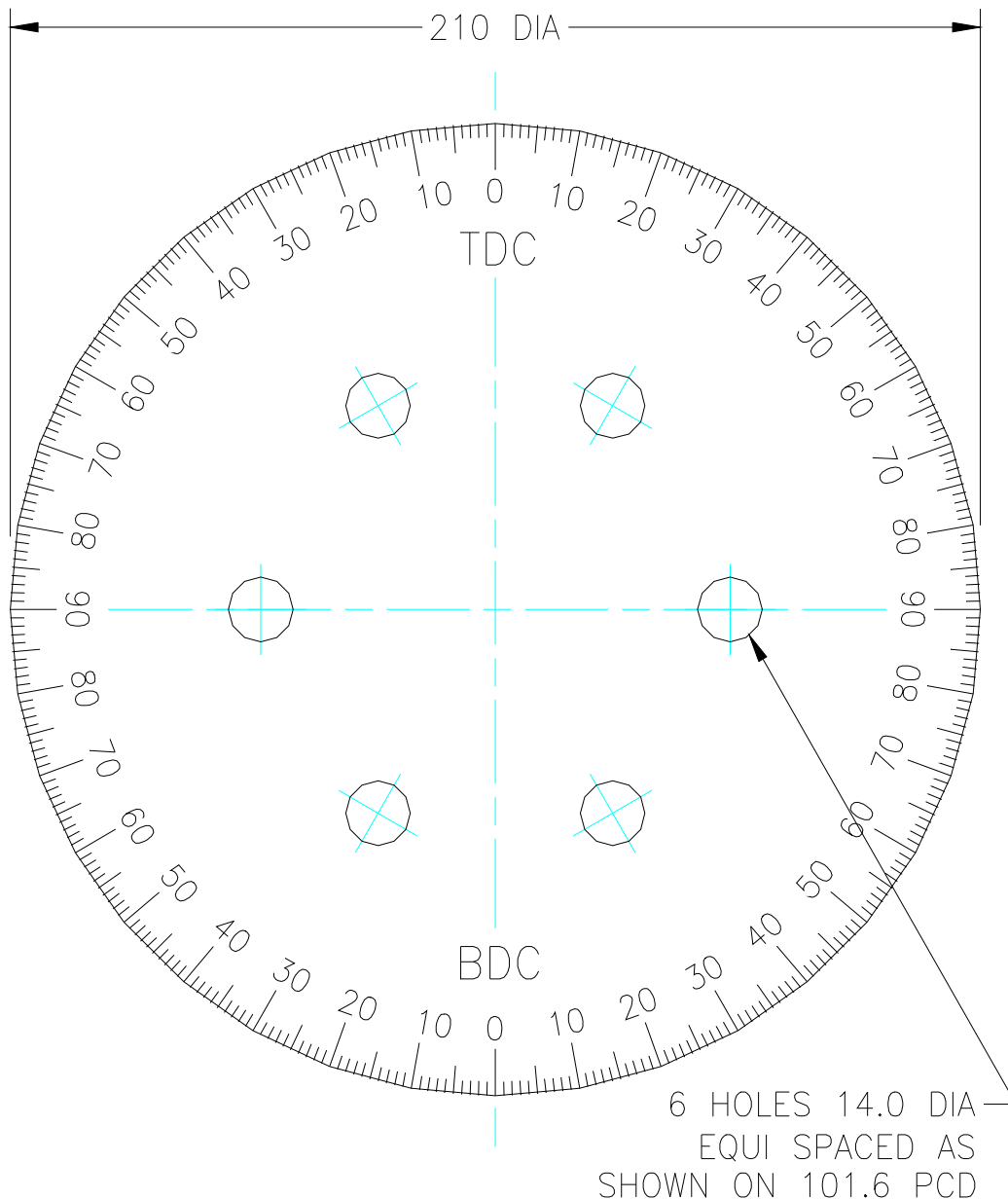


## 8.12 Jabiru 2200 Component Inspection and Assembly

### 8.12.1 Sub-Assembly H - Final Assembly

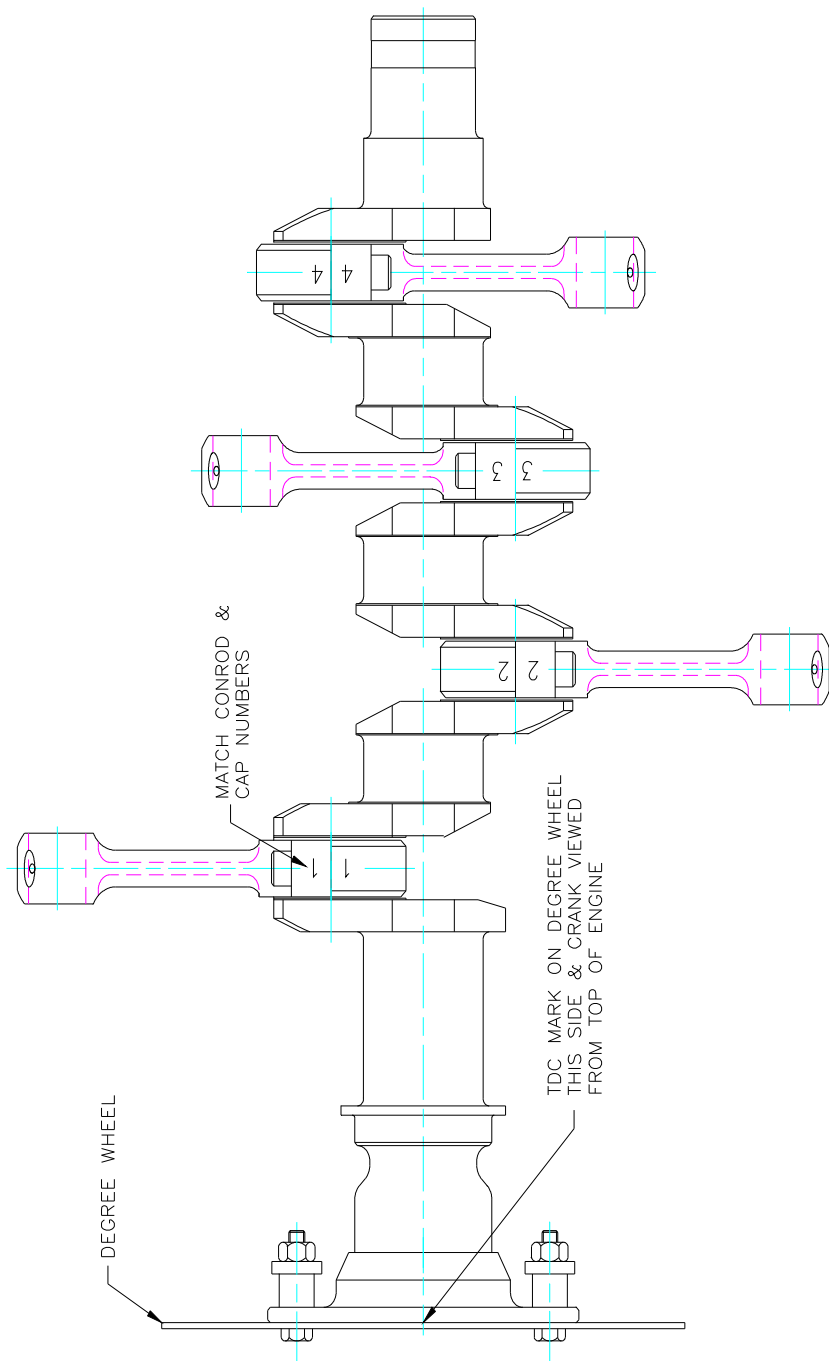
#### *Crankcase Joining*

- H1 a) Photocopy Degree Wheel (increase by 226% to bring to full size) then glue to stiff cardboard and cut out.



Drawing 9427063/1 DEGREE WHEEL

**Figure 31 - Drawing 9427063**



Drawing 9429063/2 CRANKSHAFT & CONROD ASSEMBLY

Figure 32 - Drawing 9429063

- b) Stand crankshaft (Sub assembly A) up vertically, mounted on the prop flange as per drawing.
- c) Apply Loctite 515 to left side of the crankcase.
- H2 a) Fit cam shaft to left side
- b) Inspect
- 1) Bearing shell in position
  - 2) Loctite on all areas
  - 3) O Rings in

|          |   |  |  |  |  |                  |                        |                 |
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- 4) Piston/Cylinder Assembly complete
- 5) Thrust washers in position
- 6) Dowels in
- 7) Lifter are in with moly grease on faces

- c) You will need
- 1) An extra person
  - 2) 4 7/16 Ring Open End spanners
  - 3) 2 9/16 Ring Open End spanners
  - 4) 2 3/8 Plain grade 5 nuts
  - 5) 2 3/8 Belleville Washers
  - 6) 18 through stud nuts (MS21042L6)
  - 7) 6 through studs
  - 8) Torque wrench
  - 9) 7/16 x 3/8 drive 2" extension  
(snap on no fres 14)
  - 10) Long Nose Pliers

Note: An extra person makes it easier to bolt up and also can cross check work.

- H3 a) Stage Inspection
- b) Oil all bearing, crank and camshaft
- c) Fit halves over crankshaft assembly and squeeze up by hand.
- H4 a) Using spacers and plain nuts on rear studs and belleville washers and MS21042L6 nuts on the front studs pull the crankcase together (but don't do up tight)
- b) Remove rear nuts
- H5 Put all through studs in
- H6 a) Fit piston/cylinder assemblies to No. 1 & 2 conrods
- b) Fit circlips cross check circlips for correct fit. Push cylinders home.
- c) Put cylinder base nuts (MS21042L6) on. Using 2 7/16 ring spanners do up each side together.
- If one nut goes on more than 3 turns put the 3/8 plain nut on top of it and use a 9/16 ring spanner to hold the nut, locking up the stud. Torque up by hand lightly.
- d) Turn engine 1/2 turn so No. 3 & 4 conrod are out. repeat procedure on No. 3 & 4.
- H7 a) Torque to 15 ft lbs then 30 ft lbs.

