

Re = 3000000  
Mach = 0.0000  
NCrit = 6.00

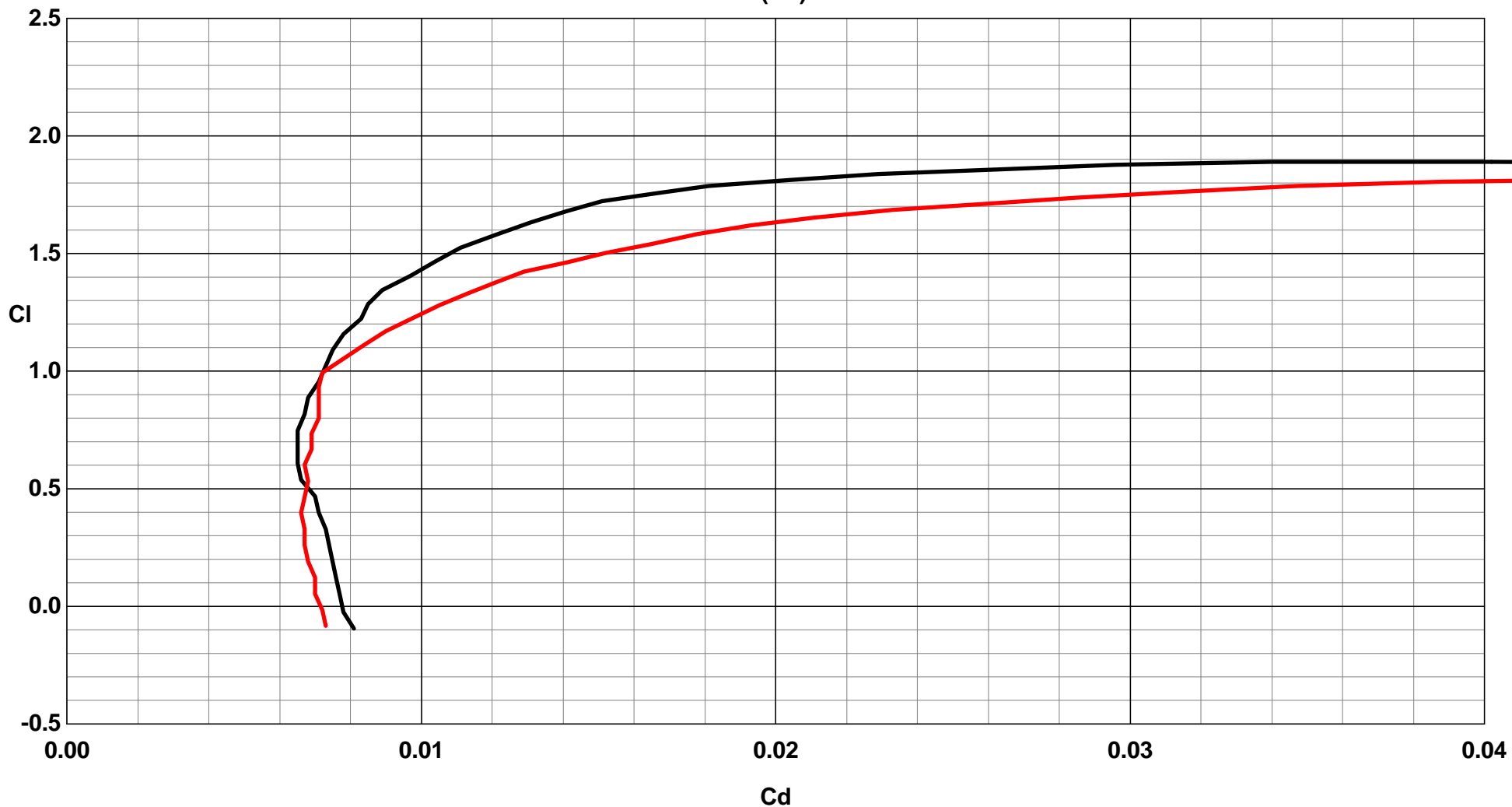
MID 415 mid393 modified 0.038 camber in =



NACA4415 =



Cl(Cd):