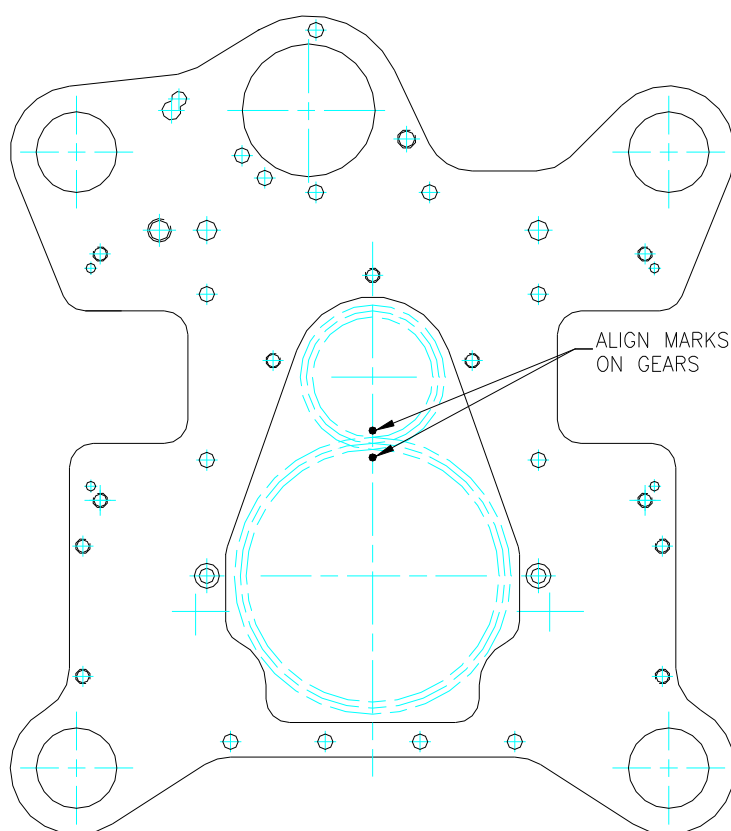
Note: Keep an eye on the amount of thread through each nut. Both sides should have approx 2 turns hanging out.

- b) Put anti sabotage. Paint on nuts.

### STAGE INSPECTION

#### Camshaft Timing

- H8 a) Fit crankshaft gear as drawing and put one bolt in crankshaft to hold gear.



Note: Timing marks on cam gear are on 2 teeth. Crankgear marked on one tooth.

Drawing 9432064/1 TIMING GEAR ALIGNMENT

**Figure 33 - Drawing 9432064**

- b) Mount a dial indicator on cylinder with the plunger on the center of the piston.
- c) Zero the indicator at T.D.C.
- d) Fix a wire pointer of the oil seal bolt hole and roughly set to T.D.C.
- e) Turn 1/4 turn backwards then forward until the piston is 2mm from T.D.C. Note degrees.

Turn past T.D.C. to 2mm down from T.D.C. Note degrees.

|          |   |  |  |  |  |                  |                        |                 |
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Add two together then divide by 2 and set pointer at this (with piston down 2mm)

*Note: The crank is mounted to turn the engine in direction of rotation turn crankcase anti clockwise viewed from rear.*

- f) Repeat again to check.

*Note: It is important to find T.D.C. accurately as it is the datum for setting the cam timing and ignition timing.*

- g) Set dial indicator up to measure lift on No. 1 exhaust. Set to 0 on max lift.

- h) Turn 1/4 turn backwards then turn until 0.5mm (0.020") from peak note degrees continue turning until 0.5mm (0.020") after peak.

*Note: Always turn in direction of rotation to eliminate back lash in gears.*

- i) Calculate total degrees and divide by 2 to find peak. Peak should be 70°-72° after B.D.C.

- j) The cam shaft gear has 22 teeth and 6 bolt holes, so if it needs adjusting. Move the gear around and re check. Record final measurement in build sheet.

### Head Fitting

- H9 a) Insure \*
- \* All O Rings are place and greased
  - \* Have all head bolts
  - \* Head Gaskets
  - \* Push rod tubes
  - \* Push Rods
  - \* Rocker Blocks
  - \* Loctite 515
  - \* Loctite 626

- b) Fit heads by hand.

- c) Push the push rod tube through the head and all the way home. Fit circlips.

*Note: Make sure outside of tubes are smooth or it will damage the O Rings.*

- d) Oil Push rods and fit.

*Note: Make sure ball end of push rod is in the socket of the lifter. It also helps if the lifter is on the base circle of the lobe.*



- e) Put loctite 515 on the base of the rocker block and fit long bolt torque lightly by hand (heads with separate rocker block).  
Put loctite 262 on rocker block cap screw and torque to 16 ft lbs.

*Note: Engine serial no: 225 on do not have rocker blocks.  
1/4" unc cap screw is torqued to 8 ft lb*

- f) Torque heads to 16 nm then 32 nm (12 ft lbs then 24 ft lbs).

- g) Adjust all valves to 0.25 mm (0.010").

*Note: Valves must be adjusted with the lifter 180° to the peak lift to ensure it is on the base circle.*

### *Sump and Induction*

- H10 a) Fit sump using loctite 515 on faces and loctite 243 on threads.

*Note: Because the engine mount plate goes on the sump and crankcase it is important to make sure the sump is level with the rear crank case face. Later engines (S/No. 1400 on) have two additional cap screws between the backing plate and the sump, covered by the gear case.*

- b) Fit 3 & 4 induction pipe to heads then 1 & 2

- c) Fit lower pipes and hoses

### *Gear Case and Rear*

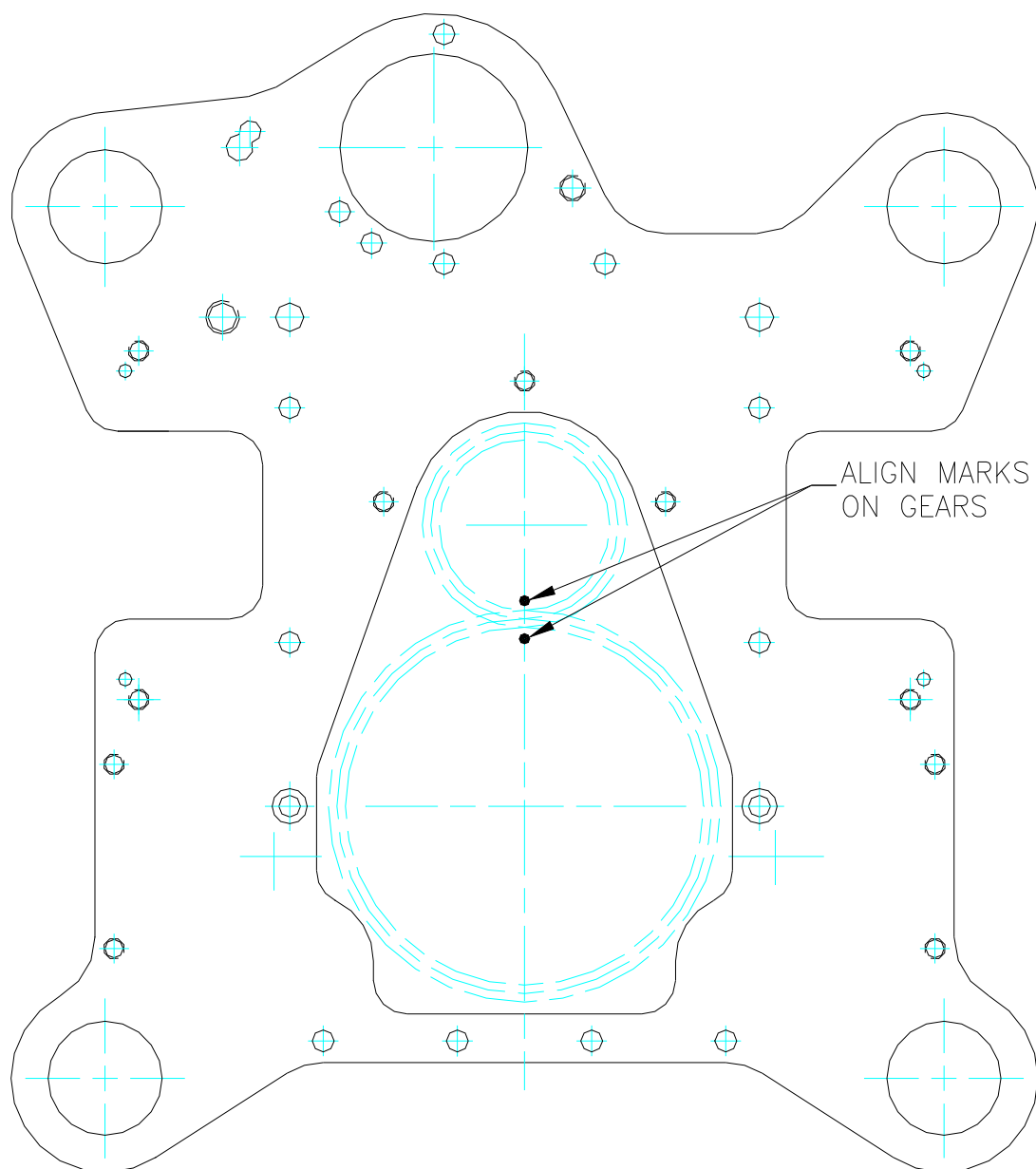
- H11 a) Insure \*
- \* Gear case is fitted up
  - \* Backing plate is ready
  - \* All bolts are ready
  - \* Loctite 515
  - \* Loctite 242
  - \* Loctite 262

- b) Turn engine around until it is 25° B.T.D.C. No. 1 compression

- c) Put loctite 515 on crankcase. Fit engine mount plate. Make sure 2 inside cap screws are in. Use loctite 242 on all cap screws.

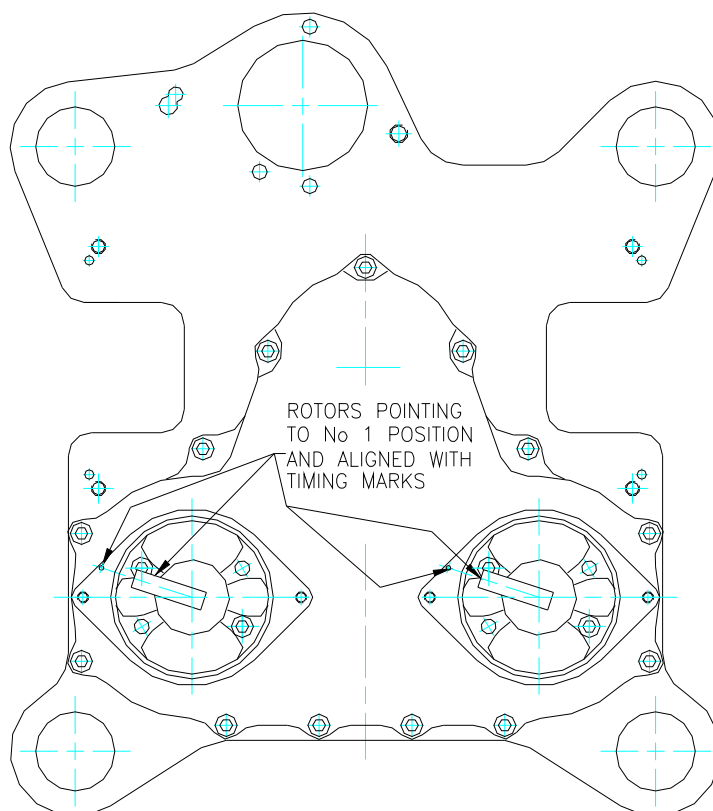
*Note: It is recommended that rotors are attached with small amount of silastic or 5 minute araldite/flox mix. This will prevent them from loosening. This still allows rotors to be removed when necessary*

- d) Put loctite on gear case and point rotors to No: 1 Position.



Drawing 9432064/1 TIMING GEAR ALIGNMENT

**Figure 34 – Drawing 9432064**



Drawing 9431064/2 REAR PLATE GEAR HOUSING &amp; ROTORS

Figure 35 - Drawing 9431064

*Note: When gear case goes on*

- 1) Make sure No. 1 is 25° before T.D.C. compression. Both valves should be seated.
- 2) The rotors point to the No. 1 position on the rotor caps.  
Fit cap screws with loctite 242 and torque.

- H12 a) Fit flywheel with the magnets horizontal (they then should line up with the coils).

*Note: The small hole in the flywheel will line up with the hole in the crankcase. (It may not line up with the hole in the gear if it has been moved for cam timing.) Put loctite 620 on new flywheel cap screws and torque to 20 nm (15 ft lbs).*

5/16 cap screws from Serial Number 437 @ 24 ft lb.

- b) Fit coil mount posts and alternator mount.
- c) Fit coils and set gap to 0.25mm (0.010")

*Note: A strip of 0.25mm (0.010") card is best because it is not a magnet. Coil direction lead follows prop rotation.*

- H13 Fit Carburettor assembly

*Note: It is very important to make sure hose clamp is at the outside end of coupling, the coupling and carby are free of oil and*

|          |   |  |  |  |  |                  |                        |                 |
|----------|---|--|--|--|--|------------------|------------------------|-----------------|
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|----------|---|--|--|--|--|------------------|------------------------|-----------------|

*the carburettor goes fully into coupling, clamp is of 10mm wide stainless type.*

#### H14 Fit Fuel Pump Assembly

*Note: Use loctite 242 on cap screws. Use molybdenum disulphide grease on push rod.*

#### Oil Pump

H15 a) Put loctite 515 on back face of oil pump backing plate and fit.

b) Fit woodruff key

c) Fit inner gear to camshaft.

d) Fit outer gear to housing and fill with oil.

e) Fit O Ring

f) Fit Oil pump on. Use loctite 242 on cap screws.

*Note: Before torquing up, rotate engine carefully to ensure oil pump is not binding. If oil pump binds it can shear the end of camshaft.*

H16 Fit exhaust system. Use a little anti seize on cap screws and shake proof washers.

#### Front Seal & Flange

H17 a) Remove from vertical stand and mount on engine mount stand.

b) Remove prop flange.

c) Clean off oil from sealing surfaces.

d) Use loctite 515 on crankshaft seal carrier to crankface and loctite on cap screws. Put a little grease on seal. Install seal.

e) Prime crankshaft and new prop flange bolts with loctite primer. Using loctite 620 fit flange and torque to 40 nm (30 ft lbs).

e) Wire flange,

#### CAUTION

**if for any reason prop flange or prop has been damaged rear flywheel capscrews should be replaced.**

### 8.12.2 Engine Run In

The Jabiru engine is run in on a DYNOMOMETER and cooled with fan driven air. In the absence of a DYNO controlled run, engines can be run in the airframe. Very large air ducts must be constructed to get adequate airflow for cooling in the static situation. The flying air ducts must not be used for this purpose. Extreme care must be taken with the cylinder head temps if run in is done in the airframe. Engine run in procedure allows progressive build up of the B.M.E.P. (break mean effective pressure) in the cylinders while carefully limiting the heat build up. In essence short amounts of hard work increasing in intensity while limiting the heat build up is the format. This can be seen in detail after the engine build sheets.

**All engines are run in before delivery including overhauled engines as well. When delivered they are ready for flight. See early run section 5 for first 25 hours of operation.**

Run Procedure to be used only if engine has not been run-in.

- 1) Mount Engine on test rig (See engine installation)
- 2) Fill with non detergent oil.
- 3) Remove spark plugs and wind over until oil pressure is reached.
- 4) Fit plugs and cooling ducts.
- 5) Run in.  
Follow run in program.
- 6) Cool down (at least 12 hours)
- 7) Retorque heads
- 8) Adjust valves
- 9) Rerun
- 10) Check leak down

*Note: There is a run in program and run in checklist to fill out.*

### 8.13 Engine Installation

#### Operation

- 1) Fit engine to engine mount
- 2) Torque engine mount bolts up

#### Tools

7/16 Tube Socket



- |   |   |
|---|---|
|   | 7/16 Ring Open End<br>Spanner           |
| 3) Fit muffler (if not already fitted) and fit carburettor<br>heat muff and hose. | 3/16 Ball End<br>Allen Key              |
| 4) Connect left and right ignition coils leads                                    |   |
| 5) Connect Tacho sender   |   |
| 6) Connect exhaust gas temp (if fitted)   |   |
| 7) Connect cylinder head sender   | 18mm Spark Plug<br>socket               |
| 8) Connect hourmeter  | Screw Driver                            |
| 9) Connect oil temp gauge sender  |   |
| 10) Connect oil pressure gauge sender   |   |
| 11) Connect starter<br>Spanner  | 7/16 Ring Open End                      |
| 12) Connect Battery   | 10mm Ring Open<br>End Spanner           |
| 13) Connect fuel line   | Screw Driver                            |
| 14) Connect Oil Breather  | Screw Driver                            |
| 15) Connect Choke Cable   | Long Nose Pliers                        |
| 16) Connect Throttle Cable  | Long Nose Pliers                        |
| 17) Connect Air Inlet   | Screw Driver                            |
| 18) Fit Propeller and spinner<br>Spanner  | 7/16 Ring Open End                      |
| 19) Fit Cooling Ducts   | Phillips Screw Driver<br>3/16 Allen Key |
| 20) Prime Fuel system with electric pump and inspect<br>for leaks                 |   |
| 21) Check for oil. Fill if needed.  | 2.3L oil (2.43 US)                      |

quarts)

- 22) Wind over to get oil pressure
- 23) Start and inspect for leaks
- 24) Test Fly  
*Note: First flight is a test flight so fly conservatively*
- 25) Remove Cowls and inspect for anything loose, rubbing or leaking.
- 26) After 5, 10 and possibly 15 hours retorque heads, adjust valves and inspect engine installation. Heads torqued to 24 (22) ft. lb. and valve clearance 0.010".
- 27) After 25 hours retorque heads, adjust valves and inspect engine installation. Change oil and filter. Cut filter open and inspect. Check engine leak down.
- 28) If oil consumption is stable fill with W100 (W80 in cold conditions and W120 in very hot conditions). If it is still using oil remain on run in oil.

#### 8.14 Prop Strike Inspection

After ground contact of a wooden propeller, check the crankshaft and flange for run-out at the front seal surface. If run-out is evident, the engine will have to be stripped and crankshaft checked for cracks. The flywheel bolts (6 OFF UNF cap screws) will also need to be replaced. Note: If a crankshaft has been severely stressed but measurements and MPI testing indicate a sound item it is Jabiru policy to not re-use but replace with new.

#### **IMPORTANT**

**Engines running a propeller or propeller flange extension not approved by Jabiru Aircraft Australia (especially composite propellers) must check the flywheel bolts for security using an allen key at each service.**

If an engine stoppage due to force is not recorded in the logbook and not advised to Jabiru, the liability for all subsequent and consequential damage will remain with the owner. This applies to both prior to and after engine overhaul.

#### 8.15 Build Sheets and Run In Programme

Included with the build sheets are \*

- \* Pre Run Check List
- \* Run In Programme
- \* Post Run Checklist

Photocopy all documents and use photocopies to fill in. When complete, fax to Jabiru Aircraft so all documents can be kept up to date. This helps us provide a

better service to our customers if we know what our engine is in and what has been done to it.

Fax Number Within Australia 07 4155 2669  
Outside Australia + 61 7 4155 2669





Jabiru Aircraft Pty Ltd  
2200 SOLID LIFTER ENGINE

## WARNING

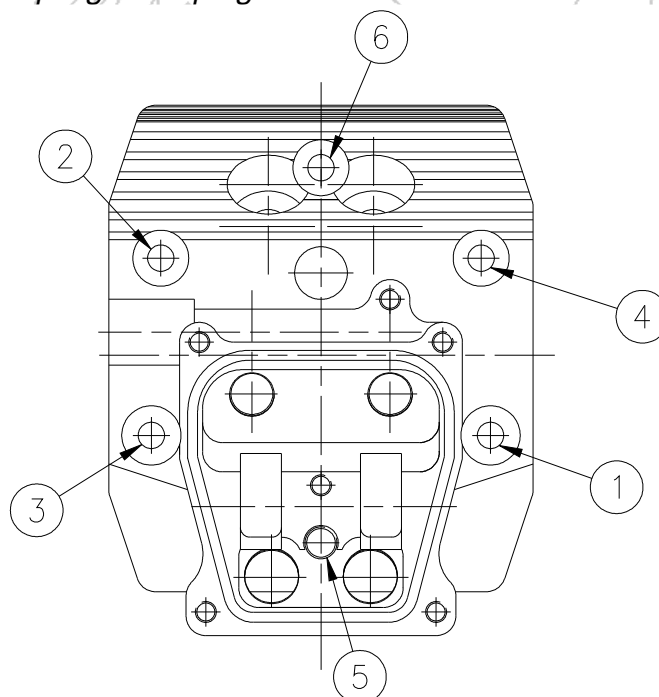
This engine has been run in and is ready for flight. This engine uses solid lifters for valve operation.

### **DO NOT GROUND RUN THIS ENGINE**

This engine has been inhibited and the oil system drained. Before first start of new engine, remove 1 spark plug from each head. Add the required run-in oil to sump (2.3 litres), engine must have oil cooler fitted. Press starter to obtain oil pressure and to throw out excess inhibiting oil in cylinders. Replace plugs. NOTE: ALL plastic bungs must be removed

At 5 hours and 15 hours check tappet clearances (0.010" cold) and cylinder head bolt torque (24 ft.lbs/32nm).