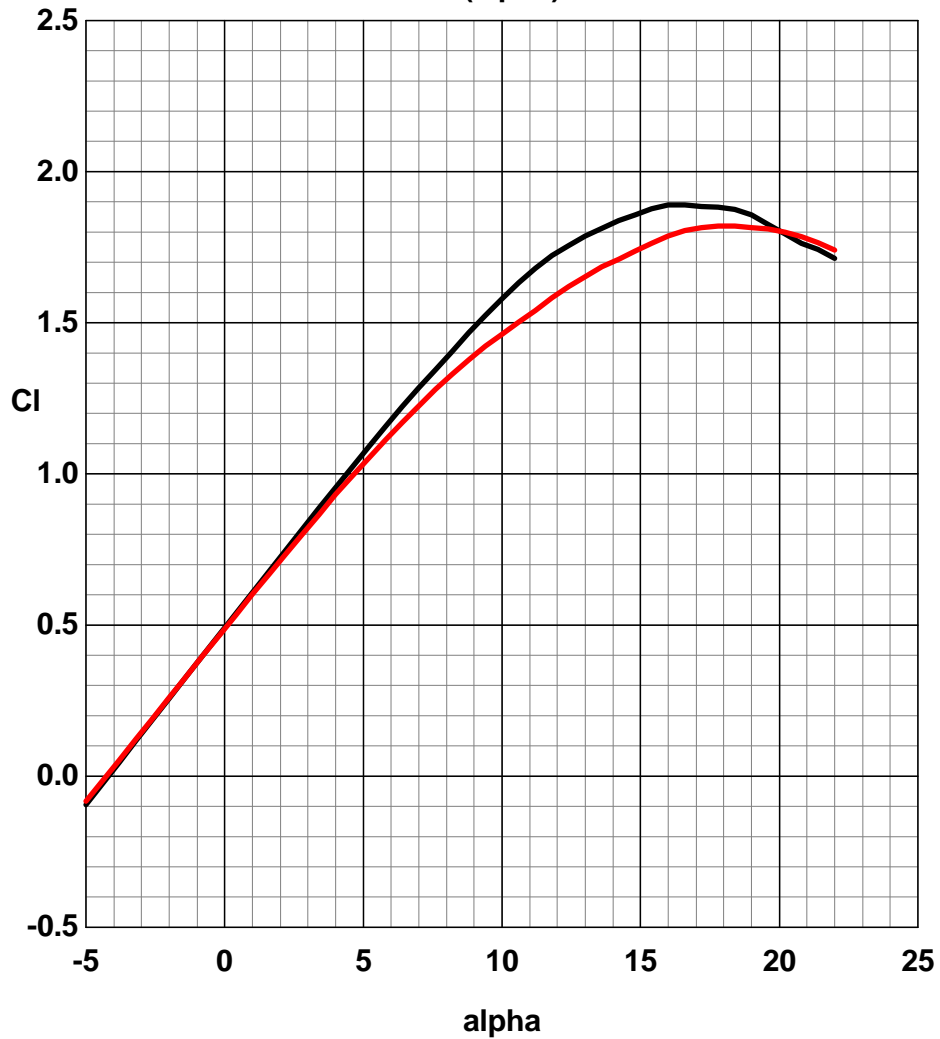
Re = 3000000  
Mach = 0.0000  
NCrit = 6.00