**Note:** New style heads have a 1/8 NPT plug to be unscrewed to uncover head bolt No. 5. After torquing reset plug.



**Figure 36 – Head Bolt Pattern**

- For the first 25 hours of operation, add 2.2 litres of Shell 100 oil or equivalent “run in” type aviation oil. After 25 hours drain oil, check torque of cylinder head bolts to 20 ft lbs and check valve clearances (inlet and exhaust 0.010" cold).
- Use 2.2 litres of W100 oil for normal operation or W80 oil for cold weather operation.
- Shell also manufacture a multigrade oil Aeroshell 15W50 which is particularly suited for operations in cold climates.

**UNDER NO CIRCUMSTANCES USE AUTOMOTIVE OIL IN THIS ENGINE.**

- Use only oils which are designed for Air Cooled Aero Engines.
- Some brands of automotive oils have been shown to cause very rapid cylinder wear. This will not be covered under warranty.
- Operate engine only on AVGAS 100LL or highest octane available MOGAS above 95 octane. Failure to do so could result in engine damage and void warranty.

**\*USE OF OIL/FUEL ADDITIVES VOID WARRANTY**



## 8.16 Jabiru 2200 – Component Assembly Procedure

### 8.16.1 Subassembly A – Crankshaft, Propeller Mount Flange and Conrods

| No.     | Details   | Sign | Sign | Date |
|---------|---|------|------|------|
| A1      | Inspect for Burrs, Oil Holes, Chamfers; Clean Crankshaft, Conrods and Propeller Mount Plate |      |      |      |
| A2      | Inspect Oil Holes and Insert Welch Plugs  |      |      |      |
| A3      | Measure Crankshaft (refer to Goods Inwards Inspection Sheet)                                |      |      |      |
| A4      | Inspect and Measure Propeller Mount Flange  |      |      |      |
| A5      | Inspect and Measure Conrods (refer to Goods Inwards Inspection Sheet)                       |      |      |      |
| A6      | Mount Propeller Mount Flange to crankshaft Torque to 30ft.lbs/ Lockwire                     |      |      |      |
| A7      | Fit the Conrods to the Crankshaft; Use Loctite 620 on the bolts and torque to 18ft.lbs      |      |      |      |
| Stage 1 | A8 Stage Inspection of Assembly – Conrods/Crankshaft  |      |      |      |

**I hereby certify that the subassembly has been completed using the approved procedures and data; and is fit for installation.**

Signed: \_\_\_\_\_ Date: \_\_\_\_\_

For Jabiru Aircraft Pty Ltd  
Production Certificate #444128

Notes: \_\_\_\_\_  
\_\_\_\_\_  
\_\_\_\_\_  
\_\_\_\_\_

## 8.17 Component Register and Parts Inspection Record

### 8.17.1 Subassembly A – Crankshaft, Propeller Mount Flange and Conrods

Batch \_\_\_\_\_ Serial No \_\_\_\_\_ Date \_\_\_\_\_

| Description                  | Part No | Batch | Item | Details |   |   |   |         |   |   |   |
|------------------------------|---------|-------|------|---------|---|---|---|---------|---|---|---|
|                              |         |       |      | Mains   |   |   |   | Big End |   |   |   |
|                              |         |       |      | 1       | 2 | 3 | 4 | 1       | 2 | 3 | 4 |
| Crankshaft                   | 4644082 |       |      |         |   |   |   |         |   |   |   |
| Propeller Mount Flange       | 4525064 |       |      |         |   | 5 | 6 |         |   |   |   |
|                              |         |       |      | B/E Dia |   |   |   |         |   |   |   |
| Conrod 1                     | 4651183 |       |      |         |   |   |   |         |   |   |   |
| 2                            | 4651183 |       |      |         |   |   |   |         |   |   |   |
| 3                            | 4651183 |       |      |         |   |   |   |         |   |   |   |
| 4                            | 4651183 |       |      |         |   |   |   |         |   |   |   |
| 5                            | 4651183 |       |      |         |   |   |   |         |   |   |   |
| 6                            | 4651183 |       |      |         |   |   |   |         |   |   |   |
| Conrod SHCS<br>5/16 x 1" UNF |         |       |      |         |   |   |   |         |   |   |   |
| Bearings ACL                 | 4B8290  |       |      |         |   |   |   |         |   |   |   |

I hereby certify that the above listed parts conform with the dimensions, have been engraved, and installed as recorded.

Signed \_\_\_\_\_ Date \_\_\_\_\_

For Jabiru Aircraft Pty Ltd  
Certificate of Approval #444128

REVISION

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Dated : Oct 2006

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## Jabiru 2200 – Component Assembly Procedure

### 8.17.2 Subassembly B – Crankcase and Camshaft Assembly

| <u>No.</u> | <u>Details</u>  | <u>Sign</u> | <u>Date</u> |
|------------|---|-------------|-------------|
| <b>B1</b>  | <b><u>Inspect case, deburr, clean, check oil holes</u></b>                  |             |             |
| <b>B2</b>  | <b><u>Fit inner stud “O” rings</u></b>                                      |             |             |
| <b>B3</b>  | <b><u>Fit all studs</u></b>   |             |             |
| <b>B4</b>  | <b><u>Fit outer oil suction and conrod welch plugs</u></b>                  |             |             |
| <b>B5</b>  | <b><u>Fit oil relief valve, oil pressure sender and pressure switch</u></b> |             |             |
| <b>B6</b>  | <b><u>Fit bearing shells (12)</u></b>                                       |             |             |
| <b>B7</b>  | <b><u>Assemble and torque to 30 ft.lbs</u></b>                              |             |             |
| <b>B8</b>  | <b><u>Measure main tunnel and camshaft bearings</u></b>                     |             |             |
| <b>B9</b>  | <b><u>Measure Cam Follower Bores</u></b>                                    |             |             |
| <b>B10</b> | <b><u>Fit Lifters</u></b>   |             |             |
| <b>B11</b> | <b><u>Check Camshaft End Float</u></b>                                      |             |             |
| <b>B12</b> | <b><u>Check Crankshaft End Float</u></b>                                    |             |             |

**I hereby certify that the subassembly has been completed using the approved procedures and data; and is fit for installation.**

Signed: \_\_\_\_\_ Date: \_\_\_\_\_

For Jabiru Aircraft Pty Ltd  
Production Certificate #444128

Notes: \_\_\_\_\_  
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\_\_\_\_\_

## 8.18 Component Register and Parts Inspection Record

### 8.18.1 Subassembly B – Crankcase and Camshaft Assy

Batch \_\_\_\_\_ Serial No \_\_\_\_\_ Date \_\_\_\_\_

| Description          | Part No  | Batch | Item | Details       |   |   |   |   |            |
|----------------------|----------|-------|------|---------------|---|---|---|---|------------|
| Crankcase LS         | 4A079A0D |       |      |               |   |   |   |   |            |
| Crankcase RS         | 4A078A0D |       |      | 1             | 2 | 3 | 4 | 5 | 6          |
| Camshaft             | 4738092  |       |      |               |   |   |   |   |            |
| Studs – Long Barrel  | 4291044  |       |      |               |   |   |   |   | Cam Tunnel |
| Studs – Short Barrel | 4292044  |       |      | Cam           |   |   |   |   |            |
| Studs - Front        | 4293044  |       |      | 1             | 2 | 3 | 4 | 5 |            |
| Outer Gear           | 4333054  |       |      |               |   |   |   |   |            |
| Inner Gear           | 4066123  |       |      | Cam Followers |   |   |   |   |            |
| Crankshaft Gear      | 4643084  |       |      | LHS           |   |   |   |   |            |
| Main bearings        | 5M1869   |       |      | RHS           |   |   |   |   |            |

I hereby certify that the above listed parts conform with the dimensions, have been engraved, and installed as recorded.

Signed \_\_\_\_\_ Date \_\_\_\_\_

For Jabiru Aircraft Pty Ltd  
Certificate of Approval #444128

## 8.19 Jabiru 2200 – Component Assembly Procedure

### 8.19.1 Subassembly C – Pistons, Cylinders and Cylinder Heads

| <b>No.</b> | <b>Details</b>   | <b>Sign</b> | <b>Date</b> |
|------------|--|-------------|-------------|
| <b>C1</b>  | <b><u>Clean and deburr all parts</u></b>                                     |             |             |
| <b>C2</b>  | <b><u>Record all measurements</u></b>  |             |             |
| <b>C3</b>  | <b><u>Install pushrod tube “O” Rings</u></b>                                 |             |             |
| <b>C4</b>  | <b><u>Check valve seats, fit spring retainers, install valves</u></b>        |             |             |
| <b>C5</b>  | <b><u>Fit intake pipes to heads</u></b>                                      |             |             |
| <b>C6</b>  | <b><u>Complete rocker shafts and rocker assemblies</u></b>                   |             |             |
| <b>C7</b>  | <b><u>Fit cylinder base “O” rings</u></b>                                    |             |             |
| <b>C8</b>  | <b><u>Fit front circlip</u></b>  |             |             |
| <b>C9</b>  | <b><u>Check ring end gaps, fit rings to pistons</u></b>                      |             |             |
| <b>C10</b> | <b><u>Install piston assembly to cylinder just clear of the oil ring</u></b> |             |             |

**I hereby certify that the subassembly has been completed using the approved procedures and data; and is fit for installation.**

Signed: \_\_\_\_\_

Date: \_\_\_\_\_

For Jabiru Aircraft Pty Ltd  
Production Certificate #444128

Notes: \_\_\_\_\_

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\_\_\_\_\_  
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## 8.20 Component Register and Parts Inspection Record

### 8.20.1 Subassembly C – Pistons, Cylinders and Cylinder Heads

Batch \_\_\_\_\_ Serial No. \_\_\_\_\_  
Date \_\_\_\_\_

| Description | Part No | Batch | Item | Details      |                 |               |
|-------------|---------|-------|------|--------------|-----------------|---------------|
|             |         |       |      | Guides       |                 |               |
|             |         |       |      | <u>Inlet</u> | <u>Outlet</u>   |               |
| Head 1      | 4901012 |       |      |              |                 |               |
| 2           | 490201N |       |      |              |                 |               |
| 3           | 4901012 |       |      |              |                 |               |
| 4           | 490201N |       |      |              |                 |               |
|             |         |       |      | <u>Bore</u>  | <u>Length</u>   |               |
| Cylinder 1  | 4554062 |       |      |              |                 |               |
| 2           | 4554062 |       |      |              |                 |               |
| 3           | 4554062 |       |      |              |                 |               |
| 4           | 4554062 |       |      |              |                 |               |
|             |         |       |      |              | <u>Ring Gap</u> |               |
|             |         |       |      | <u>Dia</u>   | <u>Top</u>      | <u>Bottom</u> |
| Piston 1    | 4891013 |       |      |              |                 |               |
| 2           | 4891013 |       |      |              |                 |               |
| 3           | 4891013 |       |      |              |                 |               |
| 4           | 4891013 |       |      |              |                 |               |
| Rockers LH  | 4769094 |       |      |              |                 |               |
| RH          | 477009N |       |      |              |                 |               |
| Gudgeon Pin | 4299054 |       |      |              |                 |               |

I hereby certify that the above listed parts conform with the dimensions, have been engraved, and installed as recorded.

Signed \_\_\_\_\_ Date \_\_\_\_\_

For Jabiru Aircraft Pty Ltd  
Production Certificate #444128





## 8.21 Jabiru 2200 – Component Assembly Procedure

### 8.21.1 Subassembly D - Sump

| <u>No.</u> | <u>Details</u>  | <u>Sign</u> | <u>Date</u> |
|------------|---|-------------|-------------|
| <b>D1</b>  | <b><u>Deburr and wash sump, inspect for defects</u></b> |             |             |
| <b>D2</b>  | <b><u>Fit oil temperature sender</u></b>                |             |             |

**I hereby certify that the subassembly has been completed using the approved procedures and data; and is fit for installation.**

Signed: \_\_\_\_\_ Date: \_\_\_\_\_

For Jabiru Aircraft Pty Ltd  
Production Certificate #444128

Notes: \_\_\_\_\_  
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## 8.22 Component Register and Parts Inspection Record

### 8.22.1 Subassembly D – Sump

**Batch** \_\_\_\_\_ **Serial No.** \_\_\_\_\_ **Date** \_\_\_\_\_

| <b>Description</b> | <b>Part No</b> | <b>Batch</b> | <b>Item</b> | <b>Details</b> |
|--------------------|----------------|--------------|-------------|----------------|
| Sump               | 4823002        |              |             |                |

I hereby certify that the above listed parts conform with the dimensions, have been engraved, and installed as recorded.

Signed \_\_\_\_\_ Date \_\_\_\_\_

For Jabiru Aircraft Pty Ltd  
Production Certificate #444128



## 8.23 Jabiru 2200 – Component Assembly Procedure

### 8.23.1 Subassembly E – Flywheel and Ignition Coils

| <u>No.</u> | <u>Details</u>                                    | <u>Sign</u> | <u>Date</u> |
|------------|---|-------------|-------------|
| <b>E1</b>  | <b><u>Deburr and Clean</u></b>                    |             |             |
| <b>E2</b>  | <b><u>Fit magnets using silastic</u></b>          |             |             |
| <b>E3</b>  | <b><u>Using loctite 620 fit the ring gear</u></b> |             |             |
| <b>E4</b>  | <b><u>Fit the alternator rotor</u></b>            |             |             |
| <b>E5</b>  | <b><u>Fit the plugs to the ignition coils</u></b> |             |             |

**I hereby certify that the subassembly has been completed using the approved procedures and data; and is fit for installation.**

Signed: \_\_\_\_\_ Date: \_\_\_\_\_

For Jabiru Aircraft Pty Ltd  
Production Certificate #444128

Notes: \_\_\_\_\_  
\_\_\_\_\_  
\_\_\_\_\_  
\_\_\_\_\_

## 8.24 Component Register and Parts Inspection Record

### 8.24.1 Subassembly E – Flywheel Assy

**Batch** \_\_\_\_\_ **Serial No.** \_\_\_\_\_ **Date** \_\_\_\_\_

| Description | Part No  | Batch | Item | Details |
|-------------|----------|-------|------|---------|
| Flywheel    | 4645183  |       |      |         |
| Ring Gear   | 4066423  |       |      |         |
| Coil 1      | PI10522N |       |      |         |
| 2           | PI10522N |       |      |         |

I hereby certify that the above listed parts conform with the dimensions, have been engraved, and installed as recorded.

Signed \_\_\_\_\_ Date \_\_\_\_\_

For Jabiru Aircraft Pty Ltd  
Production Certificate #444128



## 8.25 Jabiru 2200 – Component Assembly Procedure

### 8.25.1 Subassembly F – Gear Case

|                | No.       | Details   | Sign | Sign | Date |
|----------------|-----------|---|------|------|------|
|                | F1        | Deburr, clean and inspect all of the gears                                    |      |      |      |
|                | F2        | Measure the shaft post internal diameters and the distributor shaft diameters |      |      |      |
|                | F3        | Using loctite 515 fit the shaft posts to the gear housing                     |      |      |      |
|                | F4        | Fit the distributor shaft seals and rear crankshaft seal                      |      |      |      |
|                | F5        | Fit shafts to gears with 24 hour araldite                                     |      |      |      |
|                | F6        | Fit the distributor shafts and gears to the gear housing                      |      |      |      |
|                | F7        | Check end clearance of distributor shaft to case flange                       |      |      |      |
| <b>Stage 2</b> | <b>F8</b> | <b>Stage Inspection of Assembly</b>   |      |      |      |

**I hereby certify that the subassembly has been completed using the approved procedures and data; and is fit for installation.**

Signed: \_\_\_\_\_

Date: \_\_\_\_\_

For Jabiru Aircraft Pty Ltd  
Production Certificate #444128

Notes: \_\_\_\_\_  
\_\_\_\_\_  
\_\_\_\_\_  
\_\_\_\_\_  
\_\_\_\_\_

## 8.26 Component Register and Parts Inspection Record

### 8.26.1 Subassembly F – Gear Case

Batch \_\_\_\_\_ Serial No. \_\_\_\_\_ Date \_\_\_\_\_

| Description                | Part No | Batch | Item | Details |
|----------------------------|---------|-------|------|---------|
|                            |         |       |      | Dia     |
| Distributor Mount Spacer 1 | 4632074 |       |      |         |
| 2                          | 4632074 |       |      |         |
| Distributor Drive Shaft 1  | 4332154 |       |      |         |
| 2                          | 4332154 |       |      |         |
| Gear Housing               | 4631073 |       |      |         |
| Distributor Gear 1         | 4333054 |       |      |         |
| 2                          | 4333054 |       |      |         |

I hereby certify that the above listed parts conform with the dimensions, have been engraved, and installed as recorded.

Signed \_\_\_\_\_ Date \_\_\_\_\_

For Jabiru Aircraft Pty Ltd  
Production Certificate #444128

## 8.27 Jabiru 2200 – Component Assembly Procedure

### 8.27.1 Subassembly G – Fuel Pump and Carburettor

| No. | Details  | Sign | Date |
|-----|--|------|------|
| G1  | Fuel Pump – Disassembly                        |      |      |
| G2  | Drill hole in spacer and insert breather       |      |      |
| G3  | Final assembly checked for correct flow        |      |      |
| G4  | Drill and tap upper half for 90o elbow and fit |      |      |
| G5  | Clean and reassemble                           |      |      |
| G6  | Carburettor needle position                    |      |      |
| G7  | Fit fuel line                                  |      |      |
| G8  | Remove bowl and check jet size                 |      |      |
| G9  | Reassembly bowl                                |      |      |
| G10 | Fuel pump and carby engraved                   |      |      |

I hereby certify that the subassembly has been completed using the approved procedures and data; and is fit for installation.

Signed: \_\_\_\_\_

Date: \_\_\_\_\_

For Jabiru Aircraft Pty Ltd  
Production Certificate #444128

Notes: \_\_\_\_\_

\_\_\_\_\_  
\_\_\_\_\_  
\_\_\_\_\_  
\_\_\_\_\_

## 8.28 Jabiru 3300 – Component Assembly Procedure

### 8.28.1 Subassembly H – Final Assembly

| No.     | Details  | Sign | Sign | Date |
|---------|--|------|------|------|
| H1      | Apply Loctite 515 to crankcase halves; lubricate where necessary |      |      |      |
| H2      | Locate camshaft and crankshaft in crankcase half                 |      |      |      |
| Stage 3 | Stage Inspection Pre-joining Crankcase                           |      |      |      |
| H3      | Locate crankcase half over crankshaft and camshaft               |      |      |      |
| H4      | Tension front (2) and rear (2) studs                             |      |      |      |
| H5      | Fit oil pump assembly and front seal housing                     |      |      |      |
| H6      | Place all through bolts in crankcase                             |      |      |      |
| H7      | Fit piston and cylinder assemblies and check circlips            |      |      |      |
| Stage 4 | Stage Inspection Piston/Circlips                                 |      |      |      |
| H8      | Tension the cylinder base studs to 25 ft.lbs then 30 ft.lbs      |      |      |      |
| H9      | Fit sump   |      |      |      |
| H10     | Fit heads to cylinder torque 12 ft.lbs / 24 ft.lbs               |      |      |      |
| H11     | Fit push rod and valve gear. Set gap to 0.10" (0.25mm)           |      |      |      |
| H12     | Check camshaft timing DEGREES                                    |      |      |      |
| H13     | Fit fuel pump  |      |      |      |
| H14     | Fit engine mount plate and gearbox housing                       |      |      |      |
| H15     | Fit flywheel alternator mount plate and ignition coils and lead  |      |      |      |
| H16     | Fit carby assembly   |      |      |      |
| H17     | Fit exhaust system   |      |      |      |





**I hereby certify that the subassembly has been completed using the approved procedures and data; and is fit for installation.**

Signed: \_\_\_\_\_

Date: \_\_\_\_\_

For Jabiru Aircraft Pty Ltd  
Production Certificate #444128

Notes: \_\_\_\_\_  
\_\_\_\_\_  
\_\_\_\_\_  
\_\_\_\_\_



## 9 Jabiru 2200A – Engine Test Schedule and Report

Engine s/n:

| Date | Temp | Wind | QNH | Humidity |
|------|------|------|-----|----------|
|      |      |      |     |          |

Run-in

Use Test Rig #1, Test Propeller #J1  
Run into the Wind  
Use Schedule Provided  
Oil – Aeroshell 100

Test Results

| Time [min] | Start Time | Oil Press [kPa] | Oil Temp [°C] | EGT [°C] | CHT #1 [°C] | CHT #2 [°C] | CHT #3 [°C] | CHT #4 [°C] |
|------------|------------|-----------------|---------------|----------|-------------|-------------|-------------|-------------|
| 0          |            |                 |               |          |             |             |             |             |
| 30         |            |                 |               |          |             |             |             |             |
| 45         |            |                 |               |          |             |             |             |             |
| 60         |            |                 |               |          |             |             |             |             |
| 90         |            |                 |               |          |             |             |             |             |
| 105        |            |                 |               |          |             |             |             |             |
| 120        |            |                 |               |          |             |             |             |             |