MID 415 mid393 modified 0.038 camber in =

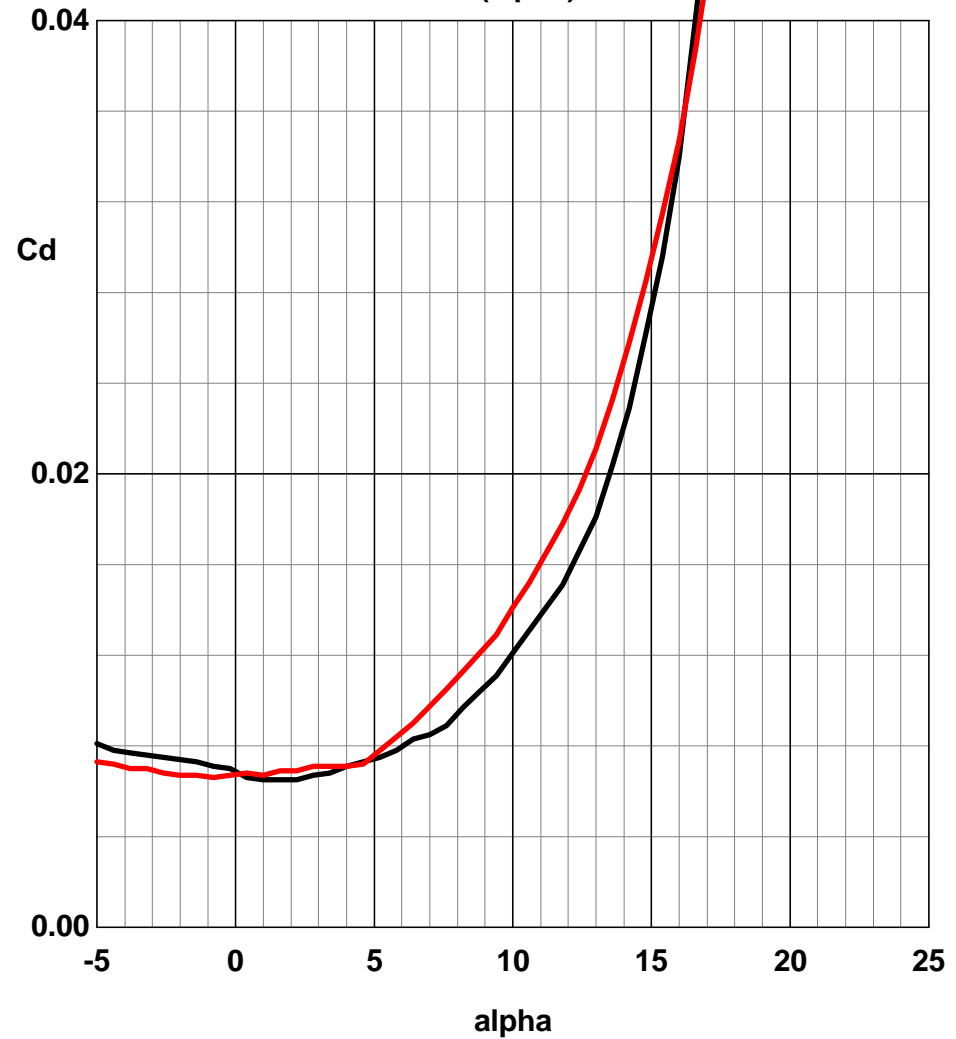
NACA4415 =



Cl(alpha):



Cd(alpha):



Re = 3000000  
Mach = 0.0000  
NCrit = 6.00

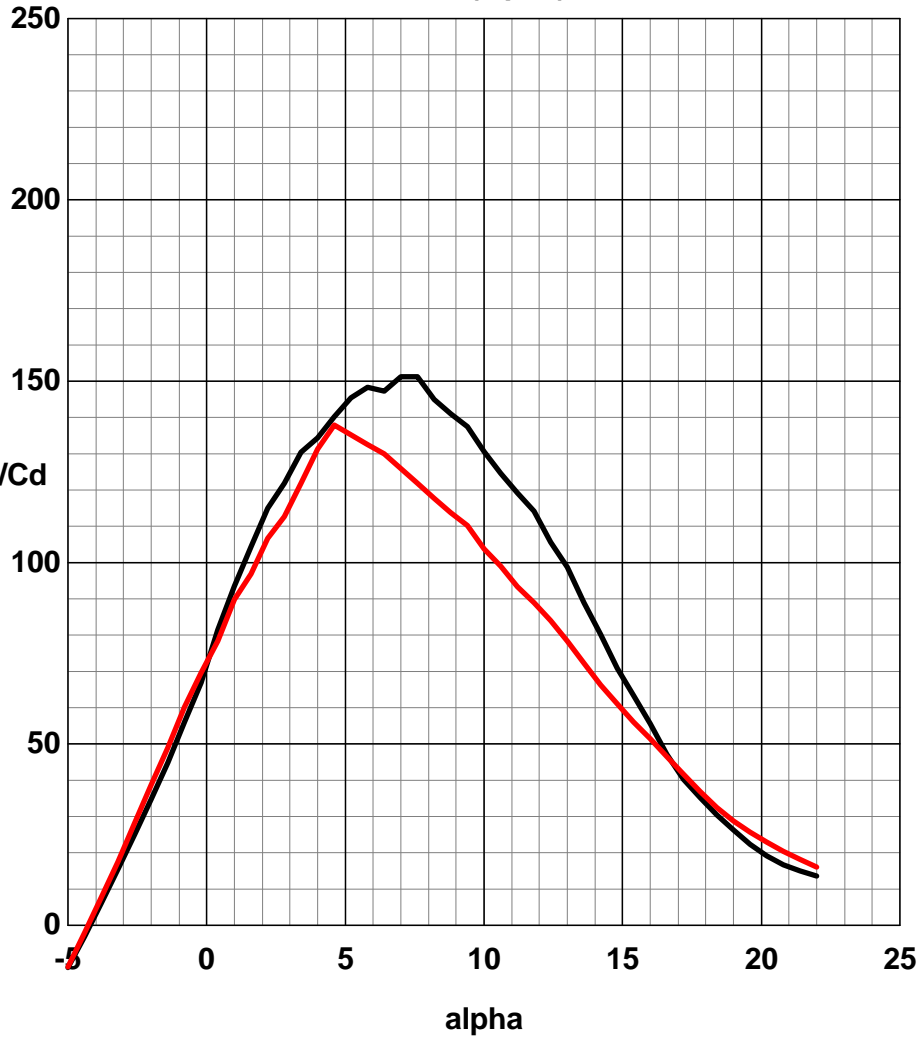
MID 415 mid393 modified 0.038 camber in =