Maximum RPM: \_\_\_\_\_

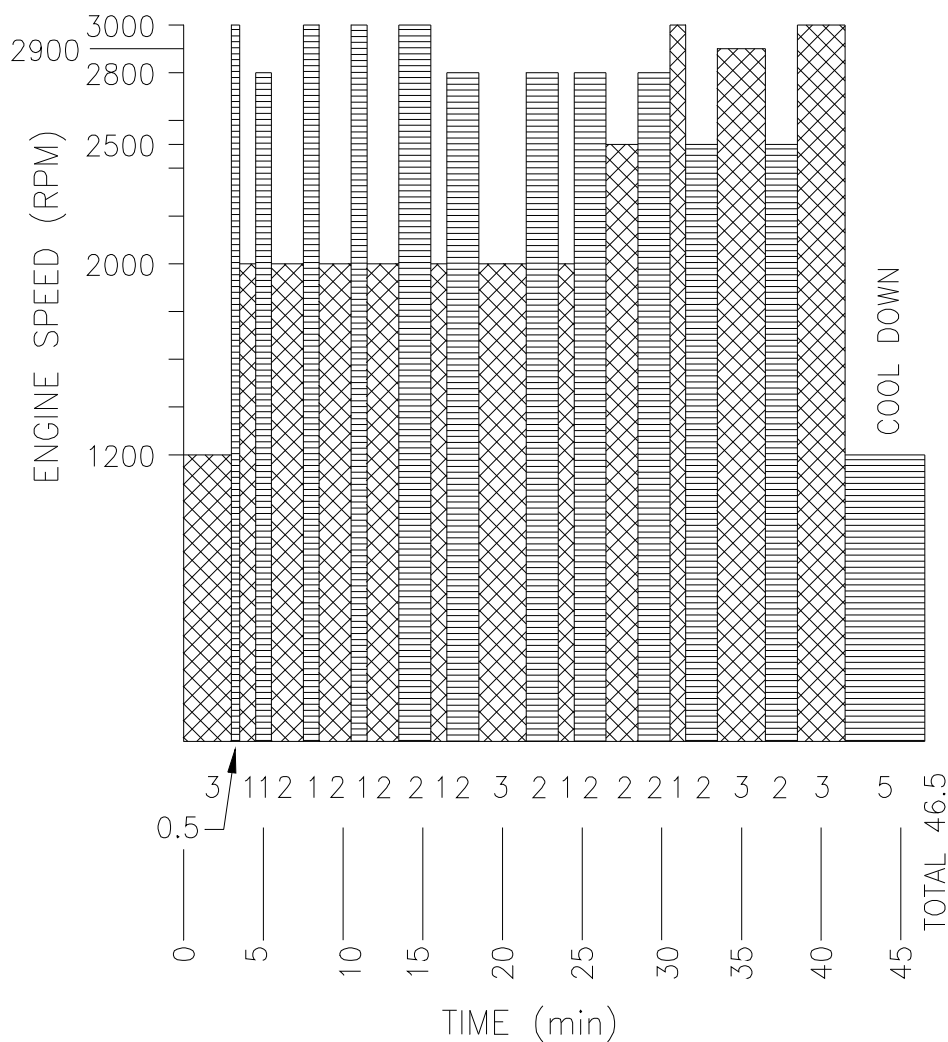
|             |                  |
|-------------|------------------|
| Fuel Grade  | Fuel Consumption |
| Avgas 100LL | L                |

I hereby certify that this engine has been tested in accordance with the approved procedures and the Jabiru Aircraft Pty Ltd, Engines Division, Policy and Procedures Manual and is fit for use.

Signed: \_\_\_\_\_

Date: \_\_\_\_\_

# JABIRU 2200 ENGINE RUN-IN SCHEDULE



DATE:

ENGINE S/N:

SIGNATURE:

Drawing 9486093/1 RUN-IN SCHEDULE

Figure 37 – Drawing 9486093

9.1 Jabiru 2200A Aero Engine Ground Test Form

Serial No \_\_\_\_\_

|          |      |             |                |      |          |      |      |      |       |       |       |     |       |          |         |
|----------|------|-------------|----------------|------|----------|------|------|------|-------|-------|-------|-----|-------|----------|---------|
| Date     |      | Start Time  |                |      |          |      |      | Temp |       | Wind  |       | QNH |       | Humidity |         |
| Operator |      | Finish Time |                |      |          |      |      |      |       |       |       |     |       |          |         |
| Block    |      | Total       |                |      |          |      |      |      |       |       |       |     |       |          |         |
|          |      |             |                |      |          |      |      |      |       |       |       |     |       |          |         |
| Seq      | Time | Duration    | Condition      | RPM  | Rec. RPM | CHT1 | CHT2 | CHT3 | CHT 4 | CHT 5 | CHT 6 | EGT | Oil T | Oil P    | Current |
| 1        |      | 3           | Start & Idle   |      |          |      |      |      |       |       |       |     |       |          |         |
| 2        |      | 30 Sec/1    | Take-Off       |      |          |      |      |      |       |       |       |     |       |          |         |
| 3        |      | 1/2         | 75% Power      |      |          |      |      |      |       |       |       |     |       |          |         |
| 4        |      | 1/2         | Take-Off       |      |          |      |      |      |       |       |       |     |       |          |         |
| 5        |      | 15 sec /2   |                |      |          |      |      |      |       |       |       |     |       |          |         |
| 6        |      | 2           | 75% Power      |      |          |      |      |      |       |       |       |     |       |          |         |
| 7        |      | 1           | Cooling Run    |      |          |      |      |      |       |       |       |     |       |          |         |
| 8        |      | 2           | 75% Power      |      |          |      |      |      |       |       |       |     |       |          |         |
| 9        |      | 3           | Cooling Run    |      |          |      |      |      |       |       |       |     |       |          |         |
| 10       |      |             | Max Cont Power |      |          |      |      |      |       |       |       |     |       |          |         |
|          |      | 2 / 1 / 2   | 2600/2000/2600 |      |          |      |      |      |       |       |       |     |       |          |         |
|          |      | 2 / 2 / 1   | 2000/2600/2700 |      |          |      |      |      |       |       |       |     |       |          |         |
|          |      | 2 / 3       | 2000/2600      |      |          |      |      |      |       |       |       |     |       |          |         |
|          |      | 2 / 3       | 2000/2700      |      |          |      |      |      |       |       |       |     |       |          |         |
| 11       |      | 3           | Cooling Run    | 1200 |          |      |      |      |       |       |       |     |       |          |         |

|                          |  |
|--------------------------|--|
| Record of Replenishments |  |
| Start Fuel Quantity      |  |
| Fuel Added               |  |
| Finish Fuel Quantity     |  |
| Fuel Used Using Block    |  |

|              |  |
|--------------|--|
| Max RPM      |  |
| Low Idle RPM |  |
| Charge Rate  |  |
| Oil Added    |  |

|                                     |  |
|-------------------------------------|--|
| Record of Maintenance or Adjustment |  |
|                                     |  |
|                                     |  |
|                                     |  |

I hereby certify that the above listed run has been carried out in accordance with the details give.

Signed \_\_\_\_\_ Date \_\_\_\_\_

Note: This chart gives a run time of 2 hours, engines now take approx. 50 minutes with this format but with shorter time intervals.

For Jabiru Aircraft Pty Ltd  
Certificate of Approval #444128

REVISION

7

Dated : Oct 2006

Document No. JEM2201-7

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## 9.2 Run-In Checks

### Preliminary

Paperwork Complete  
Sufficient Fuel  
Correct Oil Level  
All Connections Correct and Secure

### Start Run

Start Time \_\_\_\_\_  
End Time \_\_\_\_\_

### Ignition Checks (repeat a couple of times for each side)

Left Side 1000/2000 rpm  
Right Side 1000/2000 rpm

### Oil Leaks

Check Visually

### Instruments

Working and Correct Range

### Idle

Check idle screw (3/4 to 1 turn out)

### General Running

Smoothness  
Noises  
Manifold Sealing

### Oil Pressure and Temp

### Check Ignition light to Tacho Test Rig (Top End RPM)

### Check alternator Charging Voltage

### 9.3 Jabiru 2200A – Engine Post Run Procedure

|   | SIGN | DATE |
|---|------|------|
| 1. Heads retorqued 32 Nm (24 ft/lb). Valves Adjusted.   |      |      |
| 2. Check induction/exhaust bolts.   |      |      |
| 3. Any changes to be made.  |      |      |
| 4. Rerun, check for oil leaks and/or any modifications made (oil pressure/leaks etc).   |      |      |
| 5. Check charging rate of alternator Volts  |      |      |
| 6. Leak Down  |      |      |
| 80                      80                      80                      80<br>Cyl 1 <input type="text"/> 2 <input type="text"/> 3 <input type="text"/> 4 <input type="text"/> |      |      |
| 7. Check all paper work   |      |      |
| 8. Drain fuel/oil. Prepare for Shipment.  |      |      |



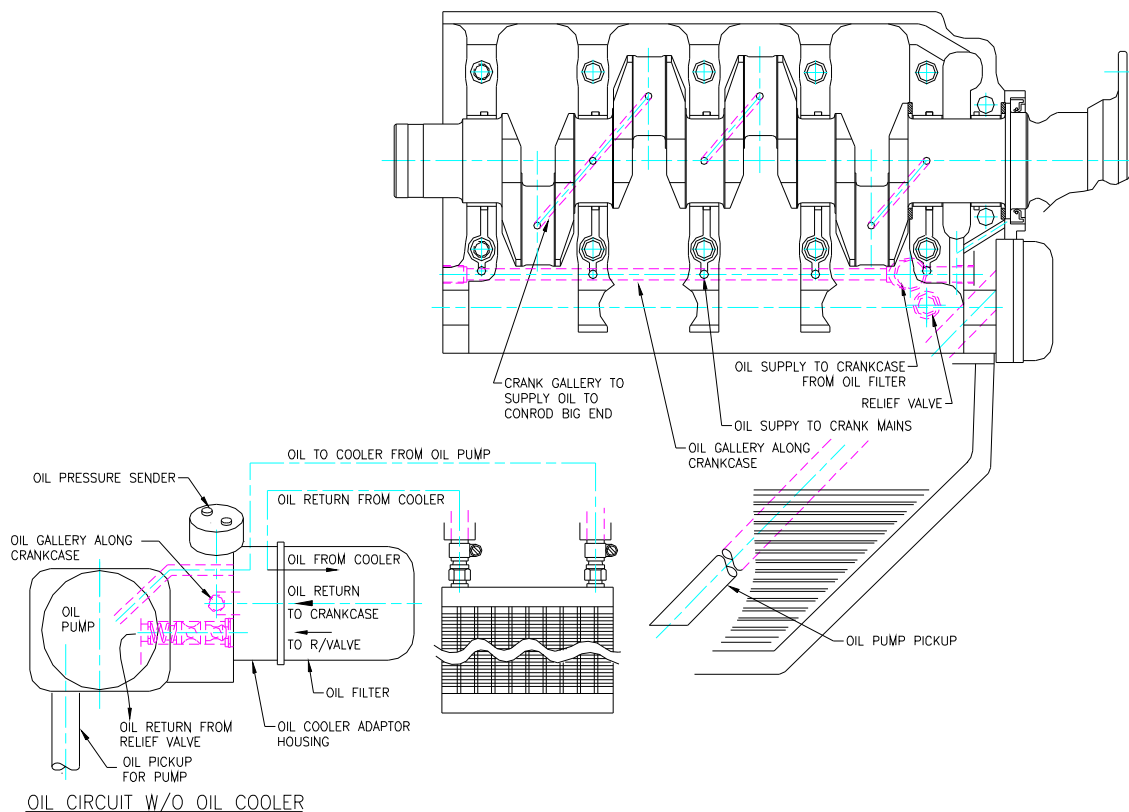
## 10 TABLE OF LUBRICANTS

Use only oils of registered brands meeting the specification detailed at para. 2.5.

### Acceptable engine lubricating oils:

| Run in Period     |               |              |       |
|-------------------|---------------|--------------|-------|
| Oil               | 80            | 100          | 120   |
| Outside Air Temp  | -17°C to 25°C | 15°C to 35°C | 35°C+ |
| Normal Operations |               |              |       |
| Oil               | W80           | W100         | W120  |
| Outside Air Temp  | -17°C to 25°C | 15°C to 35°C | 35°C+ |

**Note: Multigrade aviation oils are also available.**



### Figure 38 – Oil Circuit

## 11 TORQUE SPECIFICATION FOR BOLTS & NUTS

| Part  | Nom. Dia (mm) | Torque<br>nm (ft.lbs) |
|---|---------------|-----------------------|
| Spark Plugs                                     | 12mm          | 11 (8)                |
| Cylinder Head Bolts                             | 5/16"         | 34 (24)               |
| Crankcase Main Studs                            | 3/8"          | 40 (30)               |
| Flywheel/Gear Bolts                             | 1/4"          | 20 (15)               |
|   | 5/16"         | 33 (24)               |
| Crankshaft Prop Flange Cap Screws<br>(Lockwire) | 3/8"          | 40 (30)               |
| Oil Pump Cap Screws                             | 5/16"         | 20 (15)               |
| Tappet Cover Cap Screws                         | 1/4"          | 7 (5)                 |
| Starter Motor Bolts                             | 5/16"         | 20 (15)               |
| Carburettor Flange Bolts                        | 1/4"          | 11 (8)                |
| Engine Mount Plate Bolts                        | 1/4"          | 14 (10)               |
|   | 5/16"         | 16 (12)               |
| Gearbox Cover Bolts                             | 1/4"          | 14 (10)               |
| Alternator & Coil Mount Bolts                   | 1/4"          | 14 (10)               |
| Sump Cap Screws                                 | 1/4"          | 14 (10)               |
| Conrod Bolts                                    | 8mm           | 14 (10)               |
|   | 5/16"         | 24 (18)               |
| Propeller Bolts                                 | 1/4"          | 9.5 (7)               |
| Camshaft Gear Bolts (Lockwire)                  | 1/4"          | 11 (8)                |

### 11.1 PRESCRIBED SEALANTS AND PRIMERS

| <u>Item</u>           | <u>Sealant</u>              |
|-----------------------|-----------------------------|
| Main Bearing Studs    | Loctite 620                 |
| Conrod Bolts          | Loctite 620                 |
| Oil Pump Bolts        | Loctite 243                 |
| Sump Plate Bolts      | Loctite 243                 |
| Engine Mount Bolts    | Loctite 243                 |
| Gearbox Cover Bolts   | Loctite 243                 |
| Spark Plugs NGK D9EA* |                             |
| Spark Plug Lubricant  | Loctite "Nickel Anti-Seize" |
| Prop Mount            | Loctite 620                 |
| Flywheel-Crank bolts  | Loctite 620                 |
| Camshaft              | Loctite 262                 |

\* Tighten to finger tight to seat, then with plug socket turn an additional 1/2 turn (8 ft/lbs) for a new plug.

\*\* All gasket areas use Loctite 'Gasket Eliminator' 515



\*\*\* LOCTITE PRIMER 7471 used on threads before LOCTITE



## 11.2 NEW TOLERANCES

(All Dimensions are in Millimetres)

|   |                            |                    |
|---|----------------------------|--------------------|
| <i>Crankshaft</i>                           | Mains Journals             | 47.938 to 47.918   |
|   | Big Ends Journals          | 44.988 to 45.000   |
|   | Thrust Face                | 56.95 to 57.05     |
| <i>Crankcase</i>                            | Main Bearings              | 47.975 to 48.005   |
|   | Cam Bearings               | 20.00 to 20.020    |
|   | Cam Thrust Face            | 14.95 to 15.10     |
|   | Lifter Bores               | 9.000 to 9.050     |
|   | Crank Thrust               | 56.65 to 56.85     |
| <i>Con Rods</i>                             | Big Ends                   | 45.028 to 45.070   |
|   | Little Ends                | 23.005 to 23.015   |
|   | Length Between Bore Radius | 74.485 to 74.498   |
|   |                            |                    |
| <i>Cam Shaft</i>                            | Journals                   | 19.96 to 19.97     |
|   | Valve Lift                 | 6.900 to 7.100     |
|   | Fuel Pump Lift             | 2.45 to 2.55       |
|   | Thrust Faces               | 15.18 to 15.25     |
| <i>Lifter</i>                               | Stems                      | 8.965 to 8.990     |
| <i>Pistons</i>                              | Dia                        | 97.50 to 97.53     |
|   | Height                     | 64.00 – 65.50      |
|   | Sidering Clearance         | 0.01 to 0.02       |
|   | Pin Dia                    | 22.990 to 23.000   |
|   | Ring End Gap               | 0.400 to 1.000     |
| <i>Cylinder Bore</i><br>Length over flanges | Dia                        | 97.59 to 97.61     |
|   | (A)                        | 106.450 to 106.500 |
|   | (B)                        | 106.950 – 107.00   |
| <i>Valve</i>                                | Stem Dia Inlet             | 6.970 to 6.990     |
|   | Stem Dia Exhaust           | 6.970 to 6.990     |
|   | Guide I.D. Inlet           | 7.040 to 7.050     |
|   | Guide I.D. Exhaust         | 7.040 to 7.050     |
|   | Spring Free length         | 39.5 to 42         |
| <i>Distributor</i>                          | Shaft dia                  | 14.940 to 14.970   |
|   | Shaft Post I.D.            | 15.00 to 15.03     |
|   | Shaft End Float            | 1mm                |

**Note:** Early engine components can exhibit slight variations to these new tolerances but still be serviceable.

### 11.3 MAXIMUM ALLOWABLE CLEARANCES (Wear Limits)

(All Dimensions are in Millimetres)

|                             |                     |
|-----------------------------|---------------------|
| Mains                       | 0.10                |
| Big Ends                    | 0.08                |
| Little Ends                 | 0.03                |
| Camshaft Journals           | 0.08                |
| Crank End Float             | 0.80                |
| Cam End Float               | 0.50                |
| Lifter - Crankcase          | 0.12                |
| Piston/Cylinder             | 0.15                |
| Ring Side Clearance         | 0.05                |
| Pin/Piston                  | 0.04                |
| Ring End Gap                | 1.20                |
| Inlet Valve/Guide           | 0.10                |
| Exhaust Valve/Guide         | 0.12                |
| Min Spring Length           | 39.50               |
| Distributor Shaft/Post      | 0.10                |
| Distributor Shaft End Float | 1.20                |
| Cylinder Dia at Half stroke | 97.65               |
| Rocker Shaft/Rocker Arm     | 0.18                |
| Oil Pump Gears              | 0.15                |
| Bearing Clearances main     | 0.04 – 0.09 typical |
| Big Ends                    | 0.04 – 0.06 typical |

**Note:** Variation can occur, overall tolerances may be acceptable with advice from Jabiru Aircraft Pty Ltd.

## 11.4 ELECTRICAL SYSTEMS SPECIFICATIONS

(All Dimensions are in Millimetres)

|                   |                             |                           |
|-------------------|-----------------------------|---------------------------|
| <i>Ignition</i>   | Primary Resistance          | 8.8R to 1.0R              |
|                   | Secondary Resistance        | 5.9kR to 7.1kR            |
|                   | Coil Gap                    | 0.27                      |
|                   | Plug Gap                    | 0.55                      |
|                   | Ignition Harness Resistance | 6.7kR per 300mm of length |
| <i>Alternator</i> | Coil Resistance             | 0.5 to 1.1R               |
|                   | Coil Earth Resistance       | Infinite                  |
|                   | A.C. Output                 | 30 VAC at 3000 RPM        |
|                   | D.C. Output                 | 14.2 VDA at 3000 RPM      |
| <i>Tacho</i>      | Coil Resistance             | 160 to 170R               |
|                   | Gap                         | 0.4                       |



## 12 TROUBLE SHOOTING

### 12.1 Engine Won't Start

|     | Possible Cause  | Remedy  |
|-----|---|---|
| 1)  | Ignition OFF  | Switch ON   |
| 2)  | Spark plug gap too large  | Adjust gap to 0.6-0.7mm or renew plugs  |
| 3)  | Closed fuel tap or clogged filter   | Open tap, renew filter, check fuel system for leaks   |
| 4)  | No fuel in tank   | Refuel  |
| 5)  | Wrongly connected high tension leads  | Connect as shown on leads   |
| 6)  | Starting Speed too low, faulty or discharged battery                          | Recharge or replace battery   |
| 7)  | Coil to Magnet gap too wide   | Adjust to 0.25mm (0.010")   |
| 8)  | High tension leads loose or damaged   | Check or renew connections  |
| 9)  | Dampness in distributors  | Thoroughly dry internally   |
| 10) | Spark plugs damp due to condensation  | Thoroughly dry both inside and outside of plugs   |
| 11) | Plug face wet by fuel due to excessive actuation of choke or overflow of carb | Dry spark plugs, trace possible faults in fuel system or over flow of carb.                                       |
| 12) | Float valve dirty or jammed   | Clean or renew float valve  |
| 13) | Jets in carb. clogged   | Clean jets  |
| 14) | Water in carb.  | Drain & clean carb., fuel line & filter. Water drain fuel tank  |
| 15) | Insufficient compression  | Trace pressure loss & repair if necessary   |
| 16) | Engine damage   | Inspect oil strainer filter & oil filter for metallic particles. If present, an engine overhaul may be necessary. |