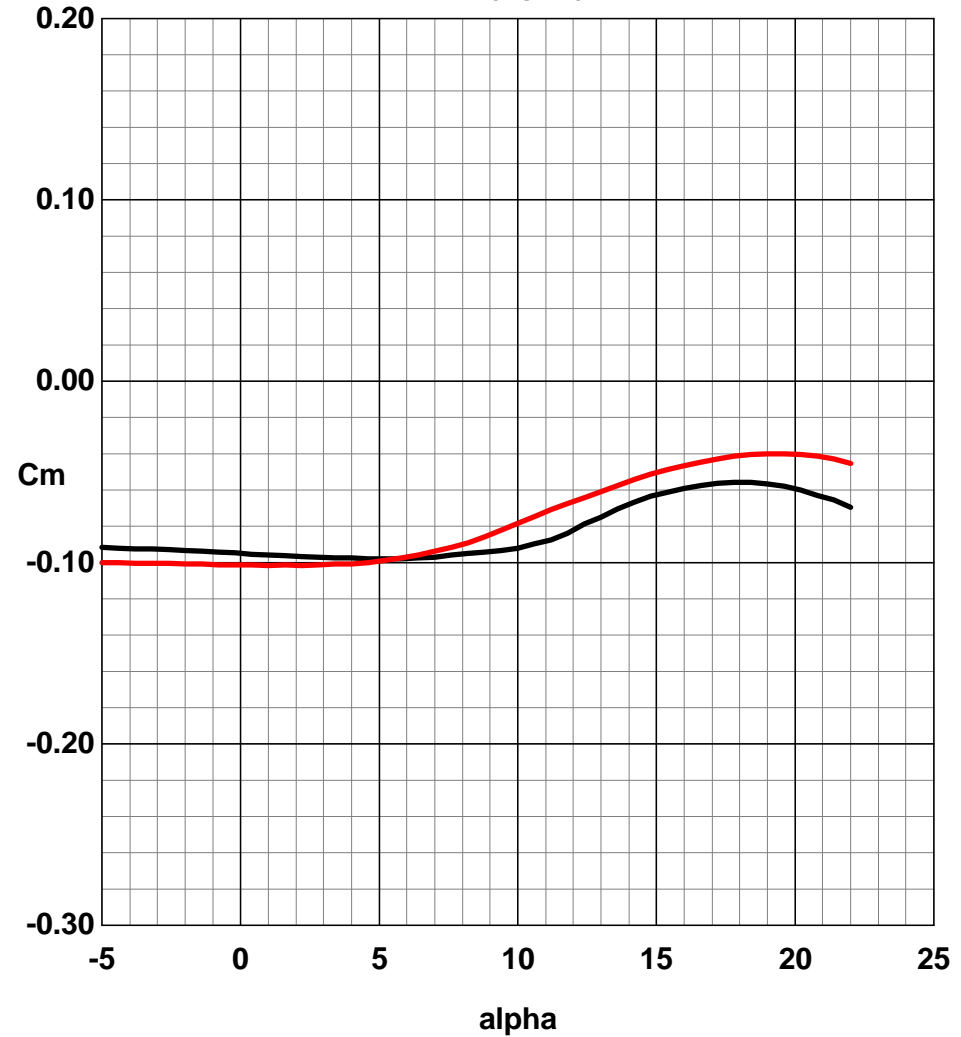
NACA4415 =



Cl/Cd(alpha):



Cm(alpha):



Re = 3000000  
Mach = 0.0000  
NCrit = 6.00

MID 415 mid393 modified 0.038 camber in =



NACA4415 =



Power Factor(alpha) =  $Cl^{1.5}/Cd(alpha)$  :