### 12.2 Engine Idles Unsteadily After Warm-Up Period: Smoky Exhaust Emission

|    | Possible Cause                    | Remedy                                      |
|----|-----------------------------------|---|
| 1) | Choke activated                   | Close choke                                 |
| 2) | Float valve dirty, jammed or worn | Clean or renew float valve                  |
| 3) | Intake manifold leak              | Tighten all connections, renew faulty items |

### 12.3 Engine Runs Erratically or Misfires Occasionally

|    | Possible Cause       | Remedy   |
|----|----------------------|--|
| 1) | Spark plug failure   | Check plugs, clean inside & outside, adjust electrode gap. If necessary, renew plugs |
| 2) | Faulty HT leads      | Dry damp leads, renew damaged leads  |
| 3) | Faulty ignition unit | Renew ignition unit  |
| 4) | Clogged fuel filter  | Renew fuel filter  |

## 12.4 Engine Runs Too Hot - Oil Temperature Above 110°C (230°F)

|    | Possible Cause               | Remedy   |
|----|------------------------------|--|
| 1) | Too much oil in crankcase    | Check oil level & adjust if necessary  |
| 2) | Low oil level                | Check oil level & add oil if necessary   |
| 3) | Poor quality oil             | Oil change, use specified oil  |
| 4) | Clogged oil filter           | Change filter  |
| 5) | Excessive piston blow by     | Common reason: worn or sticking piston rings, complete engine overhaul necessary |
| 6) | Faulty bearings              | If metallic particles are present in oil, complete engine overhaul necessary     |
| 7) | Faulty oil temperature gauge | Exchange gauge   |

## 12.5 Unsatisfactory Power Output

|    | Possible Cause                       | Remedy  |
|----|--------------------------------------|---|
| 1) | Ignition failure                     | Check ignition circuits; check wiring and pick-ups; replace ignition units. |
| 2) | Too much oil in crankcase            | Check oil level & adjust if necessary                                       |
| 3) | Insufficient fuel supply             | Check fuel supply system  |
| 4) | Fuel not according to specifications | Re-fuel with specified fuel   |
| 5) | Incorrect throttle adjustment        | Re-adjust throttle fitting  |
| 6) | Leak in air intake                   | Check and tighten all connections, check carby sockets.                     |
| 7) | Carby diaphragm damage               | renew diaphragm   |

## 12.6 Low Oil Pressure

|    | Possible Cause                          | Remedy  |
|----|---|---|
| 1) | Insufficient oil in sump                | Check oil level & replenish as necessary          |
| 2) | High oil temperature                    | Refer to Para. 10.4                               |
| 3) | Faulty pressure gauge, sender or wiring | Check gauge, sender & wiring. Renew as necessary. |
| 4) | Faulty crankshaft bearings              | Engine overhaul                                   |
| 5) | Relief valve not sealing                | Inspect, replace back after cleaning              |

## 12.7 Engine Keeps Running with Ignition Off

|    | Possible Cause         | Remedy   |
|----|------------------------|--|
| 1) | Idle speed too high    | Adjust to proper idle speed (900 RPM)              |
| 2) | Faulty ignition switch | Check switch & cables. Repair/replace as necessary |
| 3) | Overheated engine      | Conduct cooling run at 900 RPM                     |

## 12.8 Excessive Oil Consumption

|    | Possible Cause                              | Remedy                            |
|----|---|-----------------------------------|
| 1) | Worn, broken or wrongly fitted piston rings | Repair/engine overhaul necessary  |
| 2) | Poor oil quality                            | Oil change, use specified oil     |
| 3) | Worn valve guides                           | Repair of cylinder head necessary |
| 4) | Oil leaks                                   | Seal leaks                        |

## 12.9 Knocking Under Load

|    | Possible Cause                           | Remedy  |
|----|--|---|
| 1) | Octane rating of fuel too low            | Use fuel with higher octane rating  |
| 2) | Spark plug fitted without sealing washer | Ensure one sealing washer on each plug  |
| 3) | Heavy carbon deposits                    | Remove cylinder heads & in combustion chamber remove deposits. Determine oil consumption. |

## 12.10 Engine Hard to Start at Low Temperatu

|    | Possible Cause         | Remedy  |
|----|------------------------|---|
| 1) | Starting speed too low | Preheat engine  |
| 2) | High oil pressure      | At very low temperatures, a pressure reading of up to around 500 kpa doesn't necessarily indicate a malfunction |
| 3) | Low battery charge     | Fit fully charged battery   |

Note: Running this engine on low octane fuel will cause piston damage and in extreme case failure of the top ring gland or holed piston due to detonation. Changing the engine tuning to a leaner air fuel mix will cause piston damage.

## **JABIRU AIRCRAFT PTY LTD**

### **13 NEW ENGINE WARRANTY**

Jabiru Aircraft Pty Ltd, hereinafter JABIRU warrants that it will make good without charge, any defect which appears in this engine.

Provided:

1. the defect has been notified in writing to JABIRU:
  - (a) before the engine has operated a total of 200 hours or,
  - (b) within twelve (12) months of the date of delivery of the engine to the first retail purchaser or from the date of independent Authority authenticated first flight.

whichever comes first, and

2. the engine has been delivered to a JABIRU Approved Service Centre or such other service facility as advised by JABIRU, and
3. the engine has been installed in an aircraft type in accordance with a JABIRU approved installation system, and
4. JABIRU has determined that the defect complained of is one of workmanship and is not caused by:
  - (a) misuse or abuse of the engine such as by operation outside the approved Flight Manual, or Maintenance and Operation Manual, etc, or by neglect
    - (b) improper installation, including overheating.
    - (c) operation of the engine after it is known to be defective
    - (d) accident or deliberate act
    - (e) atmospheric fallout or flood, hail, salt, wind, etc.
    - (f) failure to carry out proper maintenance service
    - (g) use of incorrect types and/or grades of fuel, oil or lubricants
    - (h) alteration or modification of the engine by any party not authorised in writing by JABIRU
    - (i) the fitting of parts or accessories not marketed by JABIRU
    - (j) any work carried out on the engine by someone other than an Authorised JABIRU Service Centre or someone else authorised by JABIRU in writing,
  - (k) the use of any engine oil or fuel additives or oil stabilisers

BY JABIRU (or as otherwise decided by JABIRU) MAKING GOOD THE DEFECT BY REPAIR OR, AT THE OPTION OF JABIRU, BY REPLACEMENT.

Excluded from this Warranty are service items such as engine tuning, adjustments, replacement of air and oil filters, spark plugs, etc which are required as part of normal engine maintenance.



This Warranty is given to the person who is entitled to possession of the engine whether as owner, lessee or otherwise and is given in addition to all right conferred by law on that person.

Warranty repairs do not extend the original warranty.

Due to the substantial number of problems that can arise due to installation errors, JABIRU shall not be liable for any labour and/or service charges for removal, reinstallation and adjustment which are the responsibility of the buyer and are not covered by this Warranty. Consequential damages and freight costs are also not covered by this Warranty.

JABIRU makes no representation that this engine is suitable for installation in any particular aircraft and the responsibility for determining such suitability rests with the Buyer.

Under no condition shall JABIRU or a JABIRU Authorised Service Centre be liable for any contingent costs through the engine or aircraft being out of service for whatever reason.

### 13.1 SPECIAL NOTICE TO OWNERS

#### **AVAILABILITY OF SERVICE AND PARTS AFTER WARRANTY**

JABIRU Aircraft Pty Ltd maintains a substantial stock of spare parts and operates a Service Exchange Programme in respect to some components. Every endeavour is made to ensure that JABIRU carries adequate stocks of service parts and that Authorised Service Agents are equipped to provide satisfactory service, but JABIRU does not make any promise that after the expiration of the warranty such parts or service will be available, or available at any specific location or at any particular time.

#### **UNAUTHORISED STATEMENTS IN RELATION TO JABIRU PRODUCTS**

No JABIRU Authorised Service Centre or other person is authorised or permitted to give or make any statement assertion or undertaking in relation to the quality, performance, characteristics, descriptions or fitness for any purpose of any JABIRU product or in connection with the supply of any JABIRU product, which is at variance with any written statement assertion or undertaking on any of these subjects given or made by JABIRU in its published sales literature, and the company does not accept any such unauthorised action.

#### **WARRANTY ON JABIRU REPLACEMENT PARTS**

JABIRU warrants in respect of JABIRU parts and accessories required as replacement parts, that it will make good by repair or at its option by replacement any defect occurring in any such JABIRU parts and accessories within twelve (12) months from the date of acquisition. Normal wear and tear is excluded. This warranty does not cover those parts listed as exclusions in the New Engine Warranty and is subject to the same general exclusions.





## 13.2 JABIRU WARRANTY: CLAIM FORM

FROM: \_\_\_\_\_ DATE: \_\_\_\_\_

ENGINE PARTS: ☐AIRFRAME PARTS: ☐ENGINE NUMBER: \_\_\_\_\_  
NUMBER: \_\_\_\_\_

AIRFRAME KIT

PART NUMBER: \_\_\_\_\_ TOTAL HOURS: \_\_\_\_\_

OWNER: \_\_\_\_\_ PREVIOUS OWNER: \_\_\_\_\_

PART DESCRIPTION: \_\_\_\_\_

CLAIM: \_\_\_\_\_

FAULTY GOODS RETURNED: YES ☐ COURIER Co. REF NO: \_\_\_\_\_NO ☐

ALL GOODS THAT ARE BOXED AND HAVE TO PASS THROUGH CUSTOMS HAVE TO BE CLEANED AND FREE FROM CONTAMINATION WITH A STATEMENT ATTACHED SPECIFYING HOW THEY WERE CLEANED AND THAT THEY ARE FREE OF DIRT AND GRASS SEEDS. IF THEY ARE NOT CLEANED TO CUSTOMS SATISFACTION AN EXTRA CLEANING CHARGE WILL APPLY.

IF WARRANTY CLAIM IS NOT COMPLETELY AND CORRECTLY COMPLETED WARRANTY MAY BE REFUSED.

## OFFICE USE ONLY

APPROVED ☐NOT APPROVED ☐REASON \_\_\_\_\_  
\_\_\_\_\_

PRINTED NAME: \_\_\_\_\_

SIGNED: \_\_\_\_\_ DATED: \_\_\_\_\_